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NASA-CR-
171092

MATERIAL PROPERTIES, LOADS, AND STRESS ANALYSIS - SPARTAN REM

Appendix A

19 June 1984

Contract NAS8-35599

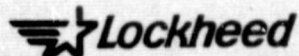
Prepared for

**NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
MARSHALL SPACE FLIGHT CENTER, AL 35812**

by

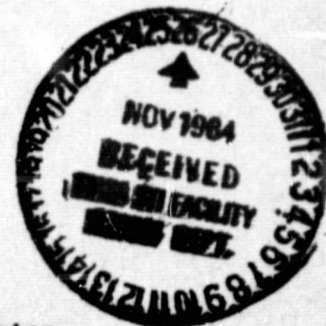
Donna S. Marlowe

Erik J. West



**Research & Development Division
Huntsville Research & Engineering Center**

**Cummings Research Park
4800 Bradford Drive,
Huntsville, AL 35807**



(NASA-CR-171092) MATERIALS PROPERTIES,
LOADS, AND STRESS ANALYSIS, SPARTAN REM:
APPENDIX A (Lockheed Missiles and Space Co.)
124 p HC A06/MF A01

CSCI 20K

N85-11375

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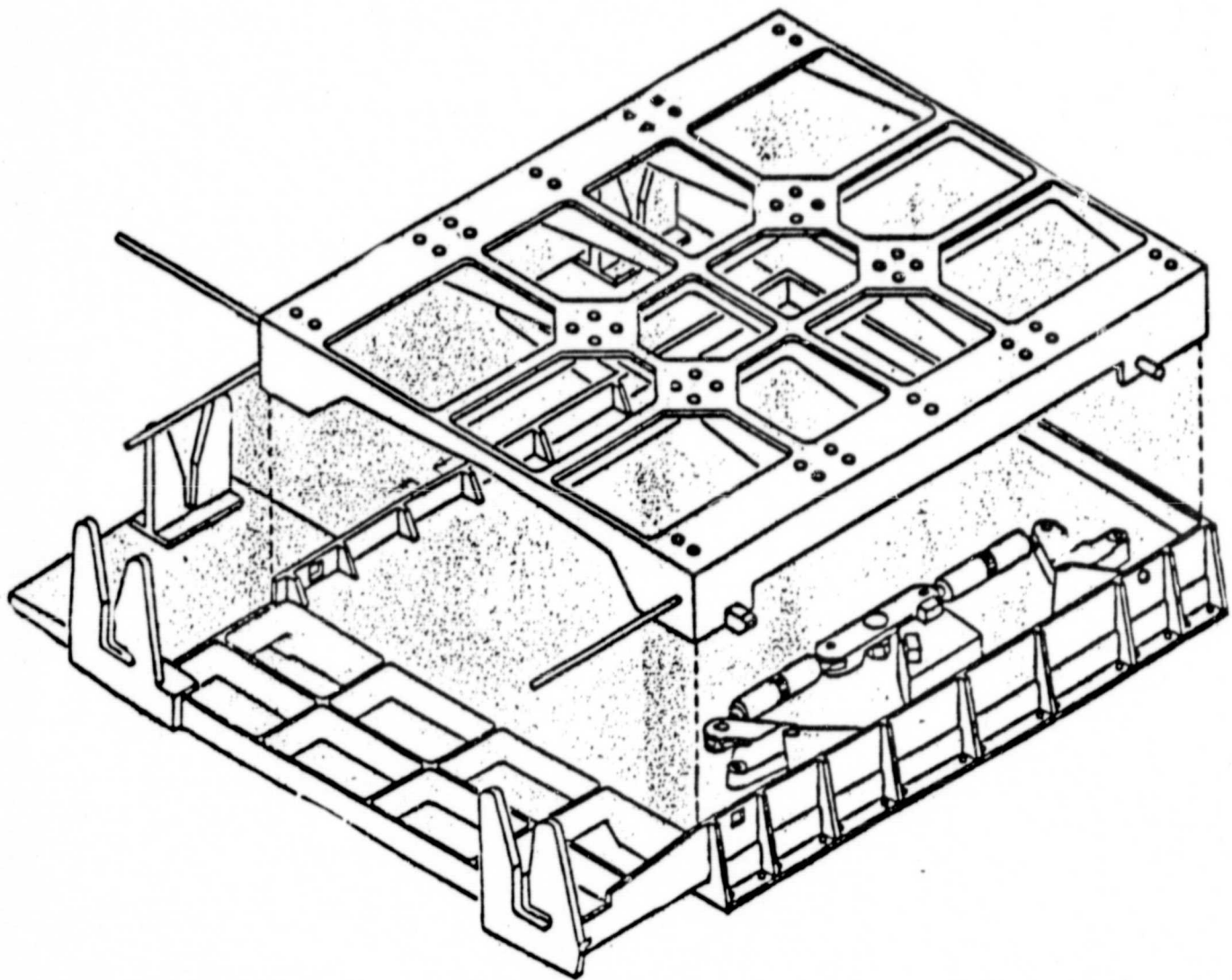


Fig. A-1 Spartan REM

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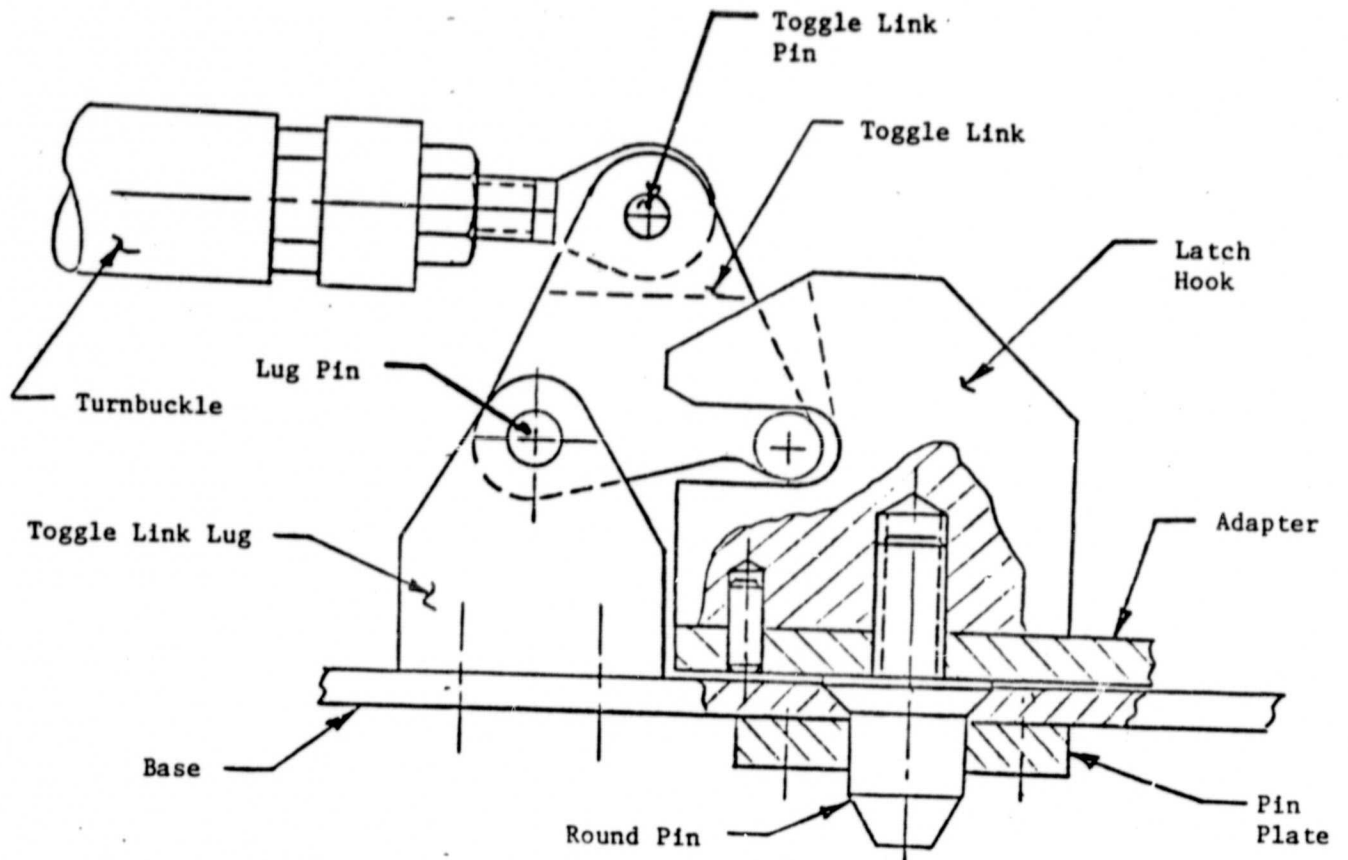


Fig. A-2 Latching Mechanism

Prepared by:	Date	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page	Temp.	Perm.
Checked by:	Date		20		
Approved by:	Date				
Title			Model		
			Report No.		

LOADS

Prepared by: <u>DSM</u>	Date: <u>6/84</u>	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page: <u>2.1</u> Temp. <u>(2.2)</u> Perm.
Checked by:	Date:	Title: <u>SPARTAN REM</u>	Model:
Approved by:	Date:		Report No.:

LOADS (REF ED23-84-24
FEB 16, 1984)

LOW FREQUENCY ACCELERATIONS

\ddot{X}
 ± 4.3

\ddot{Y}
 ± 2.0

\ddot{Z}
 $+7.1/-3.8$

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LOAD COMBINATIONS

CASE.	\ddot{X}	\ddot{Y}	\ddot{Z}
1	4.3	2.0	7.1
2	-4.3	2.0	7.1
3	4.3	-2.0	7.1
4	-4.3	-2.0	7.1
5	4.3	2.0	-3.8
6	-4.3	2.0	-3.8
7	4.3	-2.0	-3.8
8	-4.3	-2.0	-3.8

R. HURFORD (MSFC)

4/9/84

SPARTAN REM

WEIGHT

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ADAPTER = 167 LB
BASE ASSY. = 420 LB

TOTAL 587 LB

CALCULATED CYCLES

BERTHING LOAD 4000 IN-LB TORQUE

LIMIT CYCLES = 100

LIFE FACTOR = 4

TOTAL REQUIRED DESIGN CYCLE LIFE

$100 \times 4 = \underline{400 \text{ CYCLES}}$

MAJOR FLIGHT LOADING CYCLE LIFE

TIME = 50 SEC. + 20 SEC./MISSION

FOUR MISSION DESIGN TIME = 50 + 80 SEC. = 130 SEC.

LIFE FACTOR = 4

USE 50 Hz = NATURAL FREQUENCY

TOTAL REQUIRED DESIGN CYCLE LIFE

$130 \times 4 \times 50 = \underline{26,000 \text{ CYCLES}}$

TABSERP2

SPARTEN REM MASS PROPERTIES

(PRELIMINARY)

0120 SEP 7, 1983

TABLE OF CONTENTS

PAGE(S)

TABLE OF CONTENTS

FIGURE 1 - MASS PROPERTIES COORDINATE SYSTEM, BASE	1
FIGURE 2 - MASS PROPERTIES COORDINATE SYSTEM, ADAPTER	2
TABLE I - MASS PROPERTIES, BASE SUBASSEMBLY (30A60630-1)	3-5
TABLE II - MASS PROPERTIES, ADAPTER SUBASSEMBLY (30A60640-1)	6
TABLE III - MASS PROPERTIES, SPARTEN REM ASSEMBLY (30A60630-1)	7-8
TABLE IV - MASS PROPERTIES, SEPARATED BASE	9-10
TABLE V - MASS PROPERTIES, SEPARATED ADAPTER	11

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FIGURE 1
MASS PROPERTIES COORDINATE SYSTEM
(BASE)

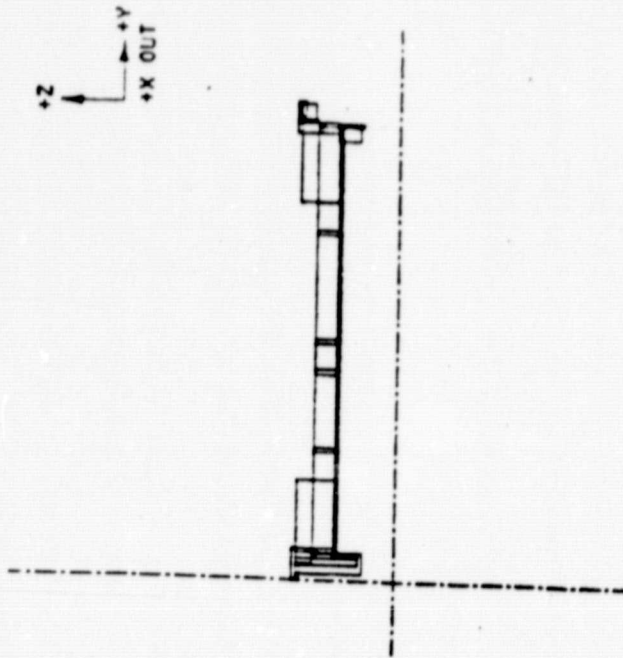
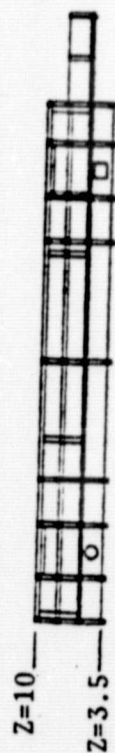
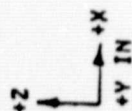
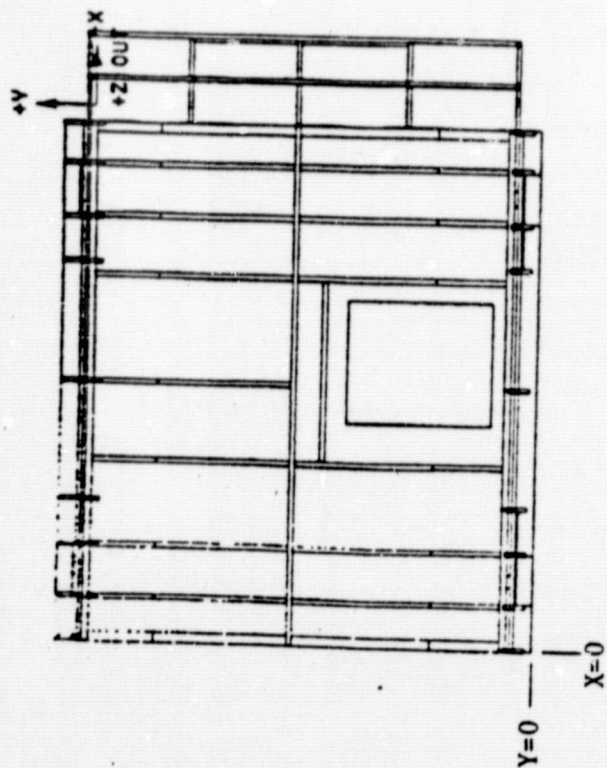
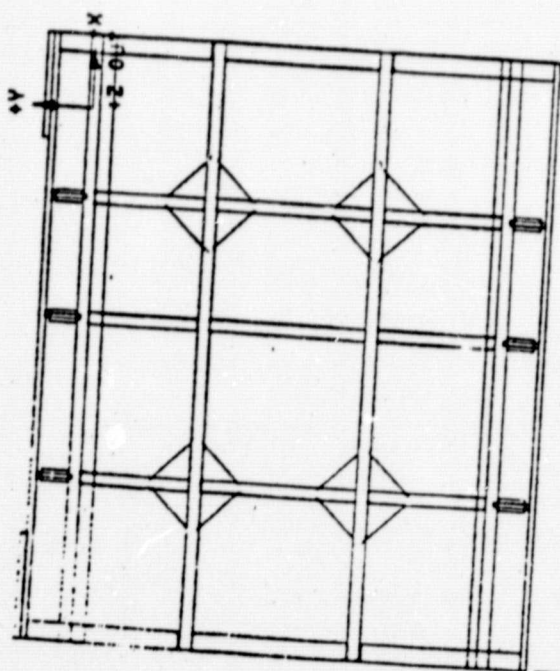


FIGURE 2
MASS PROPERTIES COORDINATE SYSTEM
(ADAPTER)



Y=2.5

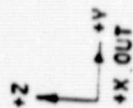
X=0.0



Z=5.681



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BASE SUBASSEMBLY (3060680-1)

9114 SEP 7, 1983

3

		WEIGHT POUNDS	X	Y	Z	IX	Y	12
			STATION-INCHES			POUND-INCHES-SQUARED		
2601	ROSE HOUSING	224.50	23.23	23.13	6.98	584.56	81664.5	35316.7
2602	LINK ROD	5.39	16.29	8.38	4.16	3.32	54.3	15.7
2603	LINK ROD	0.32	13.66	7.71	4.16	0.22	1.52	1.52
2604	ROD END	0.32	22.62	9.27	4.40	0.22	0.22	0.22
2605	ROD END	0.32	19.85	7.73	4.40	0.22	0.22	0.22
2606	NUT 9/16-18 LH	0.17	12.63	7.54	4.40	0.22	0.22	0.22
2607	THREADED PIN	0.17	13.54	8.19	4.40	0.22	0.22	0.22
2608	SPRING BELLVILLE	1.55	21.34	7.49	4.40	2.34	5.77	5.77
2609	LINK HOUSING	0.32	28.89	7.49	4.40	0.22	0.22	0.22
2610	LINK ROD	0.32	40.34	7.49	4.40	0.22	0.22	0.22
2611	ROD END	0.32	29.08	7.49	4.40	0.22	0.22	0.22
2612	NUT 9/16-18 LH	0.17	40.69	7.49	4.40	0.22	0.22	0.22
2613	THREADED PIN	0.17	22.55	7.49	4.40	0.22	0.22	0.22
2614	SPRING BELLVILLE	1.55	10.10	8.29	4.40	2.34	5.77	5.77
2615	BELLCRANK	1.23	11.00	7.71	4.38	1.14	1.33	1.33
2616	LATCH LINK SUPPORT	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2617	LATCH LINK SUPPORT	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2618	LUG PIN	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2619	LATCH LINK PIN	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2620	LATCH LINK PIN	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2621	LATCH LINK PIN	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2622	LATCH LINK PIN	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2623	Y. GUIDE, POS 1	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2624	Y. GUIDE, POS 2	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2625	Y. GUIDE, POS 3	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2626	Y. GUIDE, POS 4	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2627	GUIDE BAR	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2628	SHLF, POSITION 3	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2629	SHLF, POSITION 4	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2630	BKT, HOUSING SU MTG	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2631	MOUNTING DECK	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2632	SUITCH, POS. IND.	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2633	SUITCH, POS. IND.	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2634	SUITCH, POS. IND.	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2635	BELLCRANK SHAFT	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2636	5/16-24X1.25 CAP	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2637	5/16-24 S.L. NUT	0.22	11.00	5.55	4.44	0.22	0.22	0.22
2638	5/16 WASHER, FLAT	0.22	11.00	5.55	4.44	0.22	0.22	0.22

----> TABLE IS CONTINUED, 46 MORE LINES <---

3

BASE SUBASSEMBLY (30A60680-1)

9:15 SEP 7, 1983

		LEIGHT		X		Y		Z		IX		IV		I2	
		POUNDS		STATION-INCHES		STATION-INCHES		POUND-INCHES-SQUARED							
2649	5/16-24X1.25	CR	MS1695-33	58.77	48.27	5.50	1.1	0.0	0.0	1.1	0.0	0.0	0.0	1.1	0.0
2650	5/16-24 S.L. NUT	CR	MS1695-33	58.77	48.27	5.50	1.1	0.0	0.0	1.1	0.0	0.0	0.0	1.1	0.0
2651	5/16-24 S.L. FLAT	CR	MS1695-33	58.77	48.27	5.50	1.1	0.0	0.0	1.1	0.0	0.0	0.0	1.1	0.0
2652	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2653	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2654	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2655	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2656	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2657	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2658	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2659	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2660	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2661	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2662	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2663	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2664	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2665	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2666	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2667	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2668	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2669	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2670	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2671	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2672	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2673	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2674	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2675	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2676	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2677	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2678	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2679	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2680	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2681	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2682	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2683	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2684	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2685	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2686	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2687	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2688	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2689	5/16-24 S.L. NUT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2690	5/16-24 S.L. FLAT	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2691	5/16-24X1.25	CR	MS1695-33	47.65	54.97	5.50	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

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BASE SUBASSEMBLY (30A00680-1)

5:15 SEP 7, 1973

5

			WEIGHT POUNDS	X	STATION-INCHES	Y	Z	IX	POUND-INCHES-SQUARED	IZ
2892	5/16-24X2.13 CAP	2	MS51958-103	CR	0.13	44.30	43.66	6.77	0.0	0.2
2893	5/16-24 S.L. NUT	2	MS21045-CS	A	0.03	44.30	43.66	6.77	0.0	0.2
2894	5/16 WASHER, FLAT	2	MS5755-812	CR	0.01	44.30	43.66	6.77	0.0	0.2
TOTAL			338.56	32.99	23.67	6.18	102969.7	116854.9	216842.6	

725.4

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TABLE II

ADAPTER SUBASSY (30A60640-1)

9:16 SEP 7, 1983

6

		WEIGHT POUNDS	X STATION-INCHES	Y STATION-INCHES	Z	IX POUND-INCHES-SQUARED	IV POUND-INCHES-SQUARED	IZ
105	ADAPTER, SCREW THD	154.00	25.61	24.26	1.18	38765.4	41177.2	7951.5
3545	INSERT, SCREW THD	0.07	25.61	24.26	1.18	9.1	16.0	25.2
3546	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3547	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3548	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3549	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3550	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3551	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3552	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3553	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3554	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3555	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3556	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3557	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3558	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3559	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3560	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3561	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3562	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3563	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3564	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3565	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3566	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3567	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3568	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
3569	ADAPTER, SCREW THD	0.39	25.61	24.26	1.18	9.1	16.0	25.2
TOTAL		169.98	25.60	23.73	1.49	44669.5	46479.5	90381.6

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OF POOR QUALITY

6

ORIGINAL PAGE IS
OF POOR QUALITY

TABLE III

9116 SEP 7, 1983

SPARTEN REM ASSY (30A60630-1)

	WEIGHT POUNDS	X	STATION- INCHES	Z	IX	POUND-INCHES-SQUARED	IV	IX	POUND-INCHES-SQUARED	IZ
104 BASE SUBASSEMBLY	318.56	30.99	23.67	6.18	102559.7	1.6854.9	1.6854.9	21882.6	21882.6	1.6
106 ADAPTER SUBASSEMBLY	169.28	25.63	23.73	1.40	44660.5	1.6854.9	1.6854.9	21882.6	21882.6	1.6
4503 NUT PLATE, ROUND HOLE	1.54	6.88	1.89	8.00	2.9	2.9	2.9	2.9	2.9	1.4
4504 NUT PLATE, SQUARE HL	1.04	4.4	1.80	8.00	2.5	2.5	2.5	2.5	2.5	1.4
4505 NUT PLATE, ROUND HOLE	1.28	6.88	1.89	8.00	2.5	2.5	2.5	2.5	2.5	1.4
4506 NUT PLATE, SQUARE HL	1.09	4.4	1.80	8.00	2.5	2.5	2.5	2.5	2.5	1.4
4507 NUT PLATE, SQUARE HL	1.09	4.4	1.80	8.00	2.5	2.5	2.5	2.5	2.5	1.4
4508 7/16-20X1.50 FL H	0.13	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4509 7/16-20 NUT SL	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4510 7/16-20X1.50 FL H	0.13	4.4	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4511 7/16-20 NUT SL	0.05	4.4	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4512 7/16-20X1.50 FL H	0.13	4.4	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4513 7/16-20 NUT SL	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4514 7/16-20X1.50 FL H	0.13	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4515 7/16-20 NUT SL	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4516 7/16-20X1.50 FL H	0.13	4.4	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4517 7/16-20 NUT SL	0.05	4.4	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4518 7/16-20X1.50 FL H	0.13	4.4	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4519 5/8-18X2-4ELG	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4520 5/8-18 NUT SL	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4521 5/8 FLAT WASHER	0.07	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4522 5/8-18X2-4ELG	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4523 5/8-18 NUT SL	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4524 5/8 FLAT WASHER	0.07	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4525 5/8-18X2-4ELG	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4526 5/8-18 NUT SL	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4527 5/8 FLAT WASHER	0.07	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4528 5/8-18X2-4ELG	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4529 5/8-18 NUT SL	0.05	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4530 5/8 FLAT WASHER	0.07	6.88	1.97	8.00	0.0	0.0	0.0	0.0	0.0	1.4
4531 HOLDER, ROD 4	1.06	3.21	4.64	1.06	2.1	2.1	2.1	2.1	2.1	1.4
4532 ADAPTER, SWITCH BKT	0.61	27.68	4.50	1.06	1.0	1.0	1.0	1.0	1.0	1.4
4533 1/4-28X1.25 FL HD 4	0.08	27.68	4.50	1.06	1.0	1.0	1.0	1.0	1.0	1.4
4534 1/4-28 NUT S.L.	0.04	27.68	4.50	1.06	1.0	1.0	1.0	1.0	1.0	1.4
4535 1/4 FLAT WASHER	0.01	27.68	4.50	1.06	1.0	1.0	1.0	1.0	1.0	1.4
4536 BLANKET, BOTTOM	0.05	28.75	11.93	1.06	2.1	2.1	2.1	2.1	2.1	1.4
4537 ISULATOR-THERMAL	0.05	28.75	11.93	1.06	2.1	2.1	2.1	2.1	2.1	1.4
4538 SPACER, THERMAL	0.05	28.75	11.93	1.06	2.1	2.1	2.1	2.1	2.1	1.4
4539 SPACER, THERMAL	0.05	28.75	11.93	1.06	2.1	2.1	2.1	2.1	2.1	1.4
4540 SPACER, THERMAL	0.05	28.75	11.93	1.06	2.1	2.1	2.1	2.1	2.1	1.4
4541 SPACER, THERMAL	0.05	28.75	11.93	1.06	2.1	2.1	2.1	2.1	2.1	1.4
4542 SPACER, THERMAL	0.05	28.75	11.93	1.06	2.1	2.1	2.1	2.1	2.1	1.4
4543 BLANKET, TOP	0.05	28.75	11.93	1.06	2.1	2.1	2.1	2.1	2.1	1.4

---) TABLE IS CONTINUED, 43 MORE LINES (---)

TABLE III

SPARTEN REM ASSY (30A60630-1)

9116 SEP 7, 1983

WEIGHT POUNDS	X STATION-INCHES	Z STATION-INCHES	IX POUND-INCHES-SQUARED	IV INCHES	IX POUND-INCHES-SQUARED	IV INCHES
4544	SHIELD, INSUL ASSY	30A09124-1	1.30	3.5	1.30	3.5
4545	POWER TRAIN ASSY	30A09125-1	2.4	1.5	2.4	1.5
4546	3/8 24X1.25 HEX B	NAS05560-11	4.4	0.5	4.4	0.5
4547	3/8 NOM FLY USHR	AN05006161	0.07	0.5	0.07	0.5
4548	CLAMP, CABLE	MC1609-24	0.01	0.5	0.01	0.5
4549	SCRAP, 10-32X7	NAS0509-12M	0.02	0.5	0.02	0.5
4550	NUT 10-32 SL	NAS0509-10M	0.02	0.5	0.02	0.5
4551	WASHER, FLAT	AN0509-010M	0.02	0.5	0.02	0.5
4552	PIN, SPRG, SLOT	NAS165622-24	0.02	0.5	0.02	0.5
4553	PIN, SPRG, SLOT	NAS165622-24	0.02	0.5	0.02	0.5
4554	PIN, SPRG, SLOT	NAS165622-24	0.02	0.5	0.02	0.5
4555	PIN, SPRG, SLOT	NAS165622-24	0.02	0.5	0.02	0.5
4556	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4557	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4558	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4559	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4560	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4561	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4562	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4563	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4564	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4565	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4566	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4567	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4568	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4569	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4570	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4571	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4572	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4573	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4574	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4575	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4576	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4577	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4578	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4579	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4580	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4581	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4582	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4583	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4584	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4585	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4586	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4587	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4588	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4589	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4590	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4591	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4592	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4593	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4594	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4595	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4596	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4597	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4598	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4599	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4600	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4601	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4602	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4603	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4604	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4605	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4606	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4607	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4608	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4609	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4610	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4611	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4612	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4613	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4614	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4615	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4616	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4617	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4618	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4619	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4620	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4621	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4622	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4623	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4624	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4625	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4626	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4627	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4628	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4629	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4630	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4631	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4632	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4633	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4634	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4635	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4636	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4637	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4638	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4639	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4640	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4641	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4642	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4643	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4644	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4645	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4646	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4647	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4648	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4649	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4650	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4651	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4652	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4653	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4654	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4655	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4656	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4657	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4658	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4659	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4660	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4661	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4662	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4663	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4664	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4665	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4666	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4667	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4668	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4669	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4670	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4671	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4672	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4673	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4674	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4675	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4676	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4677	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4678	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4679	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4680	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4681	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4682	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4683	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4684	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4685	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4686	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4687	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4688	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4689	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4690	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4691	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4692	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4693	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4694	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4695	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4696	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4697	5/16 24X2.25 FH	NAS05060-105	0.02	0.5	0.02	0.5
4698	5/16 24X2.25 FH	NAS05060-105	0.			

9

TABLE IV

9117 SEP 7, 1983

SEPARATED BASE

WEIGHT POUNDS	X	Y	Z	IX	IV	12
STATION-INCHES	POUND-INCHES-SQUARED					
104 BASE SUBASSEMBLY	318.56	23.67	5.18	102960.7	11854.3	216842.6
4503 NUT PLATE, ROUND HOLE	1.54	1.88	8.00	0.00	0.00	0.00
4504 NUT PLATE, SQUARE HOLE	1.28	4.28	1.88	0.00	0.00	0.00
4505 NUT PLATE, ROUND HOLE	1.28	6.88	8.00	0.00	0.00	0.00
4506 NUT PLATE, SQUARE HOLE	1.28	4.28	1.88	0.00	0.00	0.00
4507 7/16-20X1.50 FL H	0.30	6.88	1.57	0.00	0.00	0.00
4508 7/16-20 NUT SL	0.13	6.88	1.57	0.00	0.00	0.00
4509 7/16-20 NUT SL	0.05	6.88	1.57	0.00	0.00	0.00
4510 7/16-20X1.50 FL H	0.30	4.28	1.88	0.00	0.00	0.00
4511 7/16-20 NUT SL	0.13	4.28	1.88	0.00	0.00	0.00
4512 7/16-20 NUT SL	0.05	4.28	1.88	0.00	0.00	0.00
4513 7/16-20X1.50 FL H	0.30	6.88	1.57	0.00	0.00	0.00
4514 7/16-20 NUT SL	0.13	6.88	1.57	0.00	0.00	0.00
4515 7/16-20 NUT SL	0.05	6.88	1.57	0.00	0.00	0.00
4516 7/16-20X1.50 FL H	0.30	4.28	1.88	0.00	0.00	0.00
4517 7/16-20 NUT SL	0.13	4.28	1.88	0.00	0.00	0.00
4518 7/16-20 NUT SL	0.05	4.28	1.88	0.00	0.00	0.00
4519 5/8-18X2-40LG	0.19	6.78	0.96	0.00	0.00	0.00
4520 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4521 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4522 5/8-18X2-40LG	0.19	6.78	0.96	0.00	0.00	0.00
4523 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4524 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4525 5/8-18X2-40LG	0.19	6.78	0.96	0.00	0.00	0.00
4526 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4527 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4528 5/8-18X2-40LG	0.19	6.78	0.96	0.00	0.00	0.00
4529 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4530 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4531 5/8-18X2-40LG	0.19	6.78	0.96	0.00	0.00	0.00
4532 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4533 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4534 5/8-18X2-40LG	0.19	6.78	0.96	0.00	0.00	0.00
4535 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4536 5/8-18 NUT SL	0.07	6.78	0.96	0.00	0.00	0.00
4537 ISULATOR-THERMAL	0.00	2.88	3.55	0.00	0.00	0.00
4538 ISULATOR-THERMAL	0.00	2.88	3.55	0.00	0.00	0.00
4539 SPACER, THERMAL	0.00	2.88	3.55	0.00	0.00	0.00
4540 SPACER, THERMAL	0.00	2.88	3.55	0.00	0.00	0.00
4541 SPACER, THERMAL	0.00	2.88	3.55	0.00	0.00	0.00
4542 SPACER, THERMAL	0.00	2.88	3.55	0.00	0.00	0.00
4543 SHIELD, TOP	0.00	2.88	3.55	0.00	0.00	0.00
4544 SHIELD, INSUL ASSY	0.00	2.88	3.55	0.00	0.00	0.00
4545 POWER TRAIN ASSY	0.00	2.88	3.55	0.00	0.00	0.00
4546 CLAMP, CABLE	0.00	2.88	3.55	0.00	0.00	0.00
4547 SCREW 10-32X7	0.00	2.88	3.55	0.00	0.00	0.00
4548 NUT 10-32 SL	0.00	2.88	3.55	0.00	0.00	0.00
4549 WASHER, FLAT	0.00	2.88	3.55	0.00	0.00	0.00
4550 WASHER, FLAT	0.00	2.88	3.55	0.00	0.00	0.00
4551 WASHER, FLAT	0.00	2.88	3.55	0.00	0.00	0.00

---> TABLE IS CONTINUED, 7 MORE LINES <---

9

TABLE IV

9117 SEP 7, 1983

SEPARATED BASE

		WEIGHT POUNDS	X STATION-INCHES	Y STATION-INCHES	Z	IX PCUND-INCHES-SQUARED	IV PCUND-INCHES-SQUARED	IZ
4580 BRACKET, CONN.	CR	0.31	19.12	43.70	4.12	0.2	0.7	0.5
4581 120 PIN CONN-FEMALE	E	0.20	18.92	43.55	2.93	0.0	0.0	0.0
4582 120 PIN CONN-MALE	E	0.20	18.92	45.00	2.93	0.0	0.0	0.0
4583 10-32X.88 CRES FM	CR	0.03	19.12	43.70	4.12	0.0	0.0	0.0
4584 10-32 NUT S.L.	A	0.02	19.12	43.70	4.12	0.0	0.0	0.0
4585 #10 FLAT WASHER	A	0.01	19.12	43.70	4.12	0.0	0.0	0.0
4586 CABLE UI	E	0.50	24.70	23.40	2.93	70.1	2.0	171.7
TOTAL		382.01	30.40	22.07	6.34	115398.0	122466.3	234539.9

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SEPARATED ADAPTER

11

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MATERIAL PROPERTIES

3.1 2219 AL
 3.3 Inconel 718
 3.5 4340 STFFL
 ↓ MP35N
 416 STAINLESS

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2219 ALUMINUM ALLOY SHEET AND PLATE (BARE)
Specifications QQ-A-250/30
SUMMARY OF ROOM TEMPERATURE PROPERTIES

Basis	99% Minimum					Prelim ^a
Condition	T87					
Material Thickness (in.)	0.020-0.249	0.250-3.000	3.001-4.000	4.001-5.000	5.001-6.000	
F_{tu} (ksi) L	63	63	61	60	58	
LT	64	64	62	61	59	
ST	-	-	-	-	-	
F_{ty} (ksi) L	51	51	50	49	47	
LT	52	51	50	49	47	
ST	-	-	-	-	-	
F_{cy} (ksi) L	52	51	50	49	47	
LT	55	54	50	49	47	
ST	-	-	-	-	-	
F_{su} (ksi)	37	37	36	35	34	
F_{bru} (ksi) ^b						
e/D = 1.5	91	91	88	86	84	
e/D = 2.0	113	113	109	107	104	
F_{bry} (ksi) ^b						
e/D = 1.5	71	71	70	69	66	
e/D = 2.0	86	86	84	83	79	
e (percent) L						
LT	5	6	4	3	3	
ST	-	-	-	-	-	
Physical Properties (average)						
E (10^6 psi)	10.5					
E_c (10^6 psi)	10.8					
G (10^6 psi)	4.0					
ν	0.33					
K (BTU-in./hr-sq in.-°F)	6.2					
α (10^{-6} in./in.-°F)	12.4 (70 F to 212 F, see p. 02.24.02.01)					
C (BTU/lb-°F)	0.23 (at 212 F)					
ρ (lb/cu.in.)	0.102					

^a Properties shown for 5.001-6.000 inches thick are preliminary, pending negotiation with suppliers.

^b Bearing allowables include the lubricated pin reduction factor. When the hole axis is parallel with the plane of the plate, F_{bru} values must be adjusted per p. 01.00.00.11.

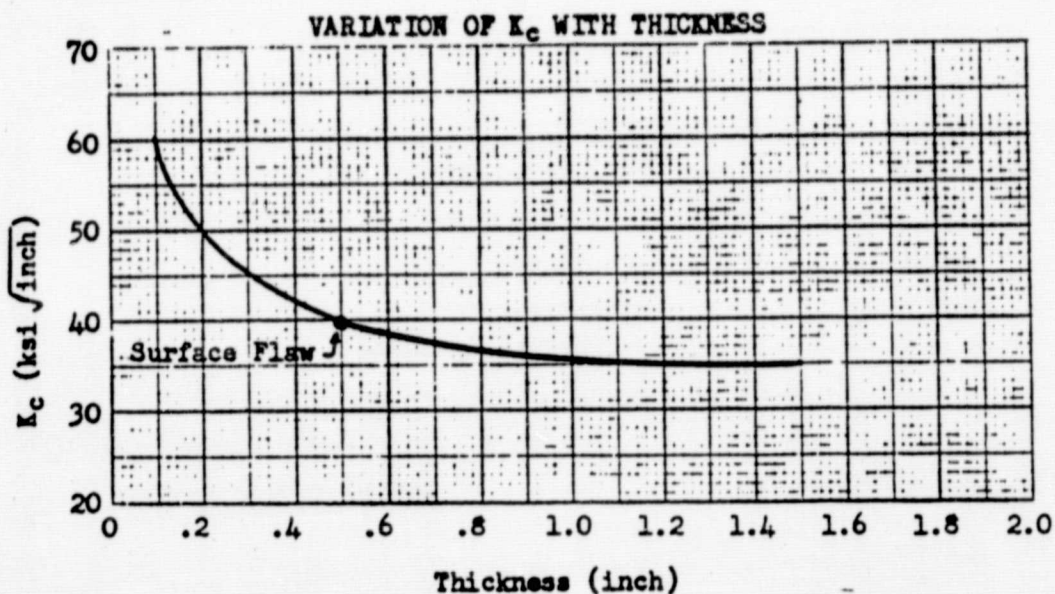
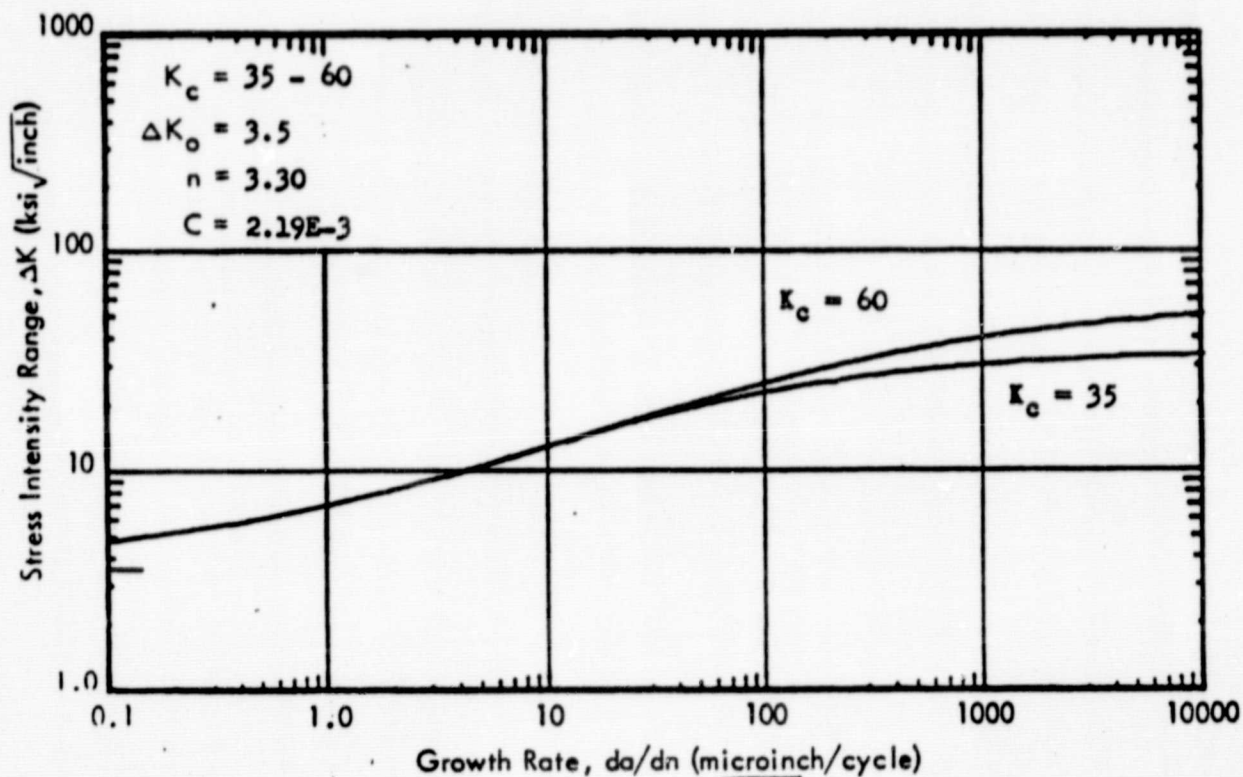
REF ROCKWELL
MATERIALS
MAXXAL
P 2.24.01.02

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**2219-T87 Aluminum Alloy Sheet and Plate, 0.10 to 1.5 Inch Thick
TL Orientation**

CYCLIC CRACK GROWTH RATE AT ROOM TEMPERATURE



Nomenclature for growth rate constants is given on page 01 00 00 00

REF. ROCKWELL MATL MANUAL

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Inconel 718 Sheet, Plate, Bar, and Forgings
Specifications MB0170-071, -074, -075, 076, and MA011-303
AMS 5596, 5597, 5663

SUMMARY OF ROOM TEMPERATURE PROPERTIES
99 Percent Probability Minimum Values

Form	All	Sheet and Plate	Bar and Forgings
Condition	Annealed	Solution Treated and Aged (STA)	
Material Thickness (in.)	≤0.500 Inch		
F_{tu} (ksi) L	110	180	185
LT	110	180	180
F_{ty} (ksi) L	50	150	150
LT	50	150	150
F_{cy} (ksi) L	50	150	150
LT	50	150	150
F_{su} (ksi)	70	113	113
F_{bru} (ksi)			
e/D = 1.5		248	248
e/D = 2.0	200	320	320
F_{bry} (ksi)			
e/D = 1.5		210	210
e/D = 2.0	85	240	240
e (percent) L	30	12	12
LT			6 (Bar); 10 (Forgings)
Physical Properties (average)			
E (10^6 psi)		29.5	
E_c (10^6 psi)		29.5	
G (10^6 psi)		11.4	
ν		0.29	
K (BTU-in./hr-sq in.-°F)		0.53 (STA, 68 F)	
α (10^{-6} in./in.-°F)		7.2 (68 F to 200 F)	
C (BTU/lb-°F)		0.104	
ρ (lb/cu.in.)		0.297	

REF. ROCKWELL MATL MANUAL
P. 7.02.01.01

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CYCLIC CRACK GROWTH RATE DATA

MATERIAL: 403

INCONEL 718 STA, LTS AND TLS ORIENTATIONS,
ROOM TEMPERATURE

YS (0.2% O.S.) KSI 150.0

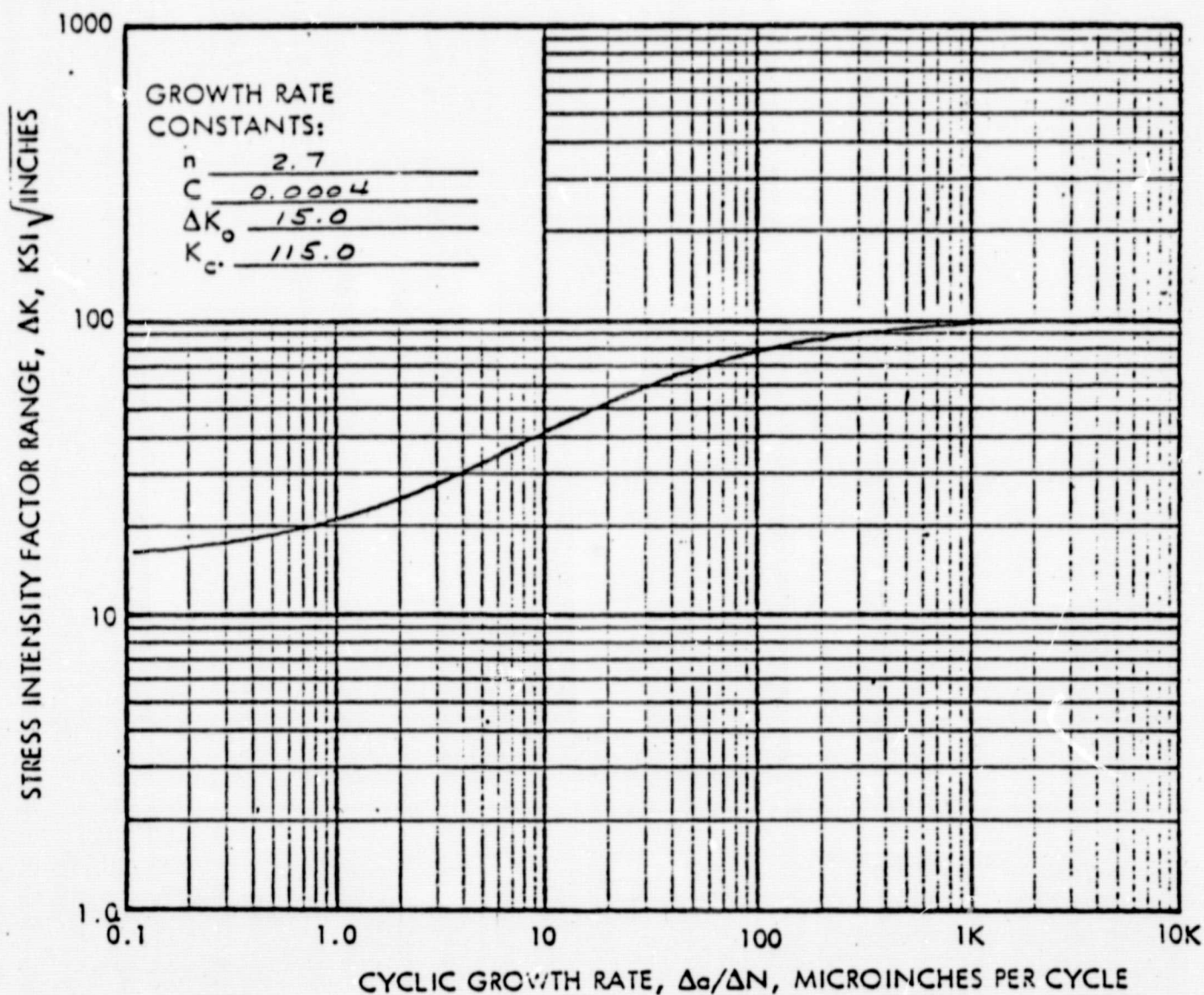


DIAGRAM X'ED) (CONTACT FRACTURE MECHANICS DATA

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- ① 4340 PLATE 180 KSI MIN
- ② INCOEL 718 PLATE 180 KSI MIN
- ③ MP35N STAINLESS ROD 180 KSI MIN
- ④ 416 STAINLESS 120 KSI

NEQ-1 (ELLIPSE) EQUATION PARAMETERS

	4340	INCOEL 718	MP35N	416 SS
(4) ΔK (KSI $\sqrt{\text{INCH}}$)	6.3	7.2	5.5	5.5
(2) N	2.38	4.00	3.40	4.00
(1) C	2.36×10^{-9}	1.39×10^{-9}	8.88×10^{-11}	2.333×10^{-10}
(3) K_{IC} (KSI $\sqrt{\text{INCH}}$)	80	80	100	40
ΔK	KSI $\sqrt{\text{INCH}}$	KSI $\sqrt{\text{INCH}}$	KSI $\sqrt{\text{INCH}}$	KSI $\sqrt{\text{INCH}}$
$\frac{da}{dN}$ crack growth rate	INCH / CYCLE	INCH / CYCLE	INCH / CYCLE	INCH / CYCLE

REPARE (MSFC)
 11/22/83

$$\frac{da}{dN} = C (\Delta K)^N$$

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SPARTAN REM STRESS ANALYSIS

Prepared by: DDM	Date: 6/84	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page: 4.0.1
Checked by: ESW	Date: 6/84	Title: SPARTAN REM	Model:
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MARGIN SUMMARY

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ITEM	LOAD CASE	M.S.	PAGE (SEC 4.0)	COMMENTS
TOGGLE LINK (30A60695)	+/-/+	3.21 1.95 5.9	12 14.1 15.1	TRANSVERSE LOAD ON LUG (TURNBUCKLE CONNECTION) SHEAR ON LATCHING POST TRANSVERSE LOAD ON LUG (SUPPORT)
TOGGLE LINK LUG (30A60694)	+/-/+	2.51 .38 .71	4.0 5.1 9.0	TRANSVERSE LOAD ON LUG BENDING INTERACTION
LUG PIN (30A60678)	+/-/+	2.39 1.05	12 15.2	SHEAR BENDING
TOGGLE LINK PIN (30A60700)	+/-/+	1.22 .53	13 14.0	SHEAR BENDING
TURNBUCKLE (30A60699)	+/-/+	1.81 2.15	16 17	BEARING TENSION ON THREADS
BELL CRANK (30A60696)	TORQUE 1521 in-lb	.6 .42 .85 .09 3.3	18 20 20 21 23	SHEAR ON SPLINE LIMIT TORQUE ON SHAFT TORSION ON SHAFT COMPRESSIVE STRESS ON SHAFT LUG TENSION
ROUND PIN (30A60642) SQUARE PIN 30A60643	+/-/+	.78 ↓	24 ↓	SHANK SHEAR ↓
ROUND PIN PLATE (30A60645) MOUNTING BOLTS (NAB600)	+/-/+	.8 1.25 1.92 1.30	30 31 32 30	SHEAR TEAR OUT SHEAR BEARING BOLT SHEAR
SQUARE PIN PL (30A60644)	+/-/+	.81 .37	32 33	SHEAR TEAR OUT BENDING

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Checked by: EJW	Date: 6/84	Title: SPARTAN REII	Model:
Approved by:	Date:		Report No.:

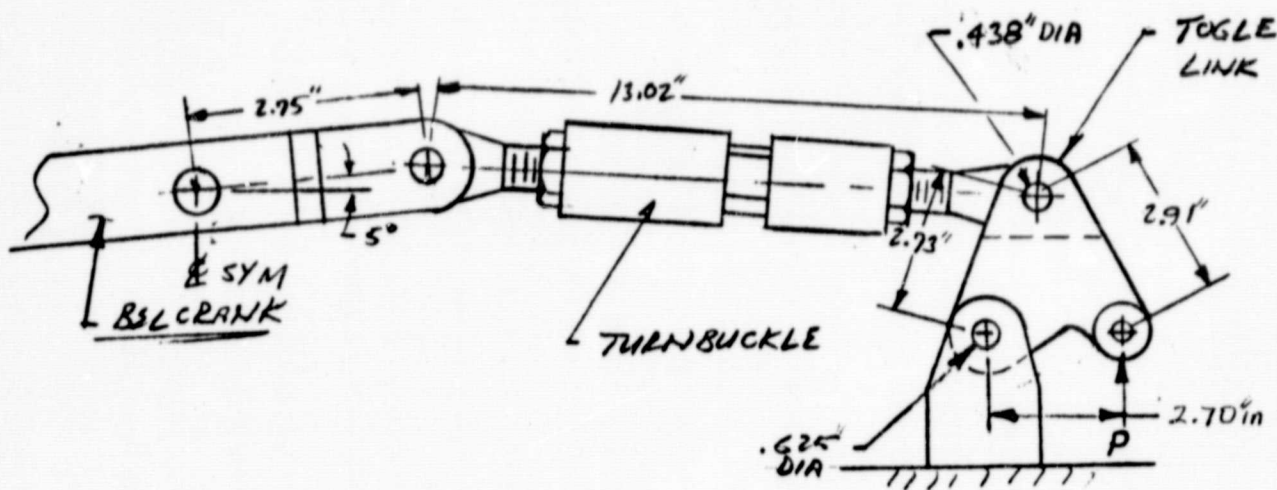
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ITEM	LOAD CASE	M.S.	PAGE	COMMENTS
RDD HOLDER SCREWS (30AL6048)	MAY RED. LOAD 1560/13	.39	36	INTERACTION
30AL6048	DOOR SPRINT	.33	38	SHEAR
BASE MOUNTING BOLTS	+/- /+	.96	39	INTERACTION
BASE/PD11 BOLTS NUTS	+ -/+	.63 .60	40 40	INTERACTION TENSION
ADAPTER	-/- /+	2.15	45	TENSION
BASE	+ /+ /+	1.29	45	TENSION
ADAPTER/PD11 INTERFACE	+ /+ /+	.798	25	SHEAR TEAROUT
PIN BEARING ON CRITICAL SHAP	+ /- /+	LARGE	27.1	
HOLE TO BE DRILLED IN ADAPTER	+ /+ /+	7.28	49	TENSION
LOCATOR ROD (30AL6048)	1560 lb BEARING	.015	38.1	

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LATCHING MECHANISM



PAYLOAD + FRAME WT = 2500lb
2.0 G - Y AXIS ACC.

LATCH MECHANISM REACTS TOTAL Y AXIS LOAD

MAY P = 5.19 kips (ref. p 51 PIN REACTION SUMMARY)

LINKAGE FORCES

$$F_T = \frac{P(2.70)}{2.45} = 5.72 \text{ kip}$$

$$R_y = -F = -5.72$$

$$R_y = -P + F_T \sin 1.05^\circ$$

$$F_T = \text{TURNBUCKLE FORCE}$$

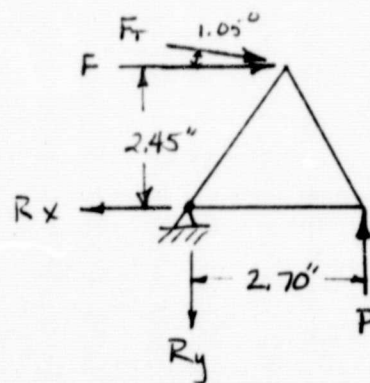
$$= F / \cos 1.05^\circ$$

$$= \frac{5.72}{\cos 1.05^\circ} = 5.729 \text{ kips}$$

$$R_y = -5.19 + 5.72 \sin 1.05^\circ$$

$$= -5.19 + .105$$

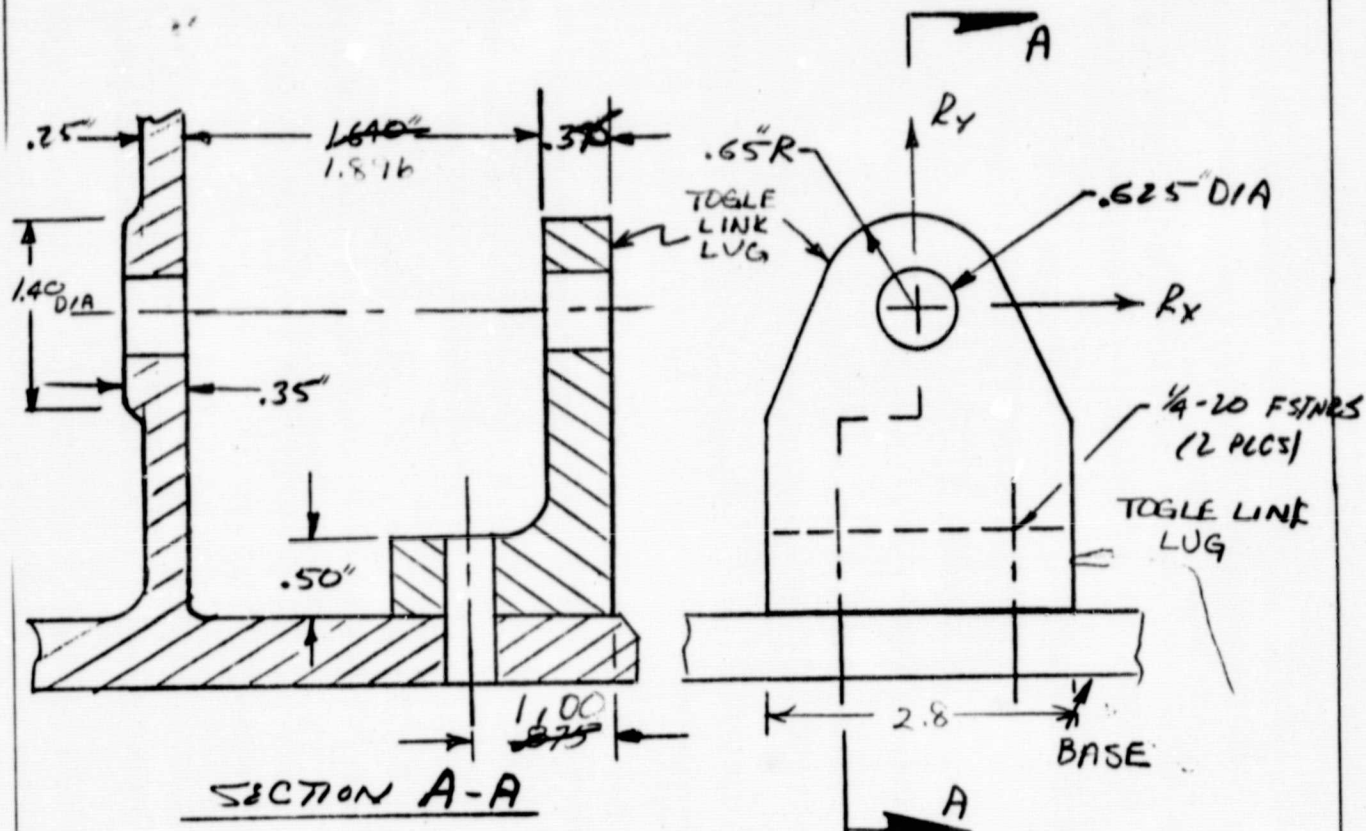
$$R_y = -5.09 \text{ kip}$$



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LATCHING MECHANISM

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MATL - INC 718
 $F_{tu} = 180 \text{ KSI}$
 $F_{su} = 113 \text{ KSI}$

TOGGLE LINK LUG (3DA60694)
 (LOADING FROM NASTRAN RUN N0003111 - App. I
 SUMMARY P. 51.)

MAY Y AXIS PIN REACTION - $R_y = 5.19 \text{ Kips}$

Resulting REACTION - from p 1.0

$$R_x = 5.72 \text{ kip}$$

$$R_y = 5.09 \text{ kip}$$

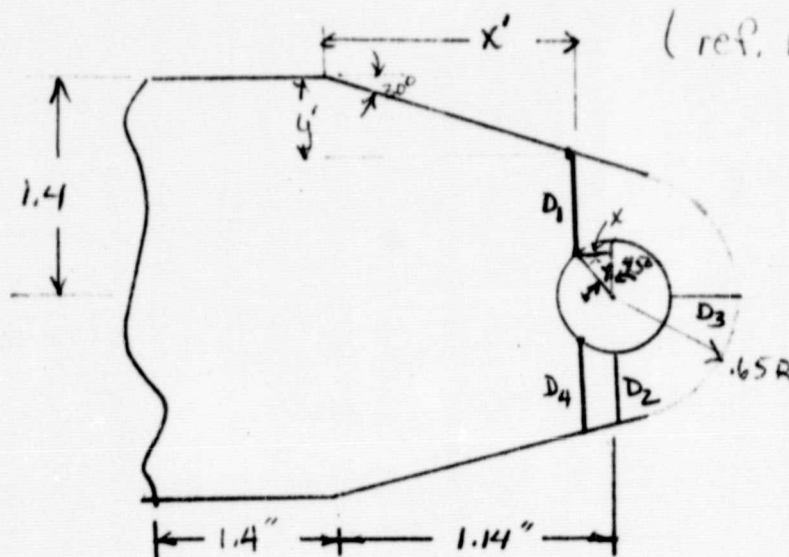
$$\text{EFFECTIVE LUG WIDTH @ HOLE} = 2(.65") = 1.30 \text{ IN}$$

Prepared by: DSM	Date: 6/84	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page: 3.0	Temp: 4.3
Checked by: EJV	Date: 6/84	Title: SPARTAN REM	Model:	
Approved by:	Date:		Report No.:	

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LATCHING MECHANISM - TOGGLE LINK LUG TRANSVERSE LOADS

ref. fig p2



(ref. AFSC STRUCTURES MANUAL
B2)

$$DIA = .625"$$

$$D_3 = .65 - \frac{.625}{2} = .3375$$

$$D_2 \approx D_3 = .3375$$

$$A_{br} = D t = .625(.375) = .234$$

$$Y = (\cos 45^\circ) \cdot .3125 = .221$$

$$\frac{X}{Y} = \tan 45^\circ \Rightarrow X = Y = .221$$

$$X' = 1.14 - X = .919"$$

$$Y' = X' \tan 30^\circ = .919(.577) = .531"$$

$$D_1 = D_4 = 1.4 - Y - Y' = 1.4 - .221 - .531 = .648 \text{ in}$$

$$t = .375"$$

$$A_1 = D_1 t = .648(.375) = .243 \text{ in}^2$$

$$A_2 = A_3 = D_2(t) = .3375(.375) = .1266 \text{ in}^2$$

$$A_4 = D_4 t = D_1 t = .243 \text{ in}^2$$

$$\Rightarrow \frac{A_{AV}}{A_{br}} = \frac{.186}{.234}$$

$$A_V = \frac{6}{\frac{3}{A_1} + \frac{1}{A_2} + \frac{1}{A_3} + \frac{1}{A_4}} = \frac{6}{\frac{4}{A_1} + \frac{2}{A_2}} = .186$$

$$= .795$$

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LATCHING MECHANISM (cont.)

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TOGGLE LINK LUG (3DA60674)

from p. 3.0 $\frac{A_{10U}}{A_{10L}} = .795 \text{ in}^2, A_{10L} = .234 \text{ in}^2$

SHEAR BEARING

$$P_{tru} = K_{tru} A_{tr} F_{tu}$$

from graph on p. 4.1 $K_{tru} = .59$

$$P_{tru} = .59 (.234) (180000) = 24.85 \text{ kips}$$

AXIAL

$$e/D = 1.04, W/D = \frac{1.3}{.625} = 2.08, \frac{D}{t} = \frac{.625}{.375} = 1.67, A_t = \frac{W-D}{t} = 1.8$$

from graph p. 4.2, $K_{br} = .86$

SHEAR BEARING

$$P_{tu} = K_{br} A_{br} F_{tu} = .86 (.234) (180000) = 36.22 \text{ kips}$$

COMBINED AXIAL & TRANSVERSE LOADING

LOADING ON LUG - (p. 2.0)

$$R_x' = 5.72 / 2 = 2.86 \text{ kips}$$

$$R_y' = 5.09 / 2 = 2.55 \text{ kips}$$

$$R_A = \frac{2.86(2)}{36.22} = .158$$

$$R_{tr} = \frac{2.55(2)}{24.85} = .21$$

$$M.S.U = \frac{1}{(R_A^{1.6} + R_{tr}^{1.6})^{.625}} - 1$$

$$= \frac{1}{.0522 + .082} - 1 = 2.51 \text{ ULT SHEAR BEARING}$$

4,4.1

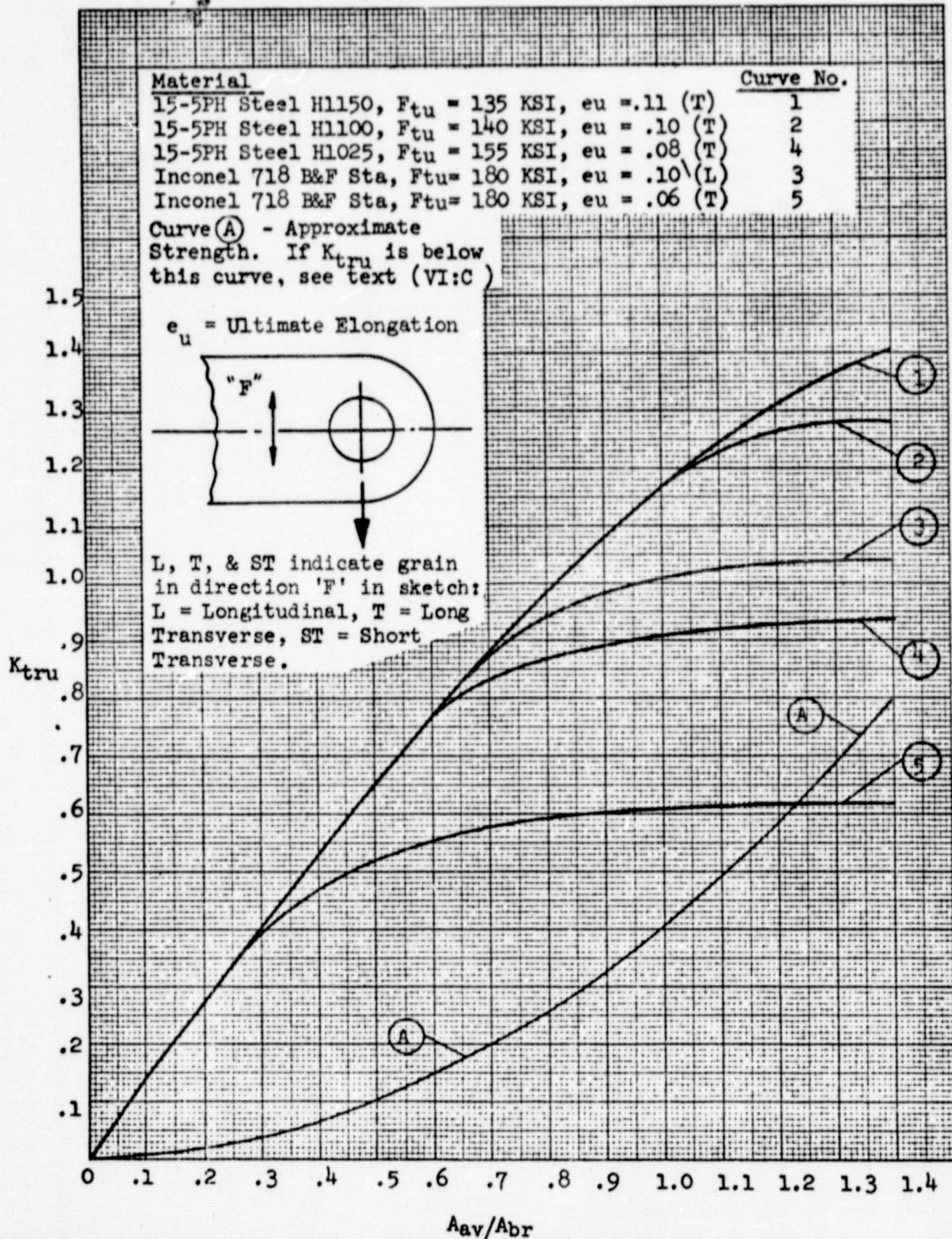
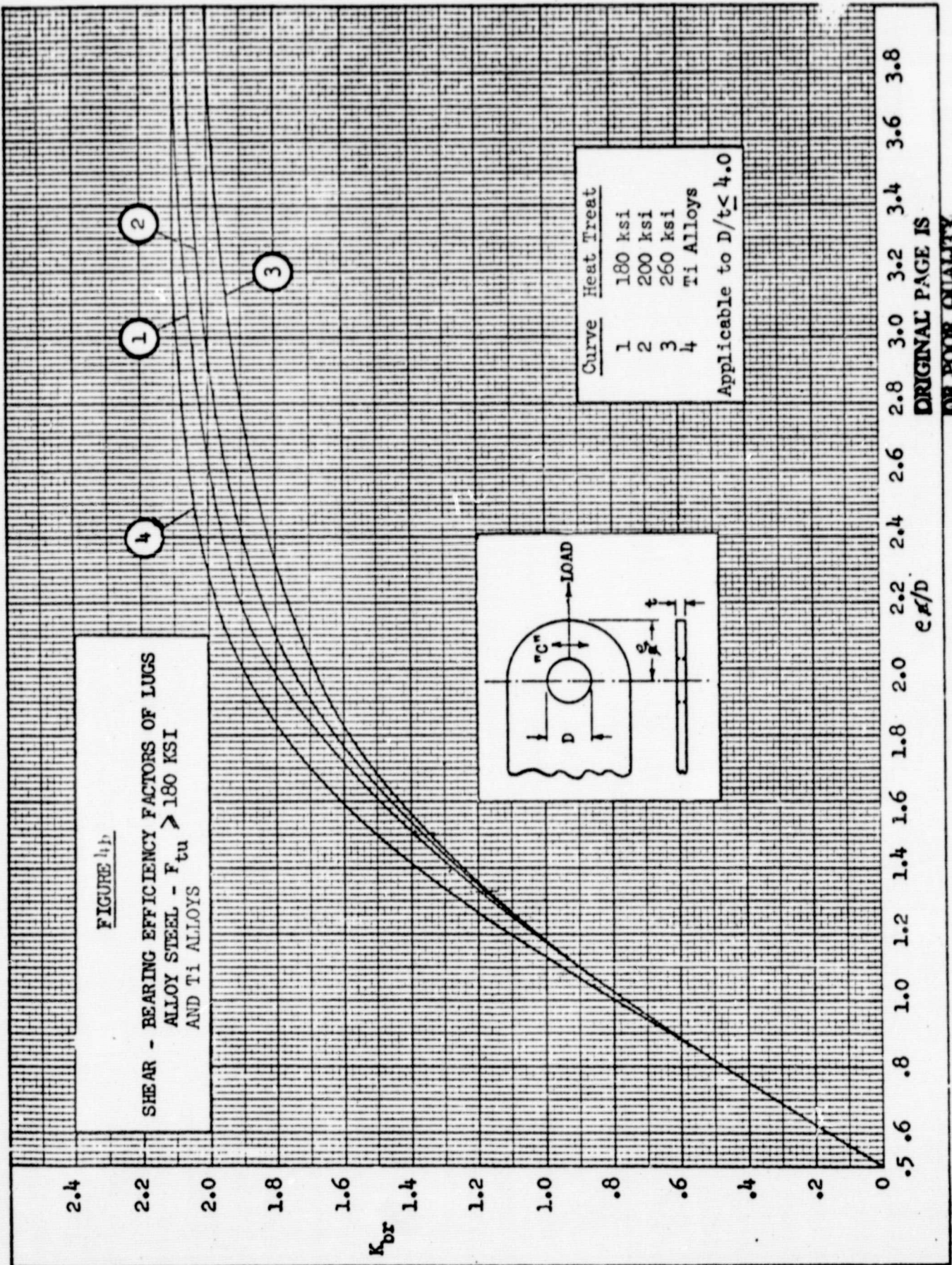


FIGURE 7e EFFICIENCY FACTORS OF LUGS FOR TRANSVERSE LOADING, ALLOY STEEL AND HEAT RESISTANT ALLOYS



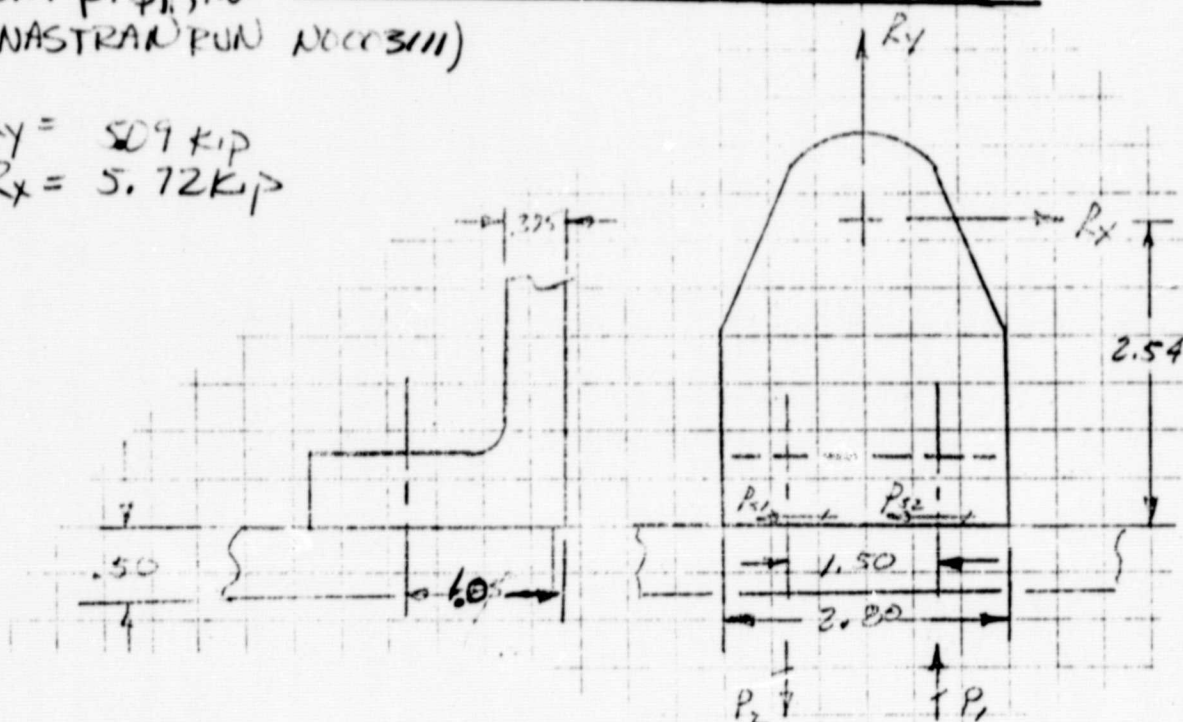
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LATCHING MECHANISM (cont.)

FROM P. 51, 110 TOGGLE LINK LUG INTERFACE
(NASTRAN RUN NO003111)

$$R_y = 509 \text{ kip}$$

$$R_x = 5.72 \text{ kip}$$



@BASE INTERFACE

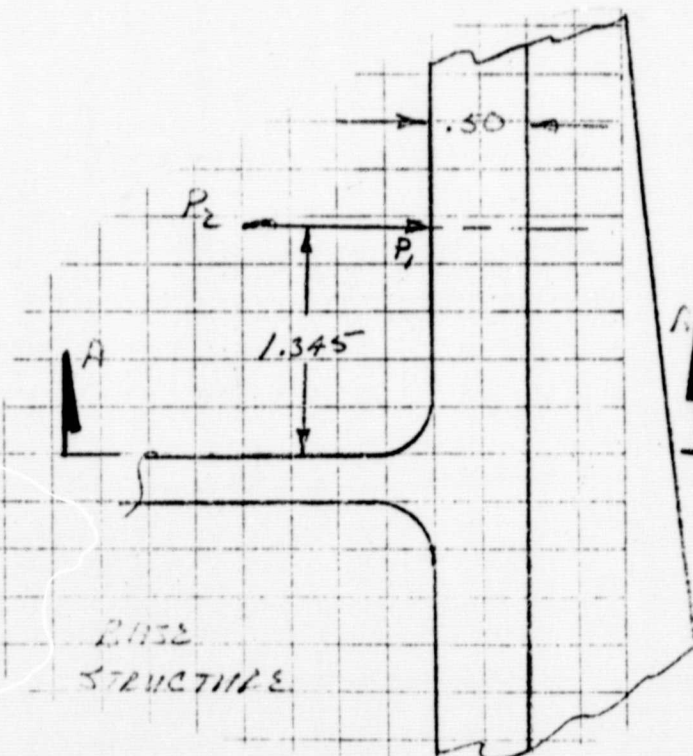
$$P_{s1} = P_{s2} = \frac{5.72}{2} = 2.86 \text{ kip}$$

$$P_2 = \frac{5.72(2.54) + 5.09}{1.5}$$

$$P_2 = 12.23 \text{ kip}$$

$$P_1 = -5.72 \frac{2.54}{1.5} + \frac{5.09}{2}$$

$$P_1 = -7.14 \text{ kip}$$



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BASE
STRUCTURE

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LATCHING MECHANISM (cont.)

LUG BENDING @ BASE

SECTION A-A

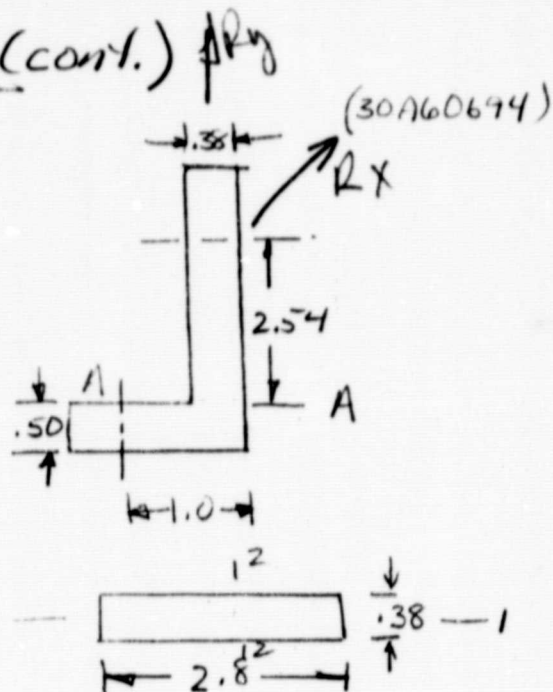
$$A = 2.8(.38) = 1.064 \text{ in}^2$$

$$I_1 = \frac{bh^3}{12} = \frac{2.8(.38)^3}{12} = .0128 \text{ in}^4$$

$$I_2 = \frac{.38(2.8)^3}{12} = .695 \text{ in}^4$$

$$I_1/c = \frac{.0128}{.19} = .067 \text{ in}^3$$

$$I_2/c = \frac{.695}{1.4} = .496 \text{ in}^3$$



$$\begin{aligned} R_x &= 5.72 \text{ kip} \\ R_y &= 5.09 \text{ kip} \end{aligned} \quad \left. \begin{array}{l} \\ \end{array} \right\} \text{ref p. 110}$$

$$M_{11} = R_y \left[(1.0) - \frac{.38}{2} \right] = 5.09(.81) = 4.12 \text{ in-kip}$$

$$M_{22} = R_{xx}(2.54) = 14.53 \text{ in-kip}$$

$$S_{b1} = \frac{M_{11}}{I_1/c} = \frac{4.12}{.067} = 61.49 \text{ ksi}$$

$$S_{b2} = \frac{14.53}{.496} = 29.29 \text{ ksi}$$

FROM MSFC STRUC.
MANUAL, 64
K = 1.5

$$S_{b \text{ MAX}} = 61.49 + 29.29 = 90.78 \text{ ksi}$$

$$S_t = \frac{R_y}{A} = \frac{5.09}{1.064} = 4.78 \text{ ksi}$$

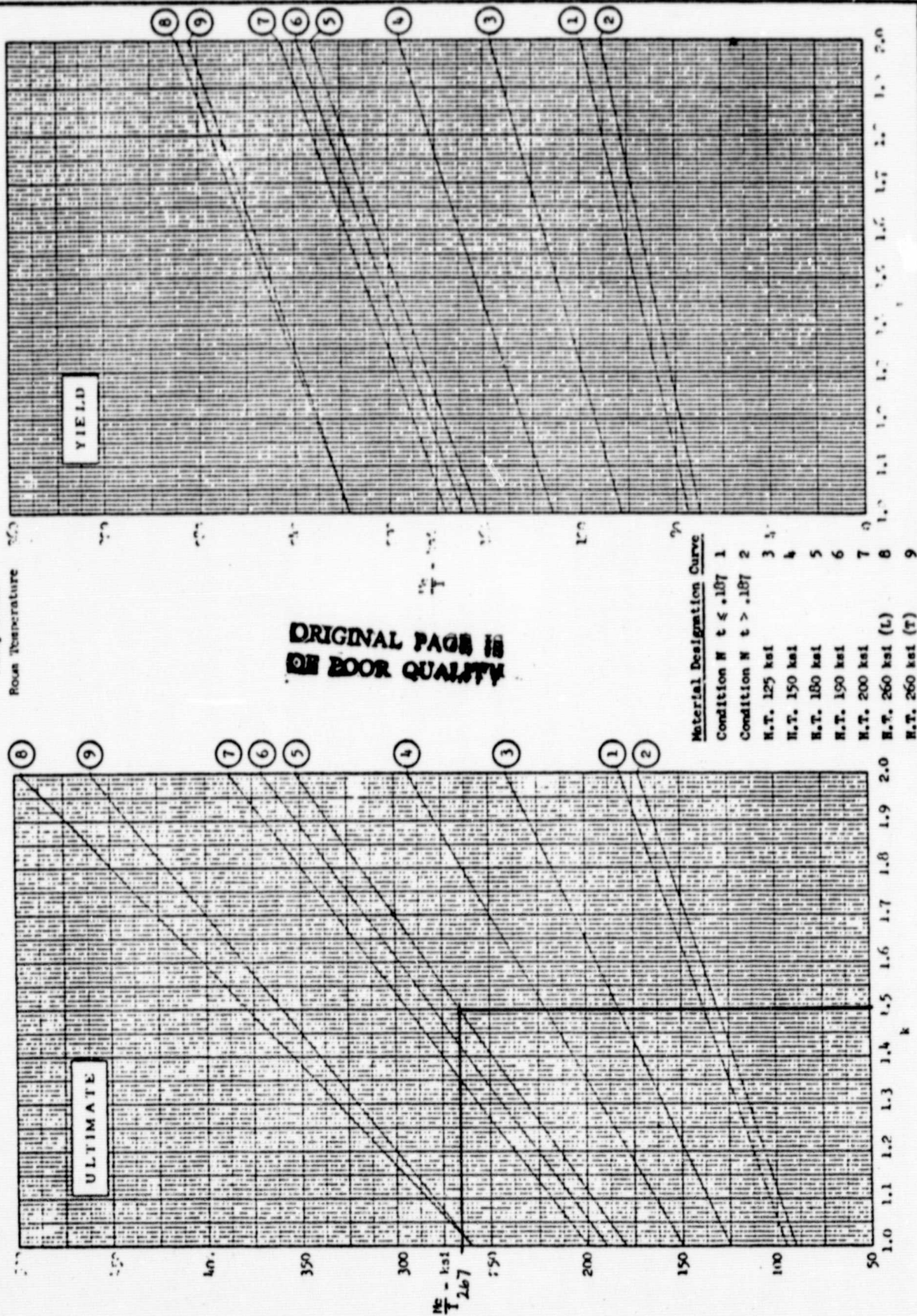
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$$F_{bu} = 270 \text{ ksi} \quad R_b = \frac{(2)(90.78)}{270} = .67, \quad R_t = \frac{2(4.78)}{180} = .053$$

$$M.S.U. = \frac{1}{R_b + R_t} - 1 = .38$$

4.6.1

FIGURE 5b
Ultimate and Yield Values of $\frac{F_y}{T}$ for Symmetrical Sections
Alloy Steel
Room Temperature



5.1

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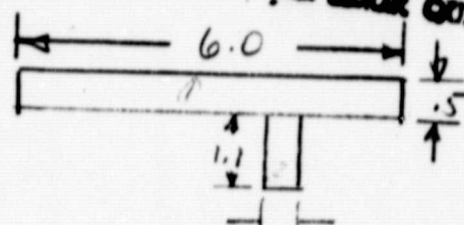
LATCHING MECHANISM (cont.)

NO	A	V	Ay	Ay ²	I _y
1	3.0	.25	.15	.187	.062
2	4.2	1.05	.441	.463	.042
	3.42		1.191	.65	.114

$$\bar{y} = .348 \text{ in} \quad C_{MAX} = 1.252$$

$$I_y = .350 \text{ in}^4$$

$$r_{C_{MAX}} = .279 \text{ in}$$



SECTION A-A
(from sketch on previous page)

$$M_{AA} = (P_2 - P_1) 1.345 = 5.09 (1.345)$$

$$= 6.85 \text{ in-lb}$$

$$f_b = \frac{6.85}{.279} = 24.55 \text{ KSI}$$

$$F_{bu} = 1.5 \times 63 = 94.5 \text{ KSI}$$

$$M.S.U. = \frac{94.5}{2(24.55)} - 1 = .92$$

ACCOUNTING BOLT LOADS

from NASTRAN RUN N0003111, P1

$$R_x = 5.72 \text{ KIP}$$

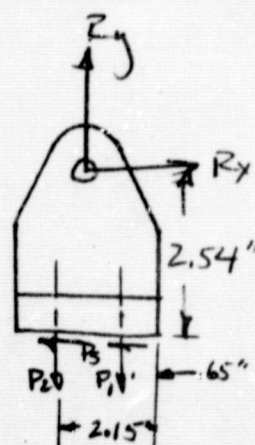
$$R_y = 5.01 \text{ KIP}$$

DUE TO R_x LOAD

$$P_2(2.15) + .65 P_1 = 5.72 (2.54)$$

$$P_1 = \frac{.65}{2.15} P_2$$

$$2.15 P_2 + \frac{.65^2}{2.15} P_2 = 14.53 \text{ K}$$



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LITCHING MECHANISM (cont.)

LUG MOUNTING BOLTS (cont.)

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$$2.346 P_2 = 14.53$$

$$P_2 = 6.19 \text{ kip}$$

$$P_{1T} = \frac{.65}{2.15} (6.19) = 1.87 \text{ kip}$$

Due to Ry LANDING (rel. Ry. in previous page)

$$P_{1T} - P_2 = \frac{5.07}{2} = 2.545 \text{ kip}$$

$$P_{2T} = 2.545 + 6.19 = 8.735 \text{ kip}$$

$$P_{1T} = 2.545 + 1.87 = 4.415 \text{ kip}$$

BOLTS

1/16 DIA NAS 1951C

$$P_{tu} = 23.2 \text{ kip}$$

$$P_{su_{ss}} = 16.25 \text{ kip}$$

SINCE BOLTS ARE TESTED ITEMS-F.S. = 1.4

$$M.S.U. = \frac{23.2}{1.4 (8.735)} - 1 = .897 \quad \text{TENSION}$$

$$\text{BOLT SHEAR} = \frac{5.72}{2} = 2.86 \text{ kip/BOLT}$$

$$M.S.U. = \frac{16.25}{1.4 (2.86)} - 1 = 3.05 \quad \text{SHEAR}$$

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LATCHING MECHANISM (CONT.)

TOGGLE LINK LUG INTERFACE (CONT.)

BOLT INTERACTION -

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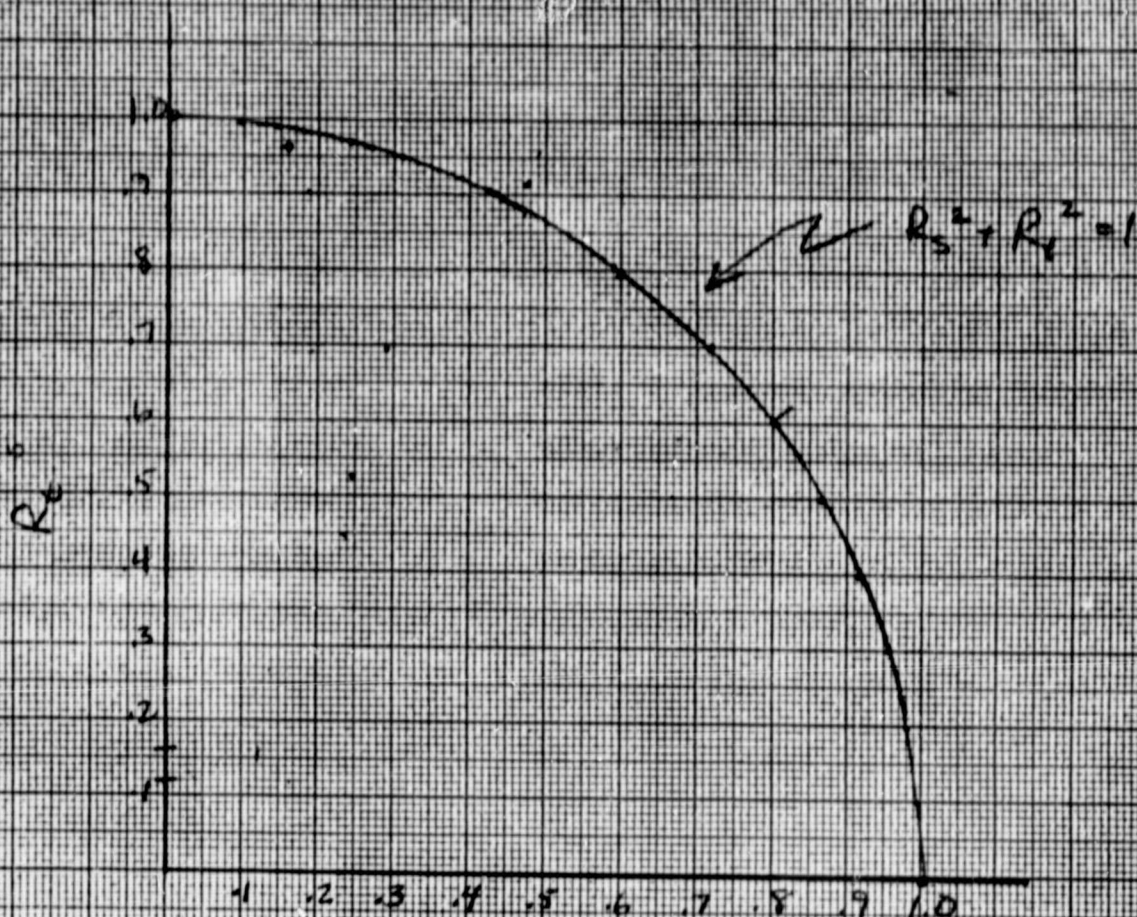
$$R_t = \frac{1.4 (8.735)}{23.2} = .527$$

$$R_s = \frac{1.4 (2.86)}{16.25} = .246$$

$$M.S. = \frac{.7}{.527} - 1 = .71 \quad \text{ULT INTERACTION}$$

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4.8.1



$$R_s = \frac{\text{ACTUAL shear}}{A_v}$$

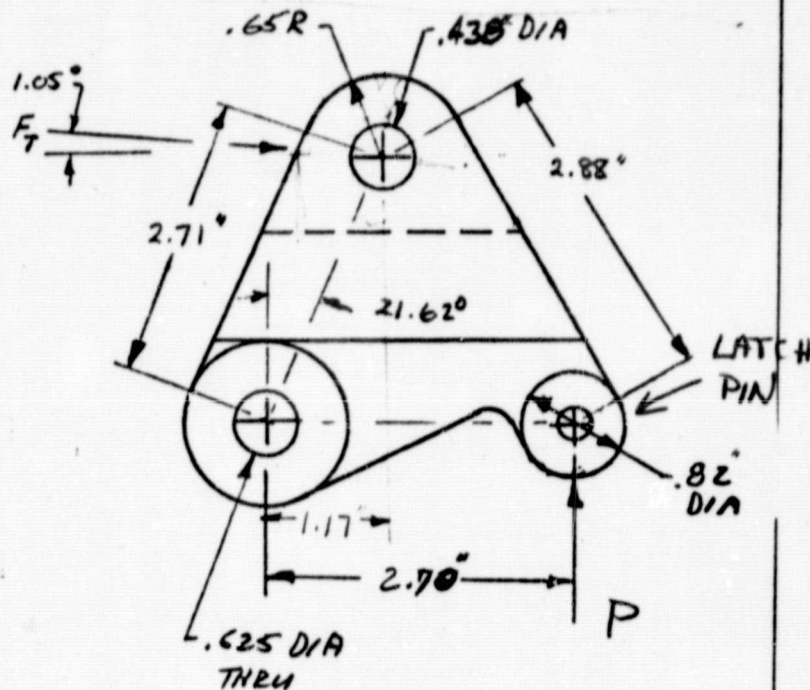
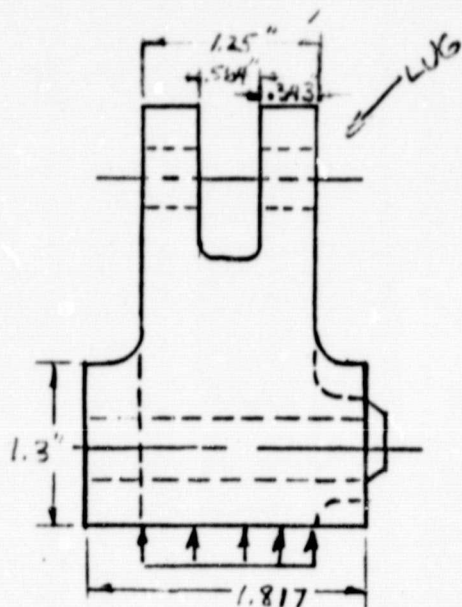
BOLT INTERACTION

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LATCHING MECHANISM

TOGGLE LINK (30A60695)

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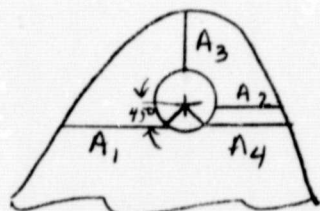
FROM NASTRAN SUMMARY $P_{max} = 5.19 \text{ kips (p. 10)}$
 $F_t = \frac{5.19 (2.7)}{2.45} = 5.72 \text{ kips}$

$$LOAD/LUG = \frac{5.72}{2} = 2.86 \text{ kips}$$

MATL 4340 STL
 $F_{tu} = 180 \text{ ksi}$

$$A_{brg} = .438 (.343) = .15 \text{ in}^2$$

$$A_L = (1.3 - .438)t = .296 \text{ in}^2$$



REF p 3 for procedure

$$A_1 \approx A_4 = .343 (.61) = .21 \text{ in}^2$$

$$A_3 \approx A_2 = (.65 - .219)t = .148 \text{ in}^2$$

$$A_{vg} = \frac{6}{\frac{3}{A_1} + \frac{1}{A_2} + \frac{1}{A_3} + \frac{1}{A_4}} = \frac{6}{\frac{4}{.21} + \frac{2}{.148}} = .184$$

$$\frac{A_v}{A_{brg}} = \frac{.184}{.15}$$

$$= 1.23$$

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LOW ALLOY STEELS

4340, 4140, 300M & Others
 $F_{tu} \approx 300$ ksi

Curve (A) - Approximate
Strength. If K_{tru} is below
this curve, see text (VI:C)
L, T, & ST indicate grain
in direction 'F' in sketch:
L = Longitudinal, T = Long
Transverse, ST = Short
Transverse.

e_u = Ultimate Elongation

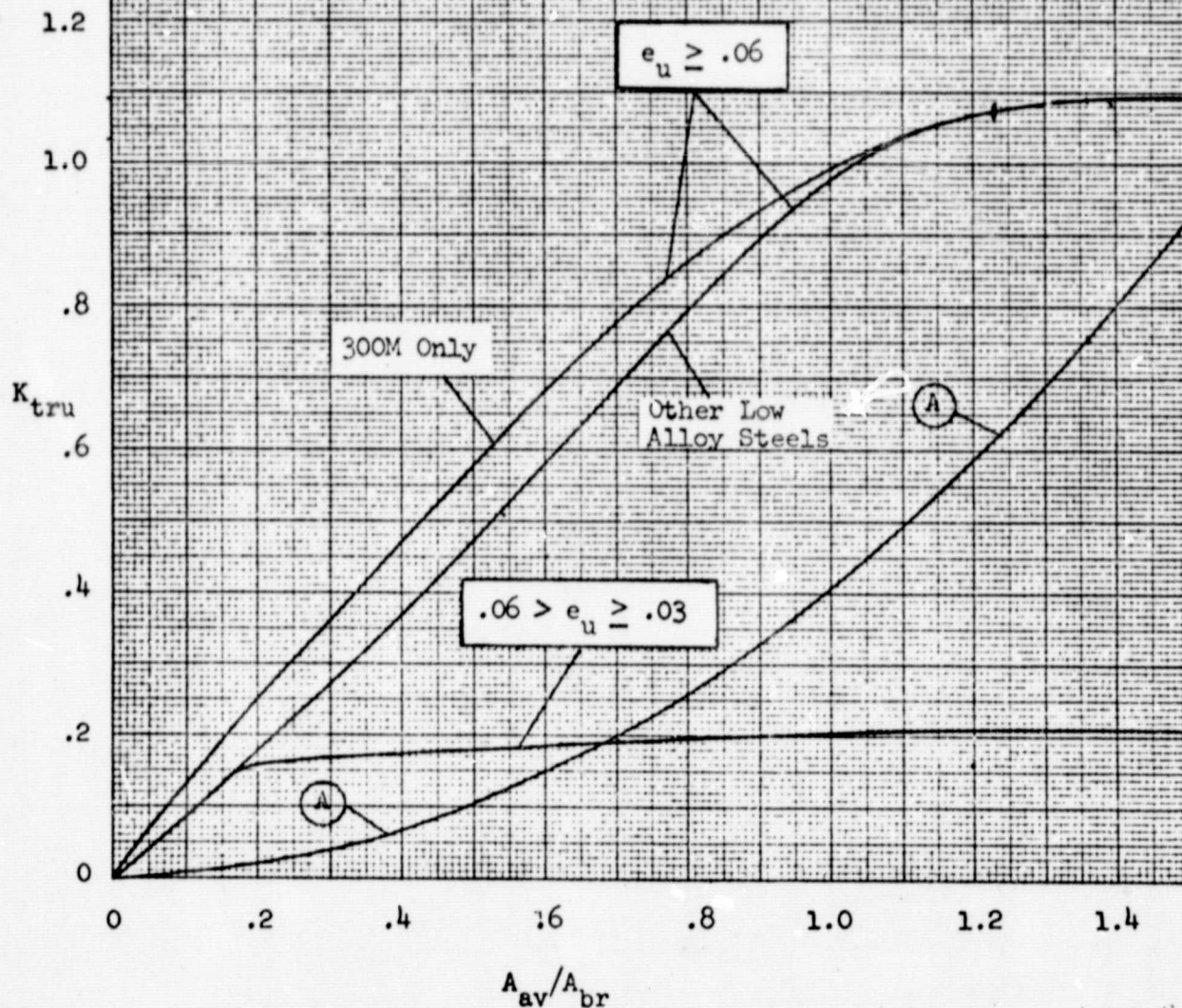
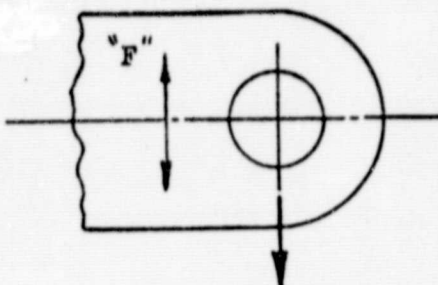


FIGURE 7d

EFFICIENCY FACTORS OF LUGS FOR TRANSVERSE LOAD
LOW-ALLOY STEELS

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IDGLE LINK (CONT.)

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TRANSVERSE LOADS - SHEAR BEARINGS

$$P_{TRU} = K_{TRU} A_{BR} F_{EU}$$

from graph on p. 11 $K_{TRU} = 1.07$

$$P_{TRU} = 1.07 (.15) (150) = 24.10 \text{ kips}$$

$$M.S._u = \frac{24.10}{2(2.86)} = 1 = 3.21 \text{ ULT}$$

LUG DIA (3-A60698) SHEAR

from p. 1.0

$$R_x = 5.72 \text{ kip}$$

$$R_y = 5.09 \text{ kip}$$

$$D_{INDIA} = .625"$$

$$A = .306 \text{ in}^2$$

MATL - INC 718

$$F_{su} = 113 \text{ KSI}$$

$$P_s = \sqrt{5.72^2 + 5.09^2} = 7.66 \text{ kips}$$

$$f_s = \frac{4/3 P_s}{2A} = \frac{4}{3} \frac{7.66}{2(.306)} = 16.69 \text{ KSI}$$

$$M.S._{su} = \frac{113}{2(16.69)} - 1 = 2.39 \text{ ULT SHEAR}$$

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LATCHING MECHANISM (CONT.)

TOGGLE LINK PIN (30A60700)

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SHEAR

MATL TNC 718

F_{su} = 113 KSI

F_{tu} = 180 KSI

RFT P 1.0, $\ell = 5.72 \text{ KIP}$

PIN DIA = .435"
A_s = .15 in²

$$S_{max} = \frac{4/5(5.72)}{2(.15)} = 25.42$$

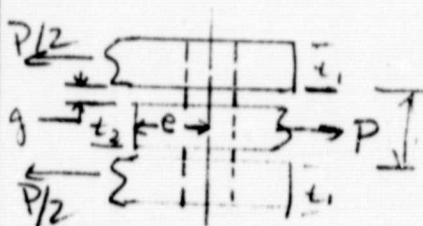
$$W.S. = \frac{113}{2(25.42)} - 1 = 1.22 \text{ ULT SHEAR}$$

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LATCHING MECHANISM (CONT.)
TOGGLE LINK PIN (30A60700)

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BENDING - ref MSFC STRUCTURES MANUAL, B2



from p 10
 $t_1 = .343"$
 $t_2 = .47"$
 $D = .435"$

$$P_{or2} = 32.14 \text{ kips}$$

$$c = .3625"$$

$$r = \left(\frac{c}{D} - \frac{1}{2} \right) \frac{D}{t_2} \quad \text{for INNER LUG}$$

$$= \left(\frac{.3625}{.435} - \frac{1}{2} \right) \frac{.435}{.47} = .731$$

$$g = .564 \text{ in}$$

$$= .047$$

$$\frac{P_u}{A_{br} f_{tu}} = \frac{32.14}{.206(150)} = 1.04$$

REDUCTION FACTOR - from Fig B2.1.0-6 $\delta = .525$

$$\text{moment ARM } b = \frac{t_1}{2} + g + \delta \frac{t_2}{4} = \frac{.343}{2} + .047 + \frac{.525(.47)}{4}$$

$$= .1715 + .047 + .062$$

$$= .281$$

$$M = P \left(\frac{b}{2} \right) = P (.1405)$$

from p 14 $P = 5.72 \text{ kip}$, $M = .804 \text{ kips-in}$

$$f_b = \frac{M y}{I} = \frac{.804(.219)}{.0018} = 97.82 \text{ ksi}$$

$$I = .0018 \text{ in}^4$$

$$y = \frac{.438}{2} = .219$$

PLASTIC BENDING $F_{bll} \approx 300 \text{ ksi}$

$$M.S. = \frac{300}{2(97.82)} - 1 = -.53$$

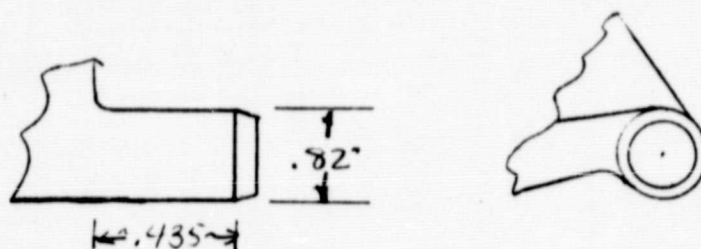
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LATCHING MECHANISM (cont.)

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TOGGLE LINK (cont.) (30A60695)

BENDING OF LATCH PIN -
THIS IS THE PIN THAT ENGAGES WITH THE
LATCH HOOK



BENDING NOT CRITICAL DUE TO SHORT
LENGTH .435"

SHEAR ON PIN

$$A = \left(\frac{.82}{2}\right)^2 \pi = .528 \text{ in}^2$$

$$P = 5.72 \text{ kips (D.10)}$$

$$S_{\text{MAX}} = \frac{4}{3} \frac{P}{A} = \frac{4}{3} \frac{(5.72)}{.528} = 14.44 \text{ KSI}$$

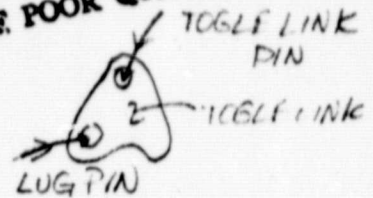
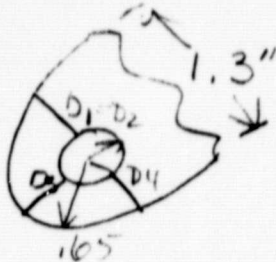
$$M.S.U = \frac{85.5}{2(14.44)} - 1 = 1.95 \text{ ULT SHEAR}$$

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LATCHING MECHANISM (cont.)

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TOGGLE LINK
LUG PIN HOLE (.625")



SEE P. 3 FOR PROCEDURE

$$\text{assume } D_1 = D_4 = D_2 = .3375 \\ D_3 = .3375$$

$$A = 1.817 (.3375) = .613 \text{ in}^2$$

$$A_{AV} = \frac{b}{b/A} = A = .613$$

$$A_{br} = Dt = .625 (1.817) = 1.14 \text{ in}^2$$

$$\frac{A_v}{A_{br}} = \frac{.613}{1.14} = .54$$

$$K_{tu} = .52$$

TRANSVERSE

$$P_{tu} = K_{tu} A_{br} F_{tu} = .52 (1.14) (150) \\ = 88.92 \text{ kips}$$

AXIAL

$$\frac{e}{D} = \frac{.65}{.625} = 1.04 \quad \frac{W}{D} = \frac{1.3}{.625} = 2.08 \quad \frac{D}{t} = \frac{.625}{1.817} = .344 \\ A_t = (1.3 - .625) (1.817) = 1.23 \text{ in}$$

from graph on p. 153 $K_t = .74$

$$P_{tu} = K_t A_t F_{tu} = .74 (1.23) (150) = 173.4$$

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LATCHING MECHANISM (CONT.)

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TOGGLE LINK - LUG PIN HOLE (CONT.)

COMBINED AXIAL & TRANSVERSE LOADS

P. 4

$$R_y = 5.72 \text{ kips}$$

$$R_y = 5.09 \text{ kips}$$

$$R_A = \frac{2(5.72)}{173.4} = .07$$

$$R_{T2} = \frac{5.09(2)}{88.92} = .114$$

$$M.S.U. = \frac{1}{(.07^{.6} + .114^{.6})^{.25} - 1} = 5.9 \text{ uot}$$

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LATCHING MECHANISM cont.

LUG PIN (30A60698)

BENDING

$$t_1 = .35 \text{ in}$$

$$t_2 = 1.817$$

$$g = g_1 P = \frac{1.896 - 1.817}{2} = .047$$

$$P_{TRU} = 28.89$$

ref MSFC STRUCT. MANUAL p B2.5

$$r = \left[\frac{g}{D} - \frac{1}{2} \right] \frac{D}{t_2} = \left(\frac{.65}{1.625} - \frac{1}{2} \right) \frac{1.625}{1.817} = .27$$

from graph B2.1.0-6

from p.

$$P_u = 88.72 \text{ kips}$$

$$\frac{P_u}{A_b r F_{tu}} = .52$$

from graph B2.1.0-6

MSFC STRUCT MAN.
 $r = .52$

$$b = \frac{.35}{2} + .047 + .52 \left(\frac{1.817}{4} \right) = .458$$

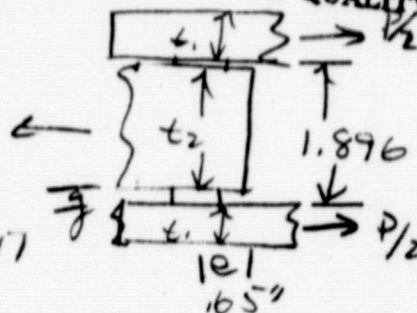
$$M = \frac{P_b}{2} = \frac{7.66(.458)}{2} = 1.75 \text{ IN-kips}$$

$$S_b = \frac{M}{I} = \frac{1.75(.3125)}{.0115} = 72.9 \text{ ksi}$$

$K = 1.7$ $F_{bu} = 300 \text{ ksi}$ PLASTIC BENDING

$$M.S.u = \frac{300}{72.9(2)} - 1 = 1.05 \text{ ULT}$$

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4.15.3

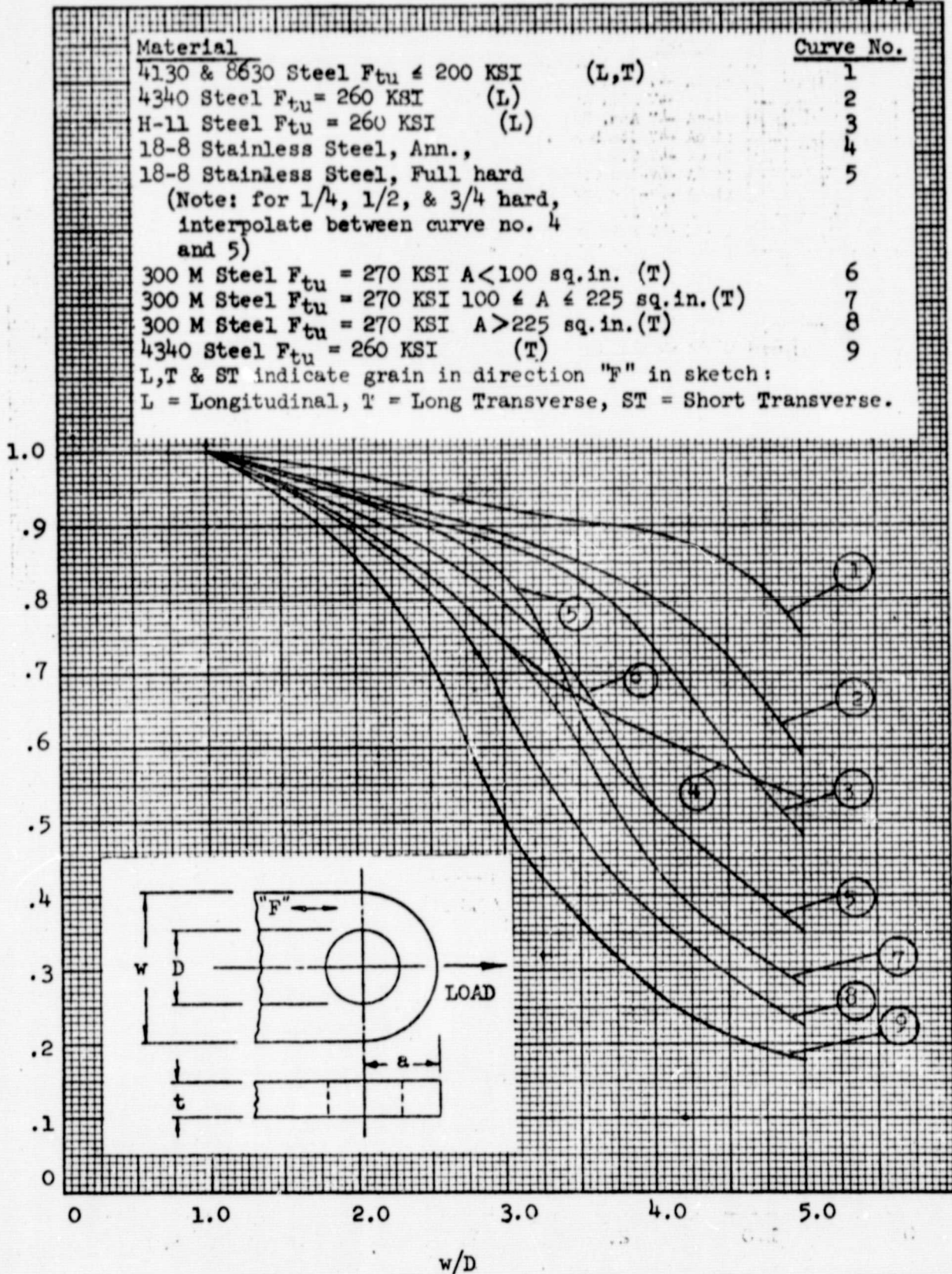


FIGURE 5c TENSION EFFICIENCY FACTORS OF LUGS FOR AXIAL LOADING
STEEL

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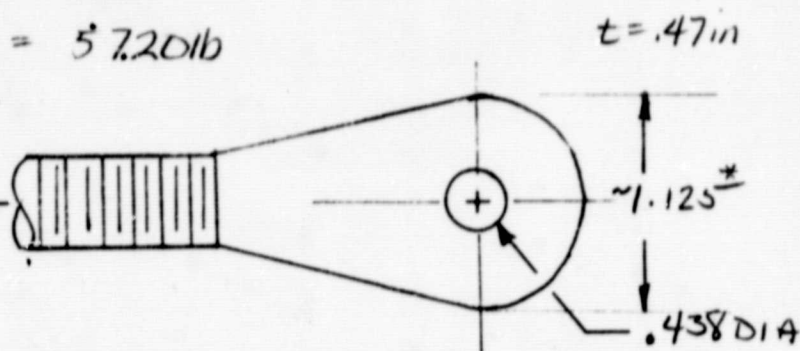
LATCHING MECHANISM (cont.)

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TURNBUCKLE ROD END

From p. 1.0 $F_t = 57,201 \text{ lb}$
MATERIAL 4340 STEEL
 $F_{tu} = 150 \text{ ksi}$

F_t ←
MAX TENSION LOAD
- ASSUMES LOAD
REVERSAL IN LATCH



$$\frac{a}{D} = \frac{.5625}{.438} = 1.28''$$

$$\frac{D}{t} = \frac{.438}{.47} = .93''$$

* MEASURED OFF
ASSEMBLY
30A60699

$$\frac{w}{D} = \frac{1.125}{.438} = 2.57''$$

$$A_{BRG} = .438 (.47) = .206 \text{ in}^2, A_t = (1.125 - .438) (.47) = .323 \text{ in}^2$$

SHEAR BEARING -

$$P_{BRG} = K_{BR} A_{BRG} F_{tu}$$

$$= 1.04 (.206) (150,000)$$

$$P_{BRG} = 32,136 \text{ lb}$$

$K_{BR} = 1.04$
ref MSFC STRUCT.
MANUAL
B2

TENSION -

$$P_{tu} = K_t A_t F_{tu}$$

$$= .715 (.323) (150,000)$$

$$P_{tu} = 44,332 \text{ lb}$$

$$K_t = .915$$

$$M.S.U. = \frac{32,136}{(2) (57,201)} - 1 = .1.81 \text{ ULT SHEAR BEARING}$$

TURNBUCKLE BARREL ADEQUATE BY INSPECTION

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LATCHING MECHANISM (CONT.)

THREADED SECTION

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$$5/8-18 \text{ THDS } A_R = .24 \text{ in}^2$$

$$f_t = \frac{5.72}{.24} = 23.83 \text{ ksi}$$

$$M.S.U. = \frac{150}{(.2)(23.83)} - 1 = 2.15$$

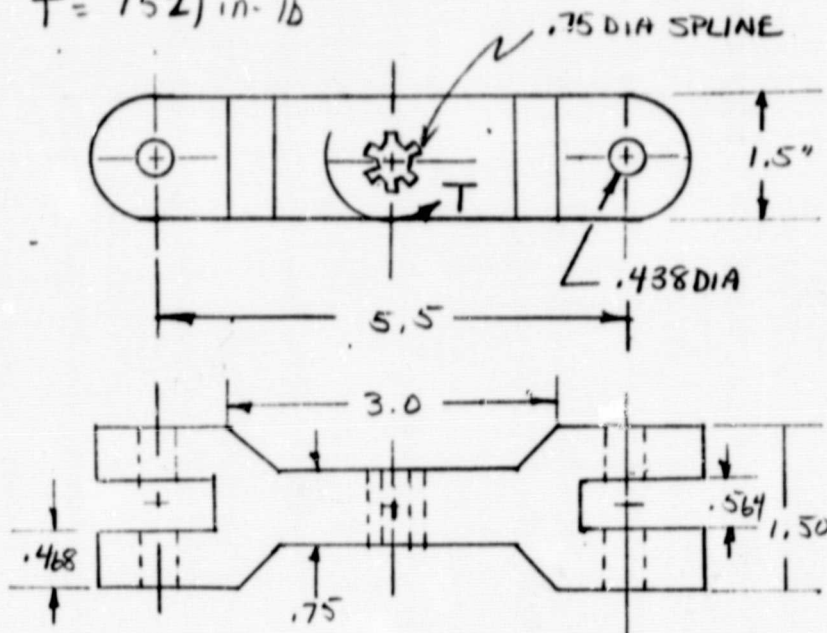
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LATCHING MECHANISM (cont.)

BELL CRANK
(30A100696)

T = 152 lbf

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SPLINE STRESSES

Shear stress @ PITCH DIA OF TEETH:
ref. MACHINERY'S HDBK p 1023

$$f_s = \frac{4TK_a K_m}{DN L_e t K_f}$$

$K_a = 1.2, K_m = 1.0, K_f = 1.0$
 $L_e = .70, t = .186$
(Assumes 1/2 of teeth
reacting T)

$$f_s = \frac{4(152)(1.2)(1.0)}{.75(6)(.70)(.186)(1.0)}$$

$$S_s = 12460 \text{ psi}, F_{su} = 40 \text{ ksi (ref MACHINERY'S HDBK TABLE 3)}$$

$$M.S.u = \frac{40}{2(12.46)} - 1 = .60 \text{ ULT}$$

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LATCHING MECHANISM (CONT.)

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MAXIMUM SPLINE TORQUE CAPABILITY

MATL - 4340 STEEL

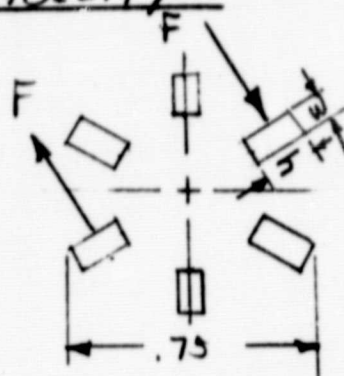
$F_{TU} = 180 \text{ Ksi}$

$F_{SU} = 108 \text{ Ksi}$

$F_{CY} = 173 \text{ Ksi}$

$L = .75 \text{ in}$, $h = .094 \text{ in}$

$W = .188 \text{ in}$



FOR SHEAR FAILURE

$$T_s = F d N, \quad N = 3 \text{ SETS OF TEETH ON SPLINE}$$

$$S_s = \frac{1.5 F}{A_s} = \frac{1.5}{.75(.188)} F$$

$$F = \frac{.75(.188)(108000)}{1.5} = 10,150 \text{ lb}$$

$$T_s = 10150 (.75)(3) = 22840 \text{ in-lb}$$

ASSUMING $1/2$ OF SPLINE TEETH TRANSMIT TORQUE

$$T_s' = \frac{T_s}{F.S. (2)} = \frac{22840}{2(2)} = 5710 \text{ in-lb}$$

COMPRESSION FAILURE

$$T_c = F d N, \quad S_c = F/A_c \Rightarrow F = A_c f_c = .75(.094)(173,000)$$

$$F = 12196 \text{ lb}$$

$$T_c = 12196 (.75)(3) = 27440 \text{ in-lb}$$

$$T_c' = \frac{27440}{2(2)} = 6860 \text{ in-lb}$$

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LATCHING MECHANISM (cont.)

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SHAFT DESIGN TORQUE

$T = 4000 \text{ in-lb}$ (REF: ED23-82-166, 12/6/82)

From p 11, LIMIT TORQUE = 5710 in-lb

$$M.S. = \frac{5710}{4000} - 1 = .42$$

TORSION ON MINIMUM SHAFT SECTION

$$\begin{aligned} \tau_{s_{max}} &= \frac{16T}{\pi d^3} = \frac{16(4000)}{\pi (.7499)^3} & d &= .7499 \text{ in} \\ &= 48,300 \text{ psi} \end{aligned}$$

SHAFT: A286 CRES.
 $F_{tu} = 140 \text{ KSI}$
 $F_{su} = 91 \text{ KSI}$

$$M.S.u = \frac{71}{48.3} - 1 = .88$$

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LATCHING MECHANISM (CONT.)

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BELL CRANK SPLINE CONT.

COMPRESSIVE STRESS ON SIDES OF TEETH -

$$\begin{aligned}
 f_c &= \frac{2 T K_m K_a}{9 D N L_c h K_f} \quad h = .094 \\
 &= \frac{2(1521)(1.2)(1.0)}{9(.75)(6)(.70)(.094)(1.0)} \\
 f_c &= 1370 \text{ psi}
 \end{aligned}$$

$F_{cu} = 3 \text{ ksi}$ ref: MACHINERY'S HDBK p 1024

$$MS_u = \frac{3.0}{2(1.37)} - 1 = .09$$

TENSILE STRESS IN SPLINE ref: MACHINERY'S HDBK p 1024

RADIAL TENSILE STRESS $S_{r1} = \frac{T \tan \phi}{\pi D L W}$

$$\begin{aligned}
 \phi &= \text{PRESSURE } \Delta = 30^\circ \\
 D &= .75 \text{ in}, L = .75 \text{ in} \\
 L W &= .562 \text{ in}
 \end{aligned}$$

$$S_{r1} = \frac{1521 \tan 30^\circ}{\pi(.75)(.562)(.75)} = 884 \text{ psi}$$

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LATCHING MECHANISM (CONT.)

BELLCRANK SPLINE (CONT.)

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BEAM LOADING TENSILE STRESS

$$S_{t3} = \frac{4T}{D^2 L_e Y}$$

$$L_e = .70, Y = 1.5 \Rightarrow S_{t3} = \frac{4(1521)}{.75^2 (.70)(1.5)} = 10300 \text{ psi}$$

TOTAL TENSILE STRESS -

$$S_T = \frac{K_A K_M}{K_F} (S_{t1} + S_{t3}), \quad K_A = 1.2, \quad K_F = 1.0, \quad K_M = 1.0$$

$$\text{from p. 13} \quad S_{t1} = 884 \text{ psi}$$

$$S_T = \frac{1.2(1.0)}{1.0} (884 + 10300) = 13420 \text{ psi}$$

$$F_{Tu} = 45 \text{ ksi (ref TBL 7, MACHINERY'S HDBK)}$$

$$M.S. u = \frac{45}{2(13.42)} - 1 = .67$$

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LATCHING MECHANISM (CONT.)

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BELLCRANK LUGS

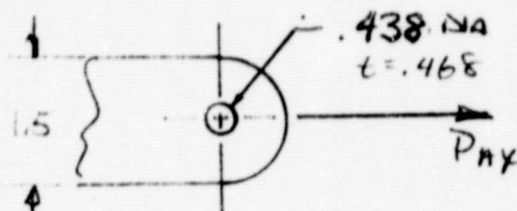
MATL - 4340 STEEL

MAY TENSION LOAD - CONSERVATIVELY
ASSUMING, LOAD REVERSAL

From P. 4

$$P_{AXIAL} = 5.72 \text{ lb}$$

$$= 2.86 \text{ lb/LUG}$$



$$\frac{e}{D} = \frac{.75}{.438} = 1.71, \quad \frac{d}{t} = \frac{1.5}{.438} = 3.42$$

$$\frac{D}{t} = \frac{.438}{.468} = .94$$

$$A_{BRG} = .438 (.468) = .205 \text{ in}^2$$

$$A_t = (1.5 - .438 (.468)) = .50 \text{ in}^2$$

HEAR BEARING

$$K_{BR} = 1.6$$

$$P_{BRU} = K_{BRU} A_{BR} F_{tU} = 1.60 (.205)(150000)$$

$$= 49.2 \text{ kips}$$

TENSION

$$K_t = .93$$

$$P_{tU} = K_t A_t F_{tU} = .93 (.50)(150000)$$

$$= 69.75 \text{ kips}$$

$$11. S_{11} = \frac{49200}{(2) 15,780} - 1 = 3.30$$

LUG BENDING @ JOGGLE ADEQUATE BY
INSPECTION!

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Checked by: EIW	Date: 6/84	Title: SPARTAN REM	Model		
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LATCHING MECHANISM (cont.)

ROUND PIN - (3DA60642)

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SHEAR

PIN DIA = 1.875 IN (SHAFT)

(From NASD-W RUN 0000311)
ref p. 51)

$$\begin{aligned} F_z &= 11.73 \text{ kip} \\ F_x &= 7.91 \text{ kip} \end{aligned} \quad \left. \vphantom{\begin{aligned} F_z &= 11.73 \text{ kip} \\ F_x &= 7.91 \text{ kip} \end{aligned}} \right\} \text{LOAD CASE 2}$$

$$R = \sqrt{11.73^2 + 7.91^2} = 14.15 \text{ kips}$$

$$A_s = \frac{\pi (1.875)^2}{4} = .60 \text{ in}^2$$

$$f_{s \text{ MAX}} = \frac{4}{3} \frac{R}{A} = \frac{4}{3} \frac{14.15}{.60} = 31.44 \text{ KSI}$$

TNC 718 $f_{su} = 113 \text{ KSI}$

$$M.S.u = \frac{113}{2(31.44)} - 1 = .78 \text{ ULT SHEAR}$$

SQUARE PIN (SAME SHAFT AS ROUND PIN)

\therefore SAME M.S.

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LATCHING MECHANISM (cont.)

ADAPTER / PIN INTERFACE

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(from NASTRAN RUN N0003111)
ref. p. 51

LOAD CASE 7

$P_{IFN} = 9.47 \text{ KIPS}$

SHEAR TEAR OUT

$$A_s = 1.0(.69)^2$$

$$= 1.38 \text{ IN}^2$$

7/18-14 TH

$$S_{s \max} = \frac{3}{2} \frac{P_T}{A_s}$$

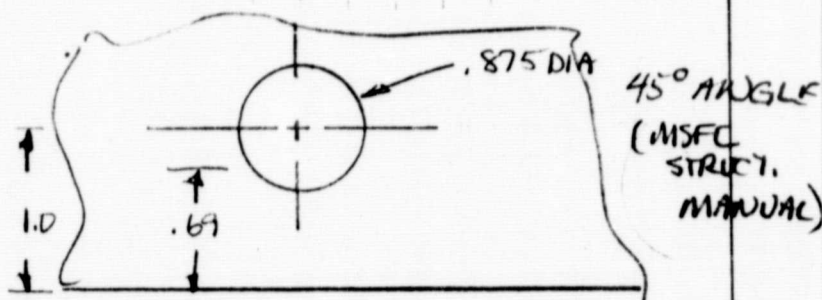
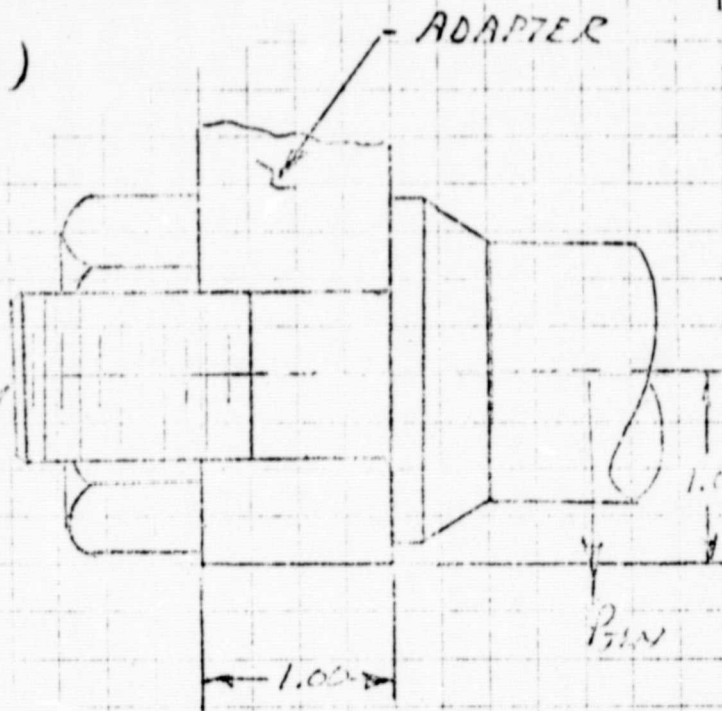
$$= \frac{3}{2} \frac{(9.47)}{1.38}$$

$$= 10.29 \text{ ksi}$$

2219-T87 AL

$F_{TU} = 63 \text{ KSI}$

$F_{SU} = 37 \text{ KSI}$



$$M.S.U = \frac{37}{2(10.29)} - 1 = .798 \text{ ULT SHEAR}$$

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ADAPTER / PINS (CONT.)

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ref p. 51 for PIN REACTIONS

$$P_H = 5.19 \text{ kip}$$

$$P_V = 11.73 \text{ kip}$$

$$I/c = .3 \text{ in}^3$$

$$M_{\text{max}} = 1.0 P_H - P_V (1.1)$$

$$= 5.19 - 12.9$$

$$= 7.713 \text{ in-kip}$$

$$f_b = \frac{7.7}{.3} = 25.7 \text{ ksi}$$

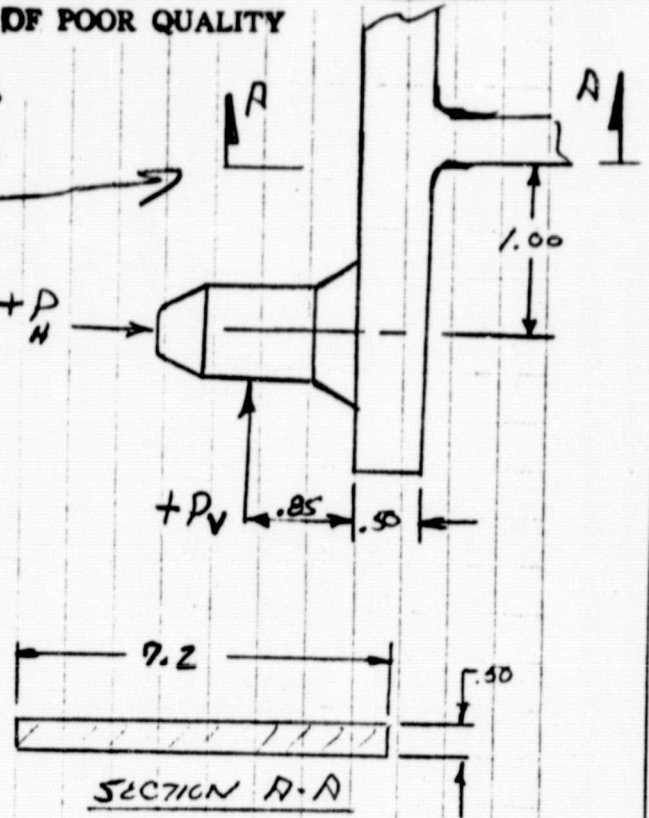
$$f_t = \frac{P_V}{A} = \frac{11.73}{3.6} = 3.26 \text{ ksi}$$

PLASTIC BENDING

$$F_{bu} = 97 \text{ ksi (see graph)}$$

NEXT PAGE

$$R_b = \frac{2(25.7)}{97} = .53$$



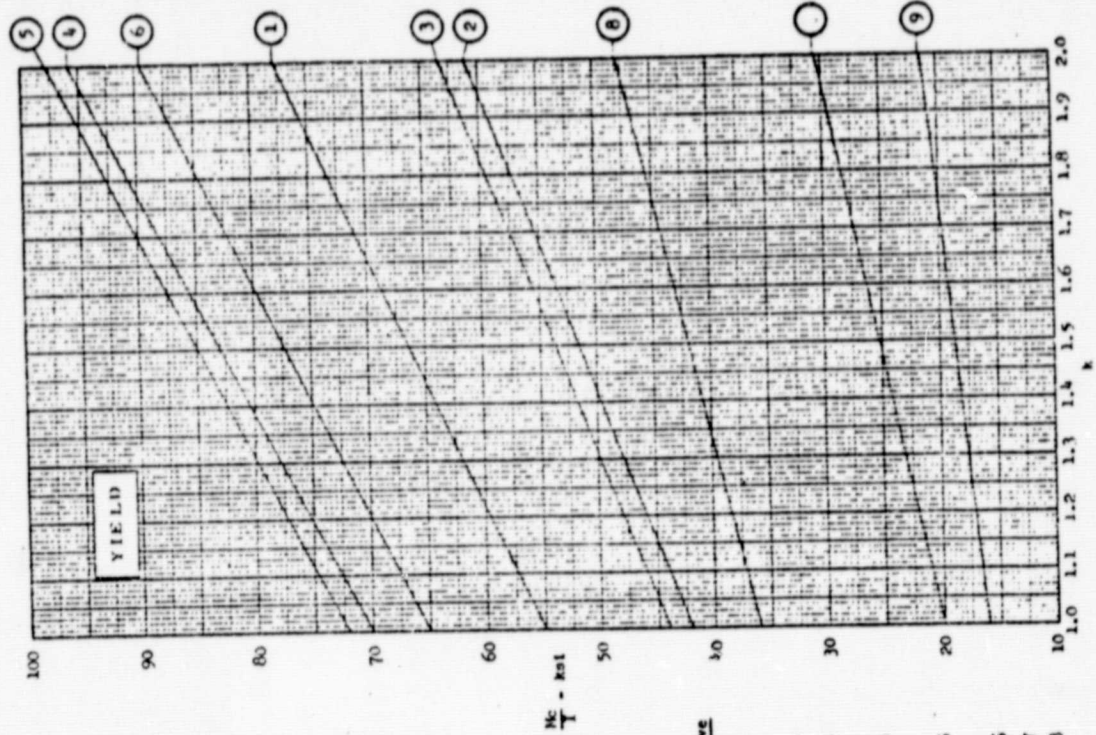
$$A = 3.60 \text{ in}^2$$

$$R_t = \frac{2(3.26)}{62} = .11$$

$$M.S.U = \frac{1}{.53 + .11} = 1 = .56 \text{ ULT}$$

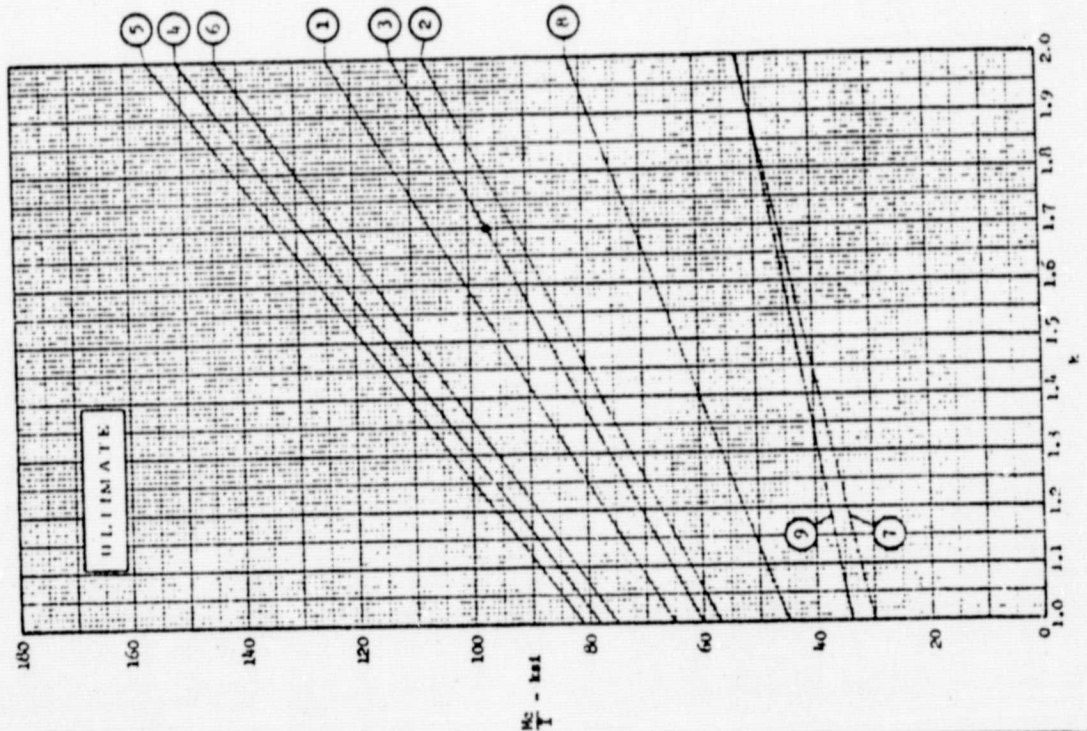
4.27

FIGURE 5a
Ultimate and Yield Values of $\frac{Mc}{T}$ for Symmetrical Sections
Aluminum and Magnesium Alloy
Room Temperature



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Material Designation	Curve
2014-T6 Die Forging	1
2024-T4 Extrusion	2
$t \leq .250$	3
$.250 \leq t \leq 3.00$	4
7075-T6 Extrusion	5
$t \leq .250$	6
7075-T6 & -T651 Extrusion	7
$.250 \leq t \leq 3.00$	8
7075-T6 Die Forging	9
$t \leq 2.00$	
356-T6 Sand Casting	
2618A-T5 Extrusion	
AZ63A-T6 & AZ19C-T6	
Sand & Perm Mold Casting	9



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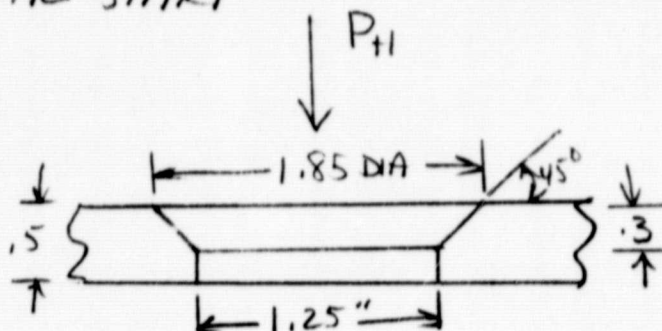
LATCHING MECHANISM (CONT.)

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BEARING ON CONICAL START

12000 PSI

$$P_{MAX} = 5.19 \text{ kip}$$



$$\text{Avg DIA} = 1.55 \text{ in}$$

$$\text{CIR} = \pi(1.55) = 4.869 \text{ in}$$

$$A_{BEG} = \frac{.3}{.707} (4.869) = 2.066 \text{ in}^2$$

$$f_{BEG} = \frac{5.19}{2.066} = 2.51 \text{ ksi}$$

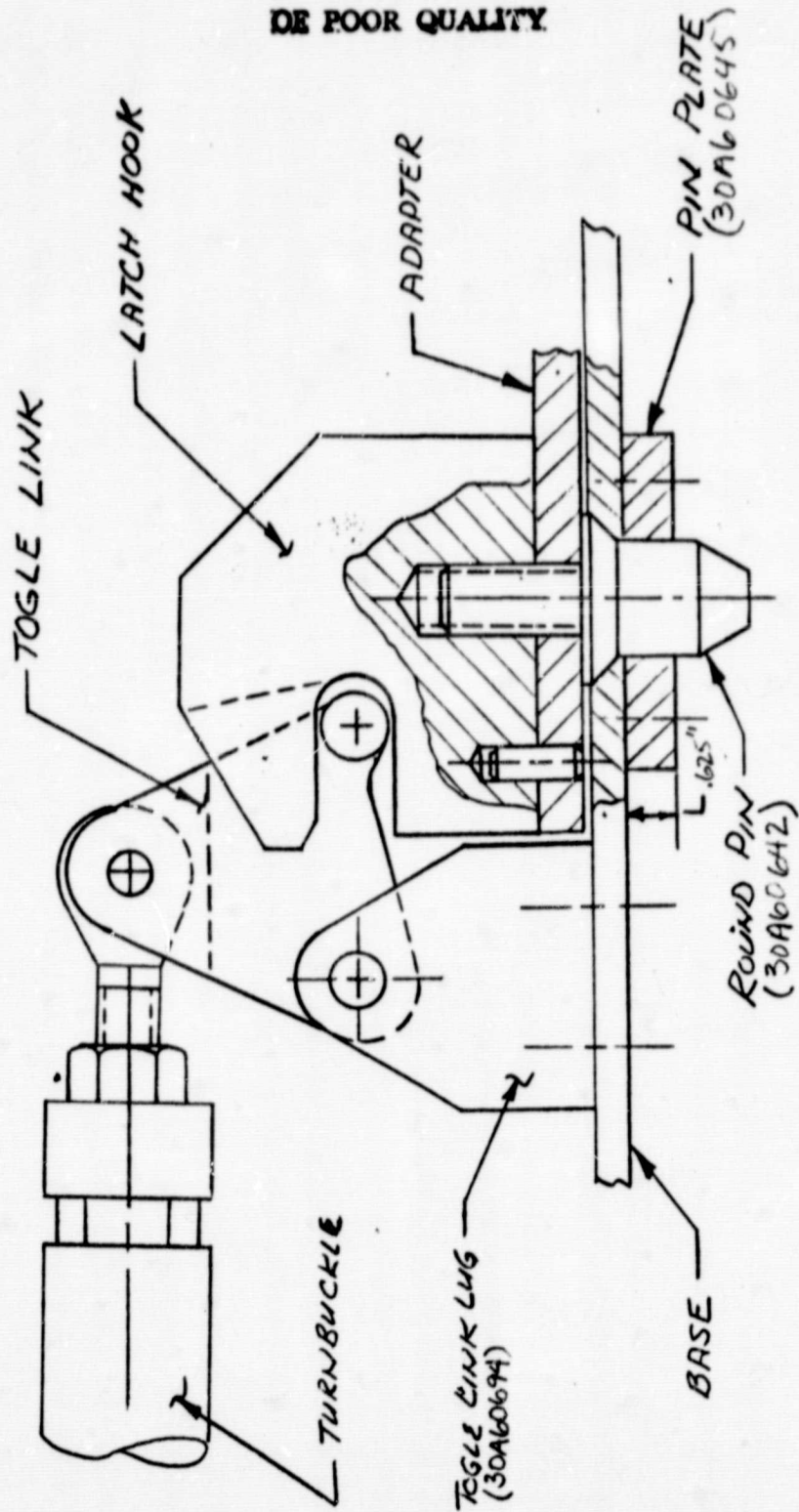
MATL: 2219 T87 AL
F_{BRU} = 99 ksi

$$M.S.R. = \frac{99}{2(2.51)} - 1 = \underline{\underline{LARGE}}$$

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LATCH MECHANISM ASSY



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ADAPTER / BASE PINS (CONT.)

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SHEAR ON MOUNTING BOLTS

(4) NAS1957C
from P. (51)
 $P_x = 7.91$
 $P_z = 11.73$
 $P_y = 5.19$

$$P_s = \sqrt{11.73^2 + 7.91^2} = 14.14 \text{ kip}$$

$$P_s / \text{BOLT} = 14.14 / 4 =$$

$$P_s = 3.54 \text{ kip/BOLT}$$

$$M.S.u = \frac{16.25}{2(3.54)} - 1 = 1.30 \text{ ULT SHEAR}$$

BEARING

$$A_{BRG} = .437(.5) = .2185 \text{ in}^2$$

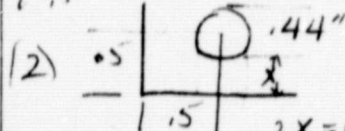
$$S_{br} = \frac{3.54}{.2185} = 16.2 \text{ ksi (LOW)}$$

SHEAR TEAR OUT IN PLATE

PLATE MATL - INC 718

$$F_{tu} = 180 \text{ ksi}$$

$$F_{su} = 113.4 \text{ ksi}$$



$$A_s = 2xt = (2)(.50)(.56) = .56$$

$$2X = (.5 - \frac{.44}{2}) / (.28) = .56$$

$$S_s = \frac{3}{2} \frac{11.73}{.56} = 31.42 \text{ ksi}$$

$$M.S.u = \frac{113.4}{(2)(31.42)} - 1 = .80$$

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Checked by: EJW	Date 6/84	Title SPARTAN REM	Model		
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FAST PIN PLATE (cont.)

(FROM NASTRAN RUN NO003111)

1st P. 5) LOAD CASE 2

$$P_H = -7.91 \text{ kip}$$

$$P_V = 11.73 \text{ kip}$$

ref diagram on p. 24

$$A_s = [(1.625)(2)(.56)] = .70$$

MAT'L - INCONEL 718

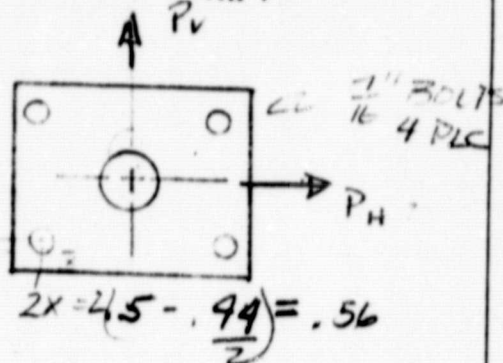
$$F_{tu} = 180 \text{ ksi}$$

$$F_{su} = 113 \text{ ksi}$$

$$S_{SMALL} = \frac{2}{2} \left(\frac{11.73}{.70} \right) = 25.14 \text{ ksi}$$

$$M.S.u = \frac{113}{2(25.14)} - 1 = -1.25$$

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LAUNCHING MECHANISM (CONT.)

ROUND PIN/ADAPTER BEARING

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$$A_{brg} = .875 (1.0) = .875 \text{ in}^2 \quad (\text{ref. fig. p. 48})$$

$$S_{brg} = \frac{14.15}{.875} = 16.17 \text{ ksi}$$

$$F_{bru} = 94.5 \text{ ksi}$$

$$M.S._{bru} = \frac{94.5}{2(16.17)} - 1 = 1.92 \text{ ULT BEARING}$$

SQ PIN PLATE (30A60644)

INC 718

SHEAR TEAROUT

(FROM NASTRON RUN 1000511)

REF p. 5V

$$\text{MAX VERT SHEAR} = 11.69 \text{ kp}$$

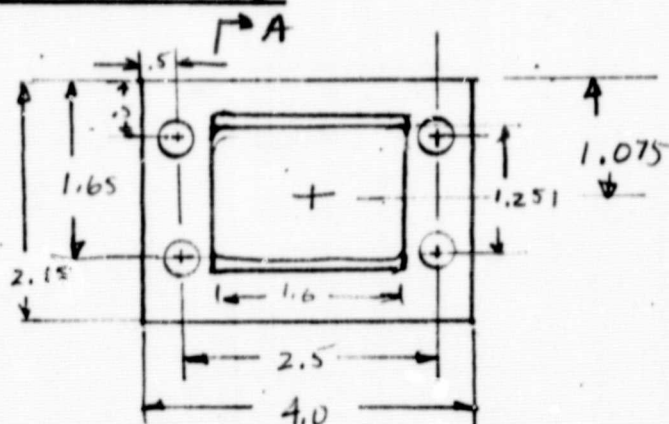
$$A_s = (1.075 - .625 \times .625) 2$$

$$= .5625 \text{ in}^2$$

$$S_{s \text{ MAX}} = \frac{11.69 (3/2)}{.5625} = 31.17 \text{ ksi}$$

$$F_{su} = 113 \text{ ksi}$$

$$M.S._u = \frac{113}{2(31.17)} - 1 = .81 \text{ ULT SHEAR}$$



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LATCHING MECHANISM cont.

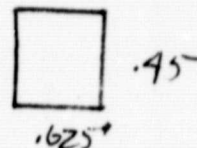
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SO PIN PLATE - BENDING

(SEE REFERENCE ON PREVIOUS PAGE)

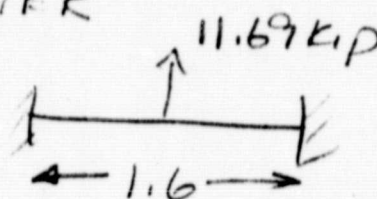
$$I_{AA} = \frac{.45^3 (.625)}{12} = .004746 \text{ in}^4$$

$$I/c = \frac{.004746}{.45/2} = .0211 \text{ in}^3$$



USING PT LOAD @ CENTER

$$M = \frac{P L}{8} = \frac{11.69 (1.6)}{8} = 2.358 \text{ in-kip}$$



$$f_b = \frac{2.358}{.0211} = 111.75 \text{ KSI}$$

$$f_{bu} = 180 \text{ KSI}$$

$$F_{bu} = 180 (1.7) = 306$$

$$M.S.U. = \frac{306}{2(111.75)} - 1 = .37$$

SHEAR

$$R = \frac{11.69}{2} = 5.845 \text{ kip}$$

$$F_{su} = 113.4 \text{ KSI}$$

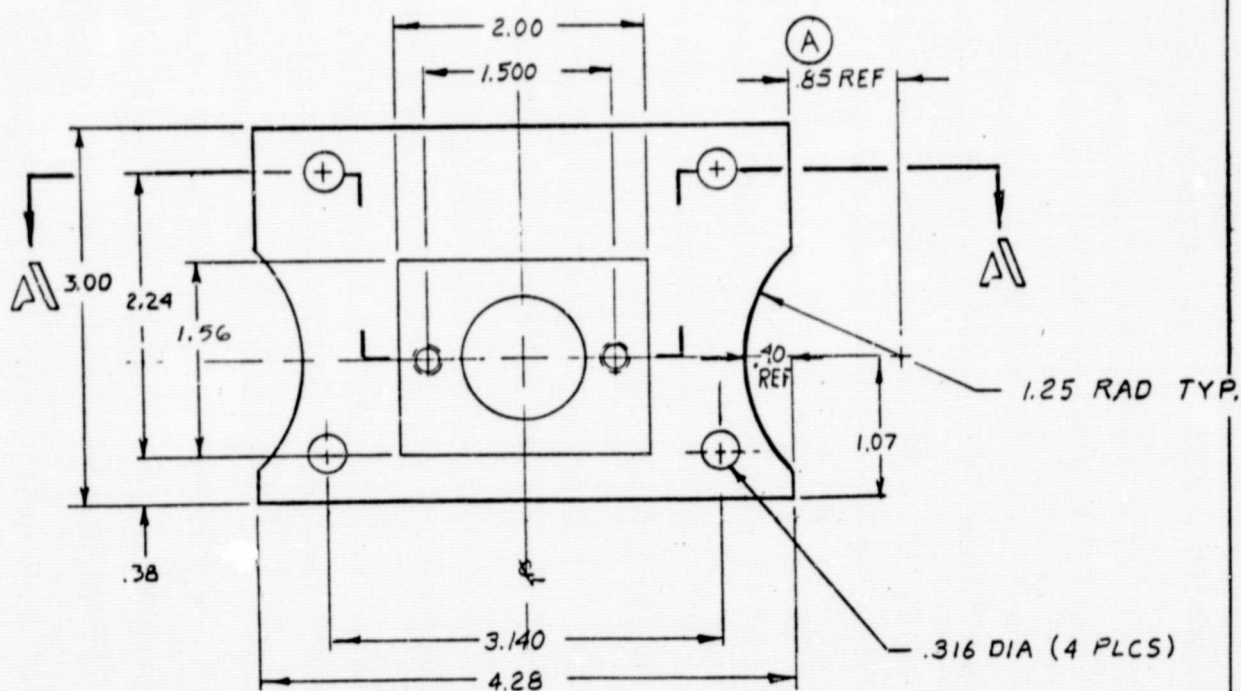
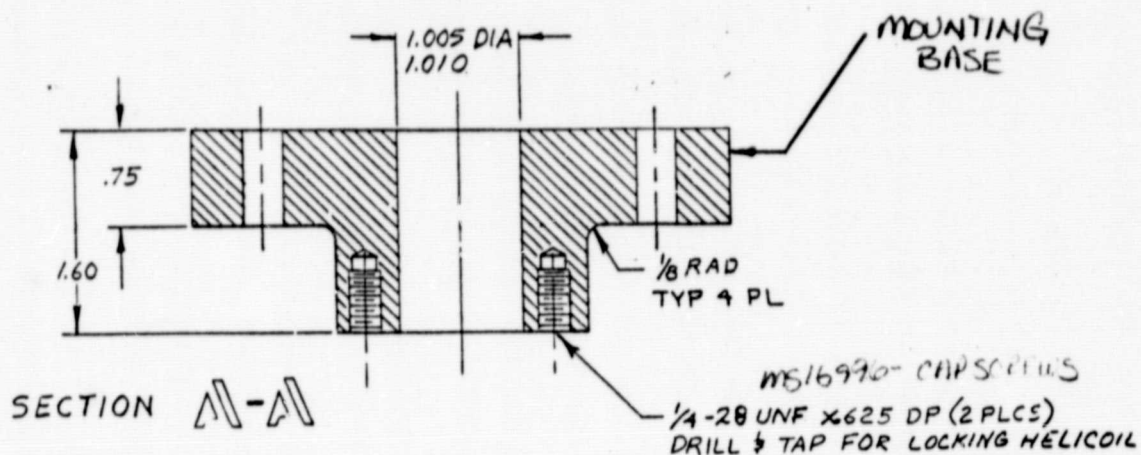
$$f_s = \frac{3/2 (5.845)}{.625 (.45)} = 31.17 \text{ KSI}$$

$$M.S.U. = \frac{113.4}{2(31.17)} - 1 = .82$$

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Checked by: EJW	Date: 6/84	Title: SPARTAN REM	Model:
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30A60647 RODS POSITION
HOLDER, 4

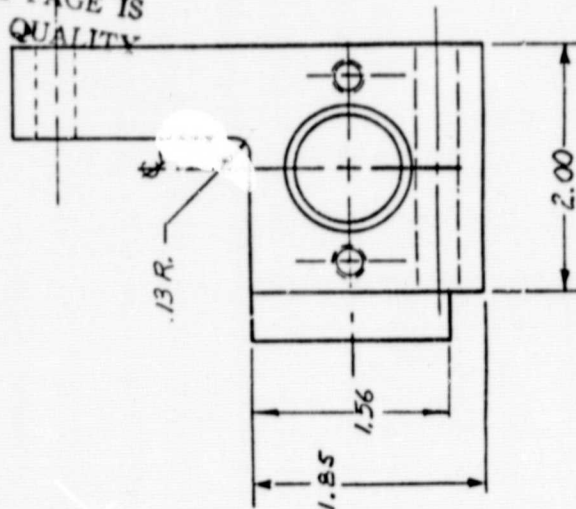


SINGLE ROD HOLDER

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30A60648
BASE ROD
POSITIONS 2 & 3

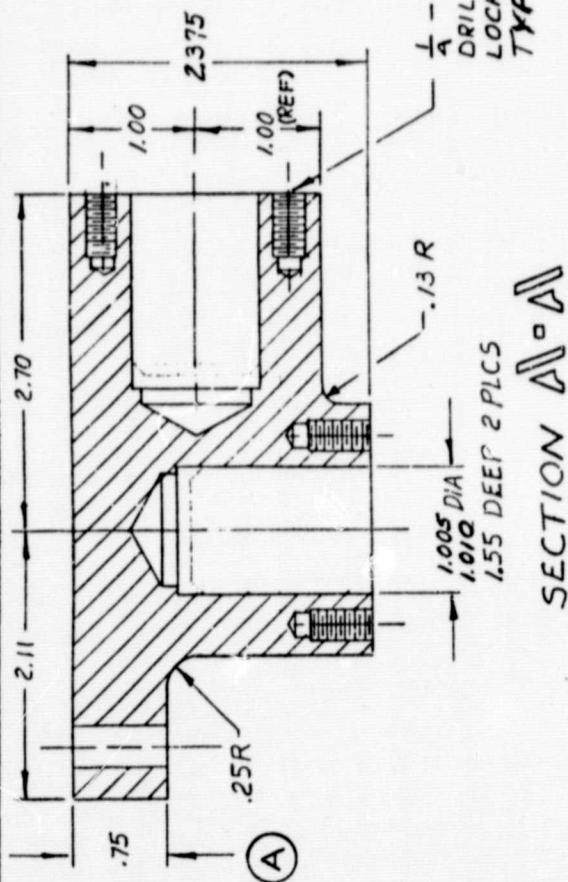
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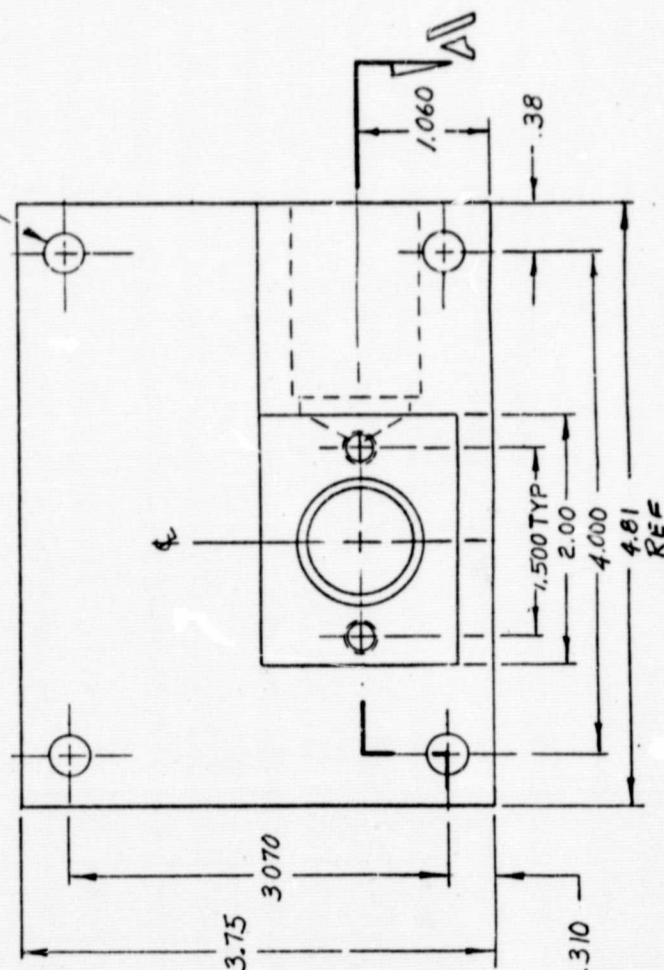
CAP
SCREW
M514976-23

1 - 28 X .50 DP
DRILL & TAP FOR
LOCKING HELICOIL
TYP 4 PL

.316 DIA THRU (4 PLCS.)



SECTION A-A



DOUBLE ROD BASE

Prepared by: DSM	Date: 5/84	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page: 4.34	Temp.	Perm.
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LOCATION ROD BASE FTG

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SINGLE ROD BASE (30A60447) MOUNTING FASTENERS -

(4) 5/16-24 SCREWS MS51960-106

$$P_{tu} = 4.64 \text{ kip}$$

$$P_{su} = 4.06 \text{ kip (based on 53 KSI CRES STEEL (5/16 DIA))}$$

MAX ROD LOAD - 1560 lb (ref. ED23-83-42)
3/20/83

Moment @ BASE - (SEE Fig p 34)

$$M = 1560(3 + 1.6) = 7.176 \text{ in-kip}$$

TENSION -

$$P/\text{FASTENER} = \frac{7.176}{2.24(2)} = 1.6 \text{ kip}$$

SHEAR

$$P_s = 1560/4 = .39 \text{ kip}$$

INTERACTION (ref p 4.9.1)

$$R_t = \frac{2(1.6)}{4.64} = .69$$

$$R_s = \frac{2(.39)}{4.06} = .19$$

$$M.S. = \frac{.90}{.69} - 1 = .39 \text{ INTERACTION}$$

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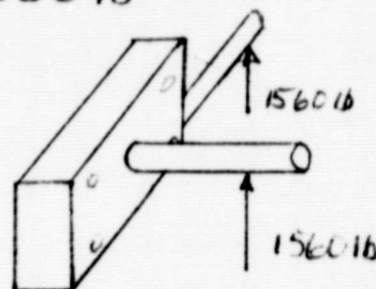
LOCATOR ROD BASE FTG (CONT.)

MOUNTING FASTENERS

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DOUBLE ROD BASE - 30A60648

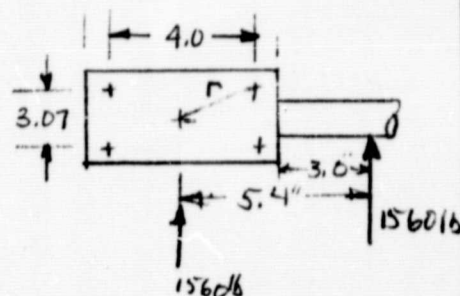
ASSUME BOTH RODS
STRIKE "V" FTGS @
THE SAME TIME.
(REF p. 17)



ULT SHEAR

$$P_s = \frac{1560(2)}{4} = 780 \text{ lb/FASTENER}$$

$$r = \sqrt{4.0^2 + \left(\frac{3.07}{2}\right)^2} = 2.52 \text{ in.}$$



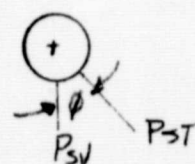
$$P_{ST} = \frac{1560(5.4)}{4(2.52)} = 835.7 \text{ lb/FAST}$$

$$\tan \phi = \frac{1.535}{2.10} = .7675$$

$$\phi = 37.5^\circ$$

$$P_{SV} = 835.7 \cos(37.5^\circ) = 663.0 \text{ lb}$$

$$P_{SH} = 835.7 \sin(37.5^\circ) = 508.7 \text{ lb}$$



RESULTANT SHEAR

$$P_{SR} = \left[(P_{SV} + P_s)^2 + P_{SH}^2 \right]^{1/2} = \left[(663 + 780)^2 + (508.7)^2 \right]^{1/2}$$

$$= 1530.0 \text{ lb}$$

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LOCATOR ROD BASE FIG (CONT.)

MOUNTING FASTENERS

DOUBLE ROD BASE (CONT.)

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from p. 19, $P_{su} = 1.530 \text{ kips}$

$\frac{7}{16}$ -24 FASTENERS

$P_{tu} = 4.04 \text{ kip}$

$P_{su} = 4.06 \text{ kip}$

$$M.S.U. = \frac{4.06}{2(1.53)} - 1 = .33, \text{ ALT SHEAR}$$

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GUIDE RODS (30H60649)

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MAT'L 2219-T87 AL $F_{tu} = 63 \text{ KSI}$, $F_{su} = 38 \text{ KSI}$

LOAD = 1560 lb (ref 4.76)

@ 3" from base

ROD DIA = 1"

$$I = \frac{\pi(1)^4}{64} = .049 \text{ in}^4$$

$$f_b = \frac{1560(3)(.5)}{.049} = 47.76 \text{ KSI}$$

Plastic bending (ref p. 4.27)

$$F_{bu} = 47 \text{ KSI}$$

$$111. S. = \frac{17}{2(47.76)} - 1 = .015 \text{ PLASTIC BENDING}$$

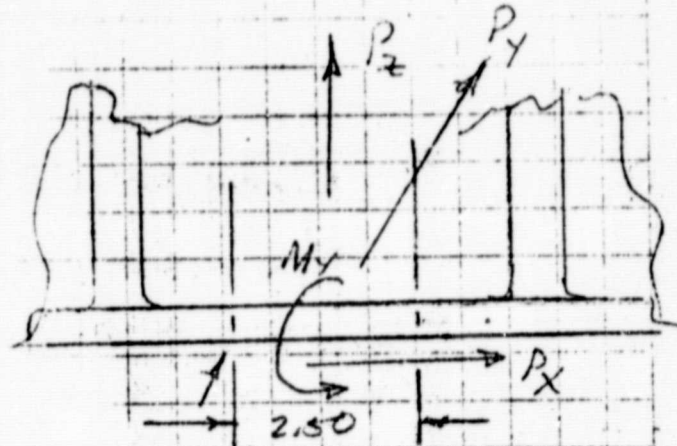
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LEM BASE MOUNTING BOLTS

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(NASTRAN RUN NO. 3111)
From p. 54

$$\begin{aligned}
 P_x &= +5.07 \text{ Kips} \\
 P_y &= 5.79 \text{ Kips} \\
 P_z &= -10.64 \text{ Kips} \\
 M_y &= 14.4 \text{ in-kips}
 \end{aligned}$$



$$\begin{aligned}
 P_t &= \frac{P_z}{2} \pm \frac{M_y}{2.5} \\
 &= \frac{10.64}{2} \pm \frac{14.4}{2.5}
 \end{aligned}$$

$$11.08 \text{ KIPS/BOLT}$$

NAS 1960 C BOLTS
2 PLCS (5/8 DIA)
P_{allow} = 49,000 LBS
P_{SA} = 33,150 LBS / SINGLE SHU.

SHEAR

$$\begin{aligned}
 P_s &= \left[\left(\frac{P_x}{2} \right)^2 + \left(\frac{P_y}{2} \right)^2 \right]^{1/2} = \left[\left(\frac{5.07}{2} \right)^2 + \left(\frac{5.79}{2} \right)^2 \right]^{1/2} \\
 &= 3.85 \text{ KIPS/BOLT}
 \end{aligned}$$

$$R_s = \frac{2(3.85)}{33.15} = .23$$

$$R_t = \frac{2(11.08)}{49} = .45$$

INTERACTION - (REF 4.9.1)

$$M.S.U = \frac{.88}{.45} - 1 = .96$$

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REM BASE / PDM FEATURES

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NAS1960C BOLT 5/8" DIA $P_{tu} = 49000 \text{ lb}$
 $P_{su} = 33150 \text{ lb}$

FROM P 51 . MAX TENSION LOAD IS $P_t = 10.66 \text{ lb}$
 $P_x = 4.26 \text{ lb}$
 $P_y = 5.69 \text{ lb}$

$$P_{s \text{ MAY}} = \sqrt{4.26^2 + 5.69^2} = 7.11 \text{ lb}$$

$$R_t = \frac{(2)(10.66)}{49.0} = .435$$

$$R_s = \frac{(2)(7.11)}{33.15} = .43$$

$$M.S.u = \frac{.71}{.435} - 1 = .63 \text{ ULT INTERACTION}$$

FOR NUT - MS21045C 5/8-18UNF-3B
(per M. Lieberman)

$$P_{tu} = 34130 \text{ lb}$$

$$M.S.u = \frac{34.13}{(2)(10.66)} - 1 = .60 \text{ ult tension}$$

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REM ADAPTER STRESS SUMMARY

FEI
NASTIKORUN
N0003111
4/24/84

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LOAD CASE	FLY	STRESS (KSI)	
		MAX	MIN
1	78	4.10	-1.54
	23	6.14	-3.29
2	3	6.04	-3.22
	6	7.10	-4.80
	18	6.55	-6.98
3	3	4.27	-2.69
	6	6.36	-4.09
	16	9.8	-5.89
	23	9.72	-5.39
4	2	8.47	-4.5
	3	9.83	-5.43
5	6	4.88	-7.83
	28	4.86	-4.58
6	2	1.47	-1.92
	6	2.92	-4.21
	11	4.85	-7.85
	23	4.05	-7.44
	28	4.19	-6.46
7	3	2.17	-3.94
	6	3.5	-4.66
	16	3.22	-2.48
	28	6.83	-5.24

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AMPLIFIED STRESS SUMMARY (CONT.)

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LOAD	FILE	STRESS (KSI)	MIN
8	12	2.65	-2.16
	16	3.55	-4.83

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NASTRAN RUN NOC0311
 REM BASE STRESS SUMMARY
 APRIL 24, 1984

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LOAD CASE	ELE	STRESS (KSI)	
		MAX	MIN
1	82	7.08	-5.37
1	32	6.19	-2.85
1	122	6.18	-7.13
1	134	7.06	-7.43
1	138	10.57	-11.14
1	128	13.51	-14.56
1	157	22.95	-22.5
1	158	20.97	-21.4
1	163	15.47	-15.67
2	82	7.42	-4.47
2	83	3.16	-3.01
2	84	2.32	-2.87
2	87	7.16	-2.7
2	157	22.35	-22.65
2	158	21.4	-20.8
2	163	15.6	-15.3
3	82	7.49	-5.57
3	83	2.55	-2.54
3	87	6.72	-3.2
3	157	22.88	-22.43
3	158	21.0	-21.4
3	163	15.4	-15.6
4	82	7.83	-4.45
4	83	2.85	-2.74
4	157	22.4	-22.7
4	158	21.3	-20.8
4	163	15.49	-15.21
5	82	3.83	-4.36
5	83	2.75	-1.999
5	157	12.22	-11.88
5	158	10.96	-11.5
5	163	8.05	-8.3

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REM BASE STRESS SUMMARY (cont.)

LOAD CASE	ELE	STRESS (KSI)	
		MAX	MIN
6	82	4.43	-4.02
6	83	3.26	-1.86
6	157	11.94	-12.37
6	158	11.63	-11.19
6	163	8.45	-8.18
7	82	3.85	-3.95
7	83	2.82	-2.31
7	157	12.18	-11.84
7	158	11.0	-11.55
7	163	8.1	-8.41
8	82	4.22	-3.61
8	83	3.33	-1.85
8	157	12.0	-12.41
8	158	11.6	-11.15
8	163	8.5	-8.25

STRESSES IN ELE 157, 158, 163 ARE
DUE TO CONCENTRATED FORCES ON BASE
AT GRID 132 REPRESENTING THE BASE
MASS. THEREFORE THESE ARE NOT REALISTIC
STRESSES AND WILL NOT BE CONSIDERED.

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REM ADAPTER STRESSES -

FROM STRESS SUMMARY p. 41 & 42
LOAD CASE 4, ELE 3

$$S_{max} = 9.83 \text{ KSI}$$

2219-T87 AL

$$F_{tu} = 62 \text{ KSI}$$

$$M.S.u = \frac{62}{2(9.83)} - 1 = 2.15$$

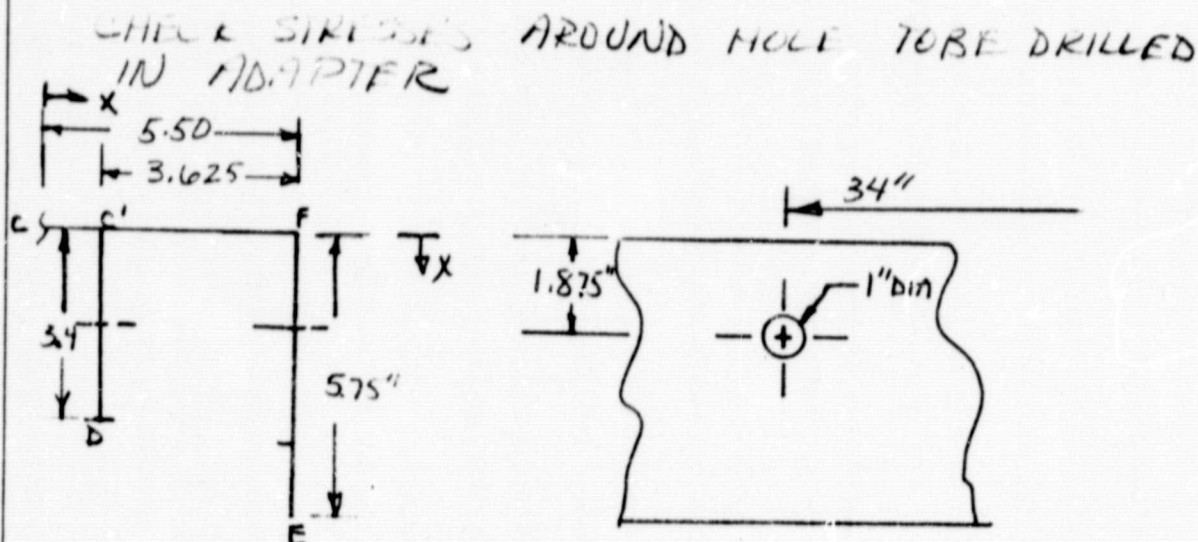
REM BASE STRESSES

FROM SUMMARY p. 43, 44
LOAD CASE 1, ELE 128

$$S_{max} = 13.51 \text{ KSI}$$

$$M.S.u = \frac{62}{2(13.51)} - 1 = 1.29$$

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FROM NASTRAN MODEL ELEMENT 48
STRESSES @ PTS

BENDING -

C 265.3 psi
D 1403.7 psi
E 623.3 psi
F 1453.8 psi

AXIAL

793.2 psi

ASSUME A LINEAR STRESS DISTRIBUTION FROM POINT TO POINT -

STRESS @ PT C'

$$\sigma_B^{C'} = \left(\frac{-1453.8 - 265.3}{5.5} \right) X + 265.3 = -312.6X + 265.3$$

$$@ C' = 5.5 - 3.625 = 1.875$$

$$\sigma_B^{C'} = -320.8 \text{ psi}$$

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ADAPTER HOLE STRESSES (cont.)

STRESS IN SECTION C'-D

$$\sigma_B^{C-D} = \left(\frac{320.8 + 1403.7}{3.4} \right) X - 320.8 = 507.2X - 320.8$$

STRESS IN SECTION F-E

$$\begin{aligned} \sigma_B^{F-E} &= \left(\frac{1453.8 + 623.3}{5.75} \right) X - 1453.8 \\ &= 361.2X - 1453.8 \end{aligned}$$

$$\sigma_{TOTAL} = \sigma_B + \sigma_{AXIAL}$$

FOR STRESSES @ HOLE EDGE, $X = 1.875 - R$
 $= 1.875 - .5 = 1.375$

$$\begin{aligned} \sigma_B^{C-D} &= 507.2(1.375) - 320.8 \\ &= 376.6 \text{ psi} \end{aligned}$$

$$\begin{aligned} \sigma_B^{F-E} &= 361.2(1.375) - 1453.8 \\ &= -957.15 \text{ psi} \end{aligned}$$

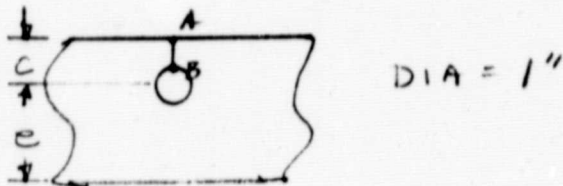
TOTAL STRESS

$$\sigma_T^{C-D} = 376.6 + 793.2 = \underline{1166.8 \text{ psi}} \quad (C'-D)$$

$$\sigma_T^{E-F} = -957.15 + 793.2 = \underline{-163.95 \text{ psi}} \quad (E-F)$$

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ADAPTER HOLE STRESSES (cont.)



FOR SECTION C'-D (INNER WALL)

$$e = 1.875"$$

$$C = 3.4 - 1.875 = 1.525$$

$$A - B = C - \frac{1}{2} = 1.375$$

$$e/C = \frac{1.875}{1.525} = 1.23$$

$$r/C = \frac{.50}{1.875} = .2667$$

FROM GRAPH P. -

$$\frac{\sigma_{MAX}}{\sigma} = \underline{3.26}$$

FOR SECTION E-F (OUTER WALL)

$$C = 1.875$$

$$e = 5.75 - 1.875 = 3.875$$

$$A - B = 1.375$$

$$e/C = \frac{3.875}{1.875} = 2.067$$

$$r/C = .5/1.875 = .2667$$

FROM GRAPH P

$$\frac{\sigma_{MAX}}{\sigma} = \underline{3.24}$$

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ADAPTER HOLE STRESSES (CONT.)

USING STRESS CONCENTRATIONS -

SECTION C-D

$$T_{MAX} = 3.26 \sigma$$

$$\sigma_T^{C-D} = 1166.8 \quad \therefore T_{MAX} = 3.26(1166.8) = \underline{3803.8 \text{ psi}}$$

SECTION E-F

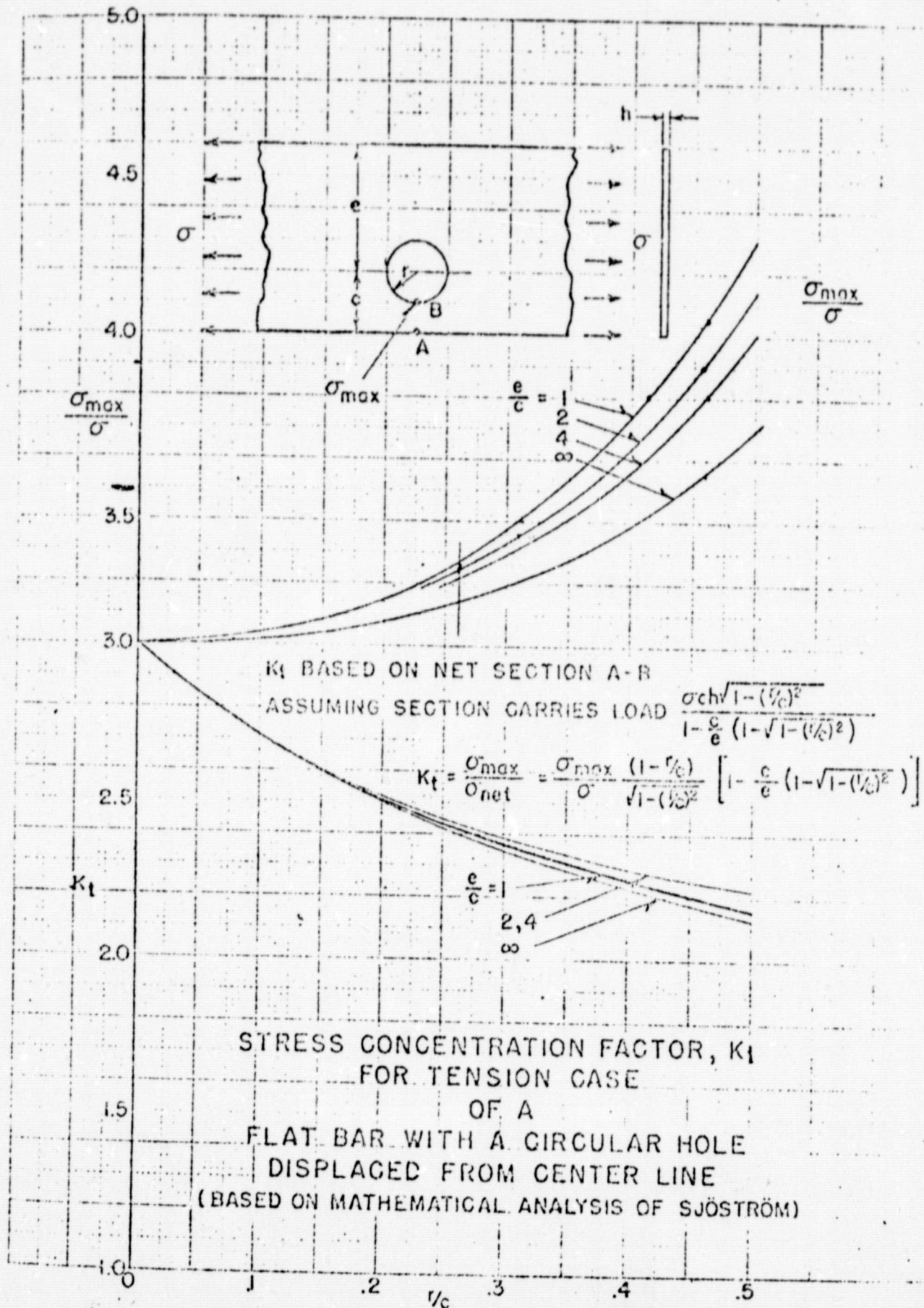
$$T_{MAX} = 3.24 \sigma$$

$$\sigma_T^{E-F} = 163.95 \text{ psi}$$

$$\therefore T_{MAX} = 3.24(163.95) = \underline{531.20 \text{ psi}}$$

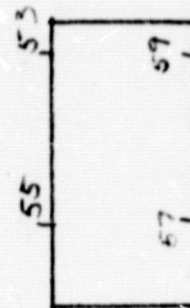
$$M.S.U = \frac{63000}{(2)(3803.8)} - 1 = \underline{7.28}$$

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		Title: SPARTAN REM			

LOAD CASE	REM PIN REACTIONS (KIPS)											
	53				55				57			
	F _x	F _y	F _z	F _x	F _y	F _z	F _x	F _y	F _z	F _x	F _y	F _z
1	-7.89	-2.21	1.02	0	-5.12	11.69	0	0	5.72	3.58	0	.52
2	7.91	-5.14	11.73	0	-.19	.98	0	0	.56	-3.54	0	5.68
3	-7.96	5.19	-2.37	0	.14	8.44	0	0	8.96	3.52	0	3.91
4	7.85	.26	8.34	0	5.07	-2.26	0	0	3.81	-3.62	0	9.07
5	-7.86	-2.25	-6.17	0	-5.09	4.45	0	0	-1.59	3.61	0	-6.83
6	7.95	-5.19	4.46	0	-.14	-6.4	0	0	-6.80	-3.52	0	-1.78
7	-7.93	5.16	-9.47	0	.17	1.34	0	0	1.71	3.54	0	-3.39
8	7.88	.22	1.16	0	5.11	-9.5	0	0	-3.5	-3.59	0	1.71



REF: NASTRAN RUN N0003111 4/24/84

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REM BASE REACTIVES

DSM 5/84
EJW 6/84

SPARTAN REM

452

PT	Force	LOAD CASE							
		1	2	3	4	5	6	7	8
201	F _x	-3.74	1.34	-3.78	1.3	-1.87	3.21	-1.90	3.18
	F _y	2.65	2.21	4.21	3.71	-2.28	-2.72	-7.79	-1.16
	F _z	6.85	.80	10.15	4.1	-1.56	-7.66	1.79	-4.3
203	F _x	-1.46	3.85	-1.35	3.74	-3.38	1.93	-3.27	2.04
	F _y	2.45	2.81	4.05	4.41	-2.81	-2.46	-1.21	-.86
	F _z	.77	6.81	4.23	10.3	-7.7	-1.72	-4.19	1.79
205	F _x	-3.3	5.59	-3.34	5.52	-4.99	3.87	-5.07	3.79
	F _y	-4.1	-7.4	.58	-2.7	1.13	-2.16	5.79	2.5
	F _z	1.09	13.26	-2.34	9.8	-7.3	4.81	-10.64	1.44
207	F _x	-4.8	2.49	-4.8	2.5	-3.0	4.26	-3.03	4.26
	F _y	-7.2	-3.84	-2.66	.70	-2.2	1.16	2.32	5.69
	F _z	13.2	1.05	9.9	-2.26	4.8	-7.48	1.62	-10.6

- LOAD FACTORS APPLIED TO PAYLOAD, ADAPTER & BASE CG's.
- ALL FORCES IN KIPS
- F_z INDICATES TENSION IN FASTENERS

(FROM NASTRAN RUN N0003111 4/24/84)

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FRACTURE ANALYSIS

REM FRACTURE CONTROL SUMMARY

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DRAWING NO.	DESCRIPTION	MATL	CLASSIFICATION		RATIONALE	PFC CAT.		CRITICAL CRACK SIZE (IN)	INSPECTION	
			EXP	PFC		NRC	FC		LIMIT (IN)	TYPE
30A60641	ADAPTER	2219-T87AL		X	REDUNDANT LOAD PATH	X				DP
30A60681	BASE	2219-T87AL	X		REDUNDANT LOAD PATH					
30A60697	BELCRANK SHAFT	A286 CRES		X	LOW STRESSES	X				ET
30A60696	BELCRANK	4340 STL		X				.415	.25	MP
30A60699	TURNBUCKLE	4340 STL		X	AT ONLY 5% OF PRELOAD	X				
30A60695	TOGGLE LINK	4340 STL		X				.23	.05	ET
30A60700	TOGGLE LINK PIN	INCONEL 718		X				.104	.1	ET
30A60694	TOGGLE LINK LUG	INCONEL 718		X				.105	.05	ET
30A60698	LUG PIN	INCONEL 718		X				.146	.10	ET
30A60690	LATCH HOOK	INCONEL 718		X	LOW COMPRESSIVE LOAD	X				
30A60642	ROUND PIN	INCONEL 718		X				—*	.1	ET
30A60645	ROUND PIN PLATE	INCONEL 718		X				1.065	.05	ET
30A60643	SQUARE PIN	INCONEL 718		X				—*	.1	ET
30A60644	SQUARE PIN PLATE	INCONEL 718		X				.98	.05	ET
NA519600	MOUNTING BOLTS	A286 CRES	X		REDUNDANT LOAD PATH					
30A60702	LINK HOUSING	4340 STL		X	LOW STRESSES	X				
30A60703	LINK ROD	4340 STL		X		X				
30A60647	ROD HOLDER - 1 & 4	2219-T87AL		X		X		—*	—	NONE
30A60648	ROD HOLDER - 2 & 3	2219-T87AL		X		X		—*	—	NONE
30A60649	LOCATOR ROD	2219-T87AL		X				.18	.10	ET
30A60684	AFT GUIDE - 1	2219-T87AL	X		LOW COMPRESSIVE LOADS					
30A60685	AFT GUIDE - 2	2219-T87AL	X							
30A60686	POST GUIDE - 3	2219-T87AL	X							
30A60687	POST GUIDE - 4	2219-T87AL	X							
30A60701	GUIDE BAR	2219-T87AL	X							
NA51955C	PLATE BOLTS	A286 CRES	X		REDUNDANT LOAD PATH					
NA51955C	GUIDE HOLDER BOLT	A286 CRES	X		LOW COMPRESSIVE LOADS					

* NO CRACK GROWTH

ET - EDDY CURRENT

MP - MAGNETIC PARTICLE

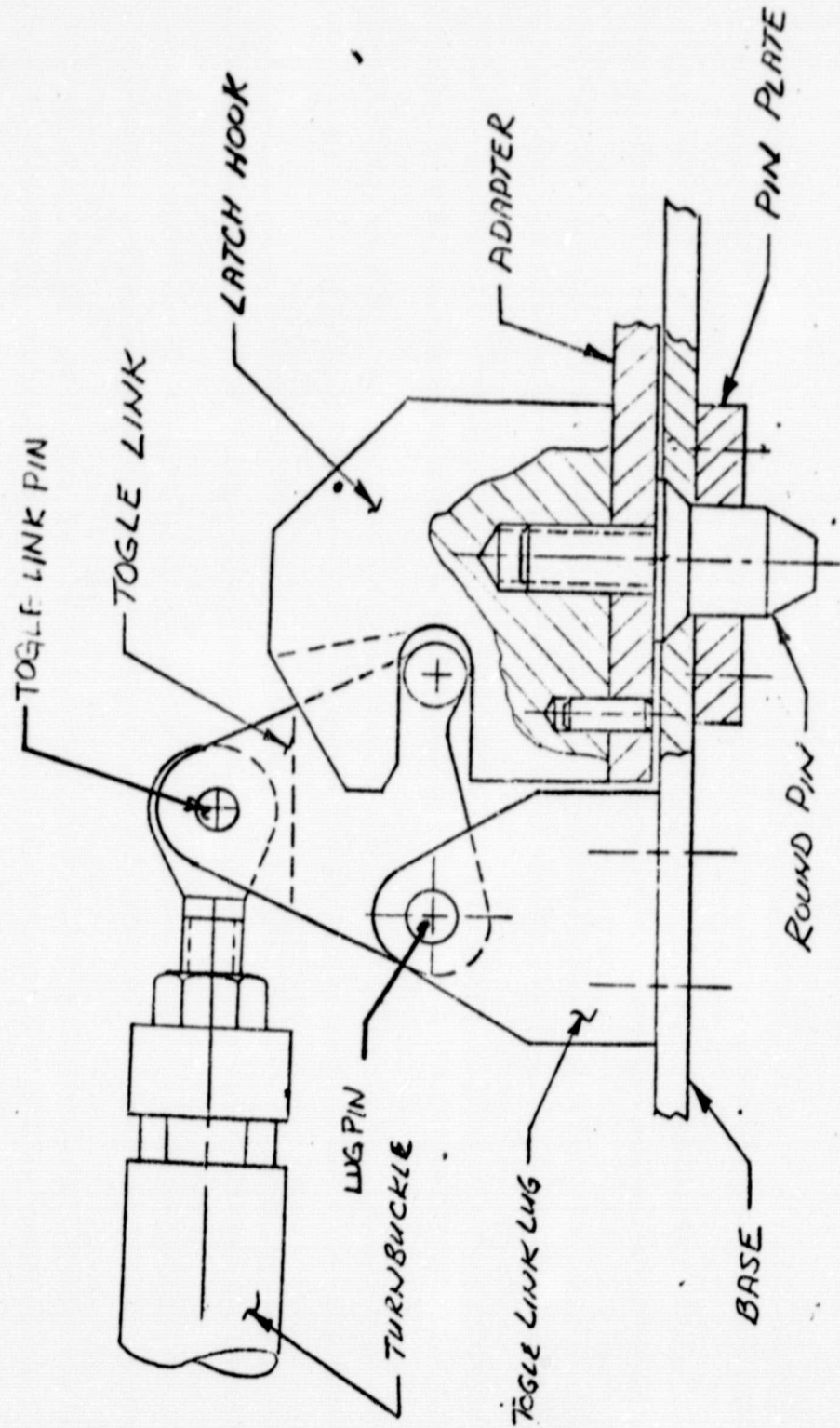
REM FRACTURE CONTROL SUMMARY

ORIGINAL PAGE IS
OF POOR QUALITY

[illegible]

* NOT DONE.

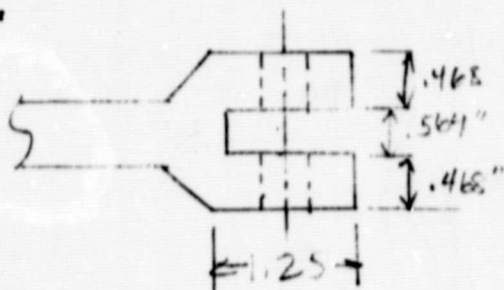
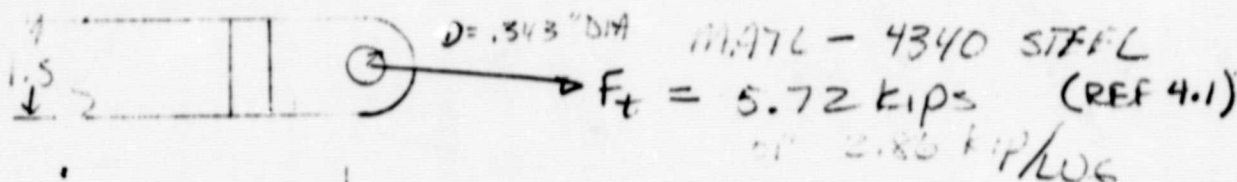
LATCH MECHANISM ASSY



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Prepared by: <u>DCM</u>	Date: <u>6/84</u>	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page: <u>5.1</u>	Temp.	Perm.
Checked by: <u>g/c</u>	Date: <u>6/84</u>	Title: <u>SPARTAN REF</u>	Model		
Approved by:	Date:		Report No.		

FRONT VIEW ANALYSIS
REFLECTOR LUG, (30100696)

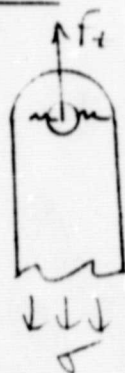


ASSUME $F_{MIN} = 0 \text{ KST}$
(CONSERVATIVE)

LOADED 26000 cycles
(4 MISSIONS WITH S.F. = 4)

CHECKING FOR CRACKS OFF HOLE

MODEL



WORST CASE DBL CRACKS

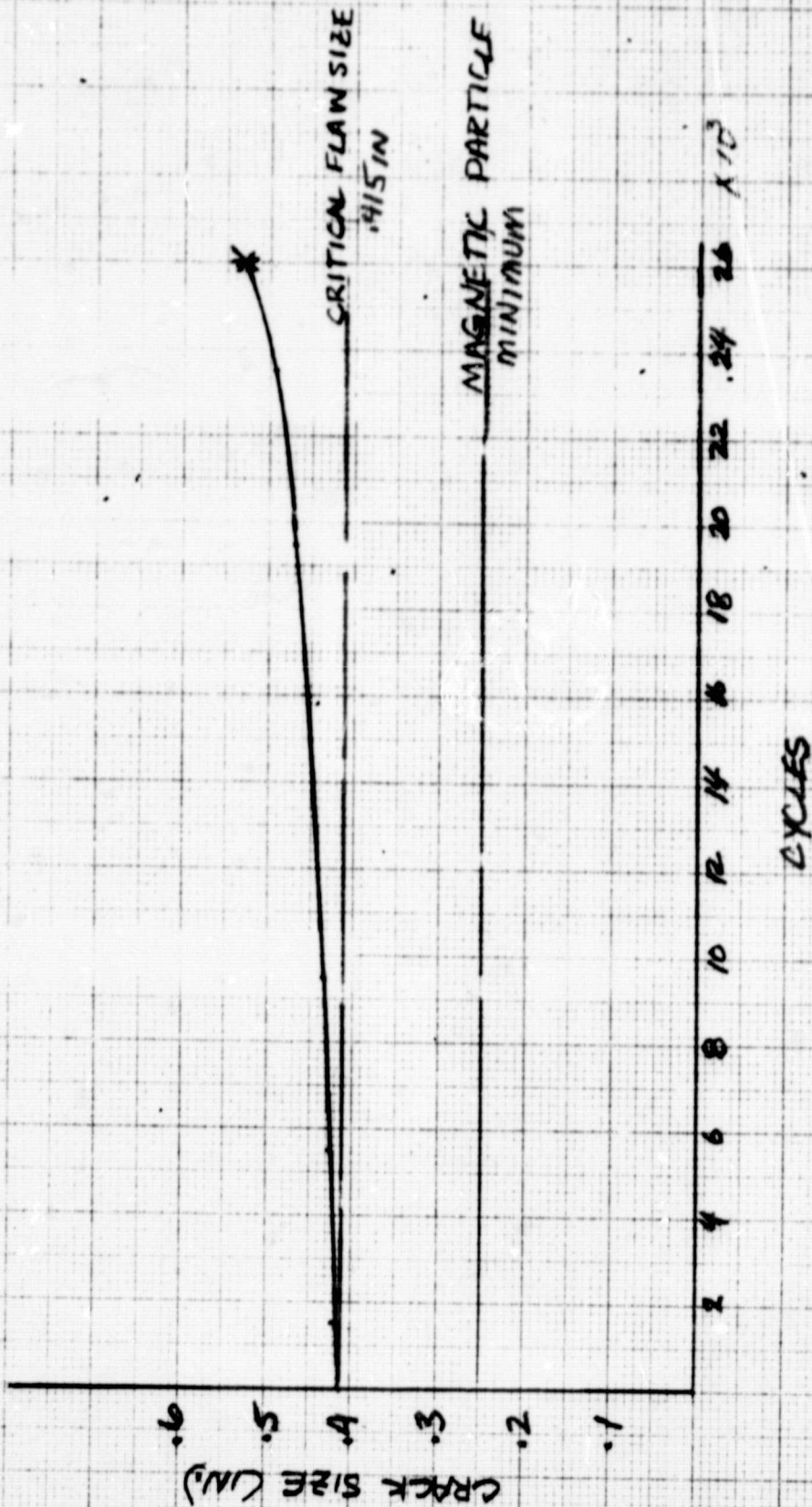
FROM COMPUTER ANALYSIS -
 $CF = .415 \text{ IN}$

INSPECTION - MAGNETIC PARTICLE
LIMITS - .25 IN

$CF > \text{INSP. LIMIT}$

$.415 > .25$

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BELLCRANK LUG (30A600696) - DOUBLE CRACK IN HOLE

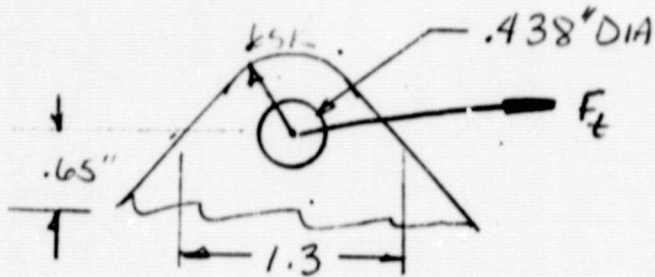
ORIGINAL PAGE IS
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Checked by: ECU	Date: 6/84	Title: SPARTAN REM	Model:		
Approved by:	Date:		Report No.:		

FRACTURE ANALYSIS —

TIGLE LINK (2A60695)

$$t = .313''$$



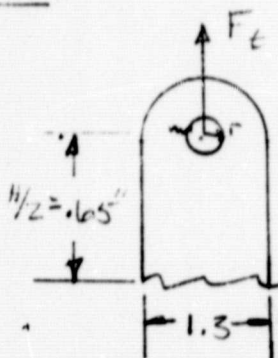
LUG HREN

MATL 4340 STL
LOADED 26000 CYCLES
- 4 MISSIONS WITH
S.F. = 4

FRONT STRESS ANALYSIS

FROM P 4.10, $F_T = 5.72 \text{ kips} \rightarrow \text{LOAD/LUG} = 2.86 \text{ kips}$
ASSUME $P_{MIN} = 0$ (CONSERVATIVE)

MODEL



$$r = .219''$$

$$t = .313$$

CHECK — CRACK OFF HOLE

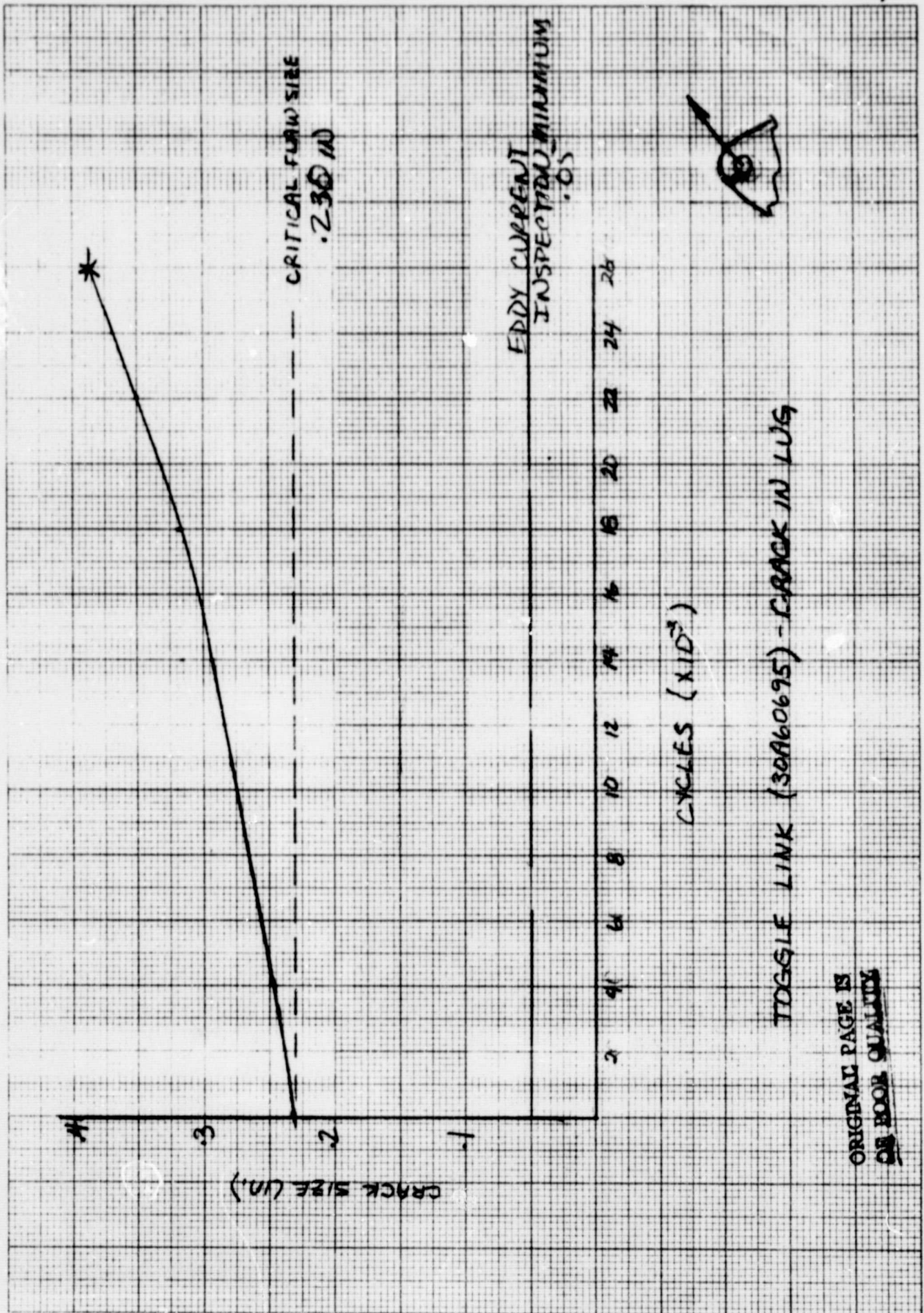
FROM COMPUTER ANALYSIS
 $CF = .230 \text{ in.}$

INSPECTION — EDDY CURRENT
MINIMUM SIZE — .05 IN

$$CF > \text{INSP. LIMIT}$$

$$.23 > .05$$

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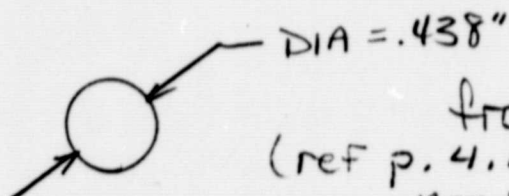


Prepared by: DSM	Date: 6/84	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page: 5.5	Temp.	Perm.
Checked by: E/W	Date: 6/84	Title: SPARTAN REM	Model		
Approved by:	Date:	Report No.			

FRACTURE ANALYSIS

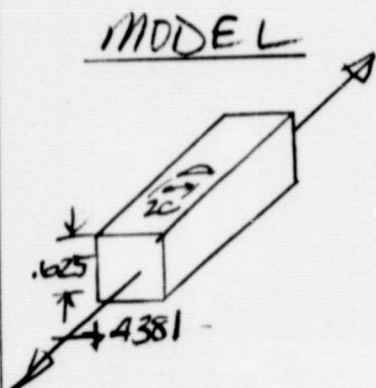
TOGGLE LINK PIN (30A60700)

MATL - INC 718



FROM STRESS ANALYSIS -
(ref p. 4.14) $\sigma_{MAX} = 92.78 \text{ KSI}$
ASSUME $\sigma_{MIN} = 0 \text{ KSI}$ (CONSERVATIVE)

LOADED 26000 cycles (INCLUDES 4 MISSIDAS)
WITH S.F. = 4



ASSUME ROD IN TENSION
check for surface flaws

from computer ANALYSIS
 $\frac{1}{2}CF = .052$
or $CF = .104$

INSPECTION -
EDDY CURRENT
LIMITS - .1 IN MINIMUM

$$\underline{.104 > .1}$$

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100% HALF CIRCLE SIZE (IN)

20

15

1

0.5

HALF CRITICAL
FLAW SIZE

EDDY CURRENT
INSPECTION
MIN.

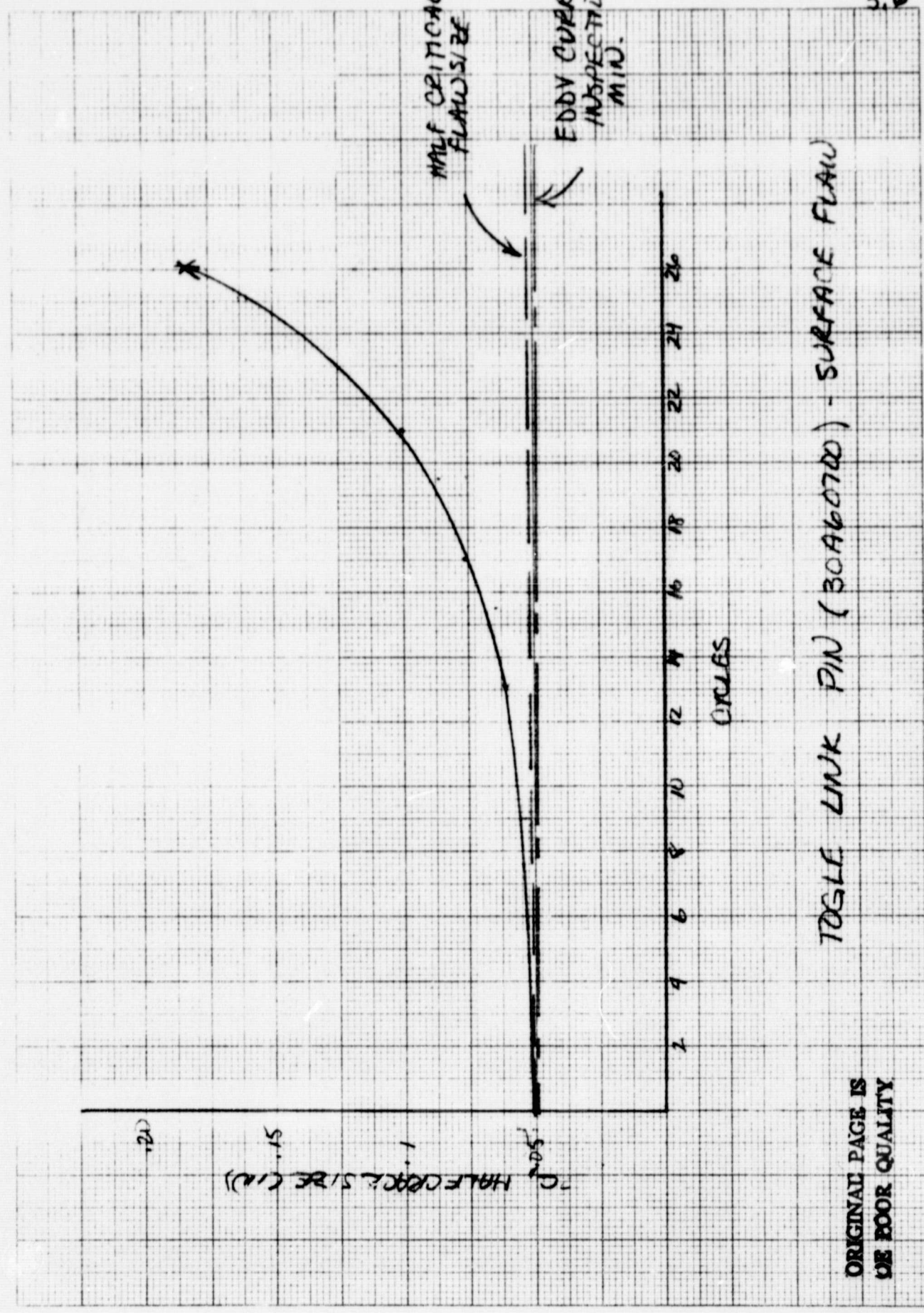
2 4 6 8 10 12 14 16 18 20 22 24 26

CYCLES

TOGGLE LINK PIN (30A60700) - SURFACE FLAW

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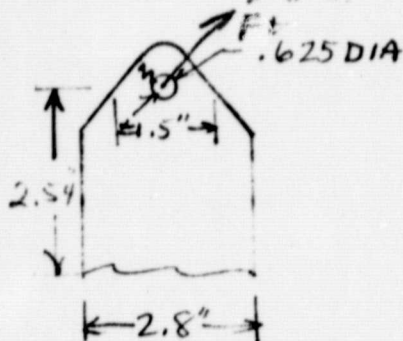
5.6



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Approved by:	Date:		Report No.:

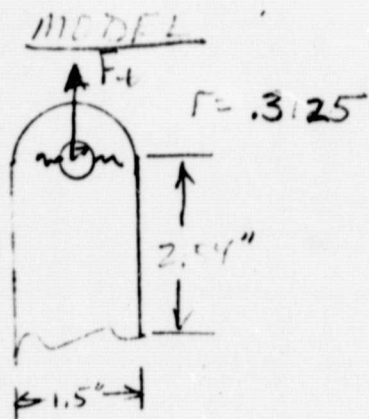
FRACTURE ANALYSIS

TOGGLE LUG (30A60694)



MATL - INC 718
 (CONSERVATIVELY LET ONE LUG
 CARRY FULL REACTION)
 $F_{MAX} = 7.66$ KIPS (p 4.4)
 ASSUME $F_{MIN} = 0$ (CONSERVATIVE)

LOADED 26000 CYCLES -
 4 MISSION WITH S.F. = 4



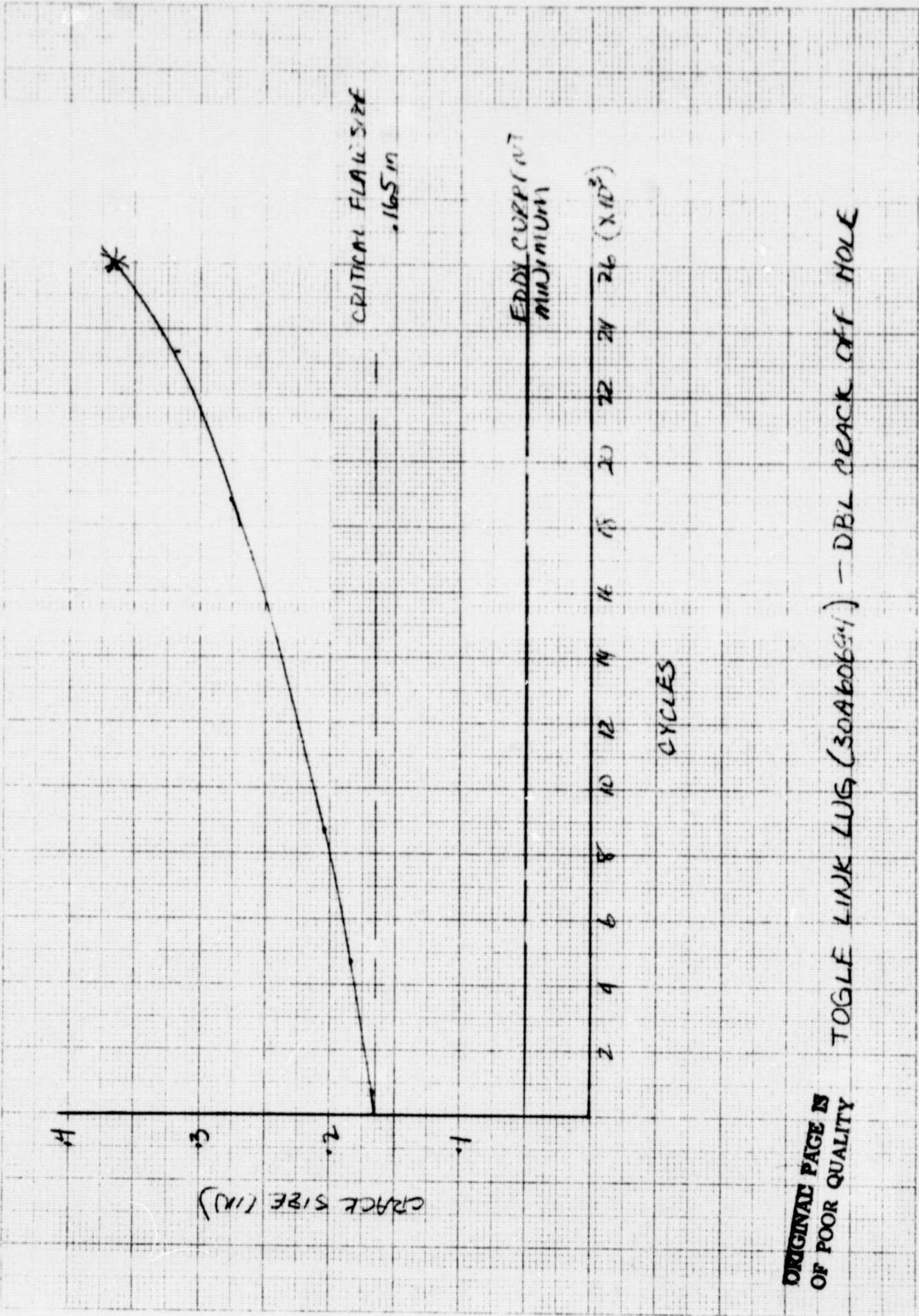
CHECK FOR CRACKS
 OFF HOLE

FROM COMPUTER ANALYSIS -
 $CF = .165$ IN

INSPECTION - EDDY CURRENT
 MIN SIZE - .05 IN

$CF > INSP. LIMIT$
 $.165 > .05$

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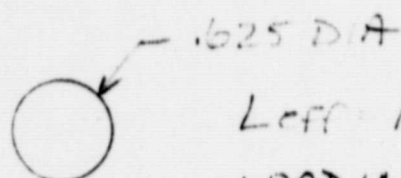


TOGGLE LINK LUG (30A60654) - DBL CRACK OFF HOLE

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Checked by: EPC	Date: 6/84	Title: SPINNING PEAN	Model		
Approved by:	Date:		Report No.		

FRACTURE ANALYSIS LUG PIN (30A60698)



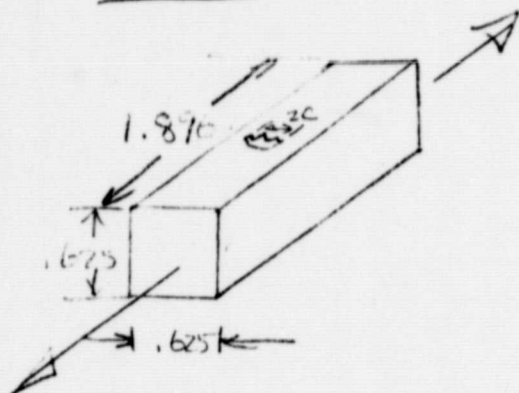
$L_{eff} = 1.64 \text{ in}$

MATL - 718 TMC
S.F. = 4

LOADING IS FOR 4 MISSIONS @ 50 Hz
WITH SF=4, OR 26000 cycles

(ref. p 4.15.2) $\sigma_{max} = 72.8 \text{ KSI}$
ASSUME $\sigma_{min} = 0 \text{ KSI}$
(CONSERVATIVE)

MODEL -



ASSUME ROD IN
TENSION

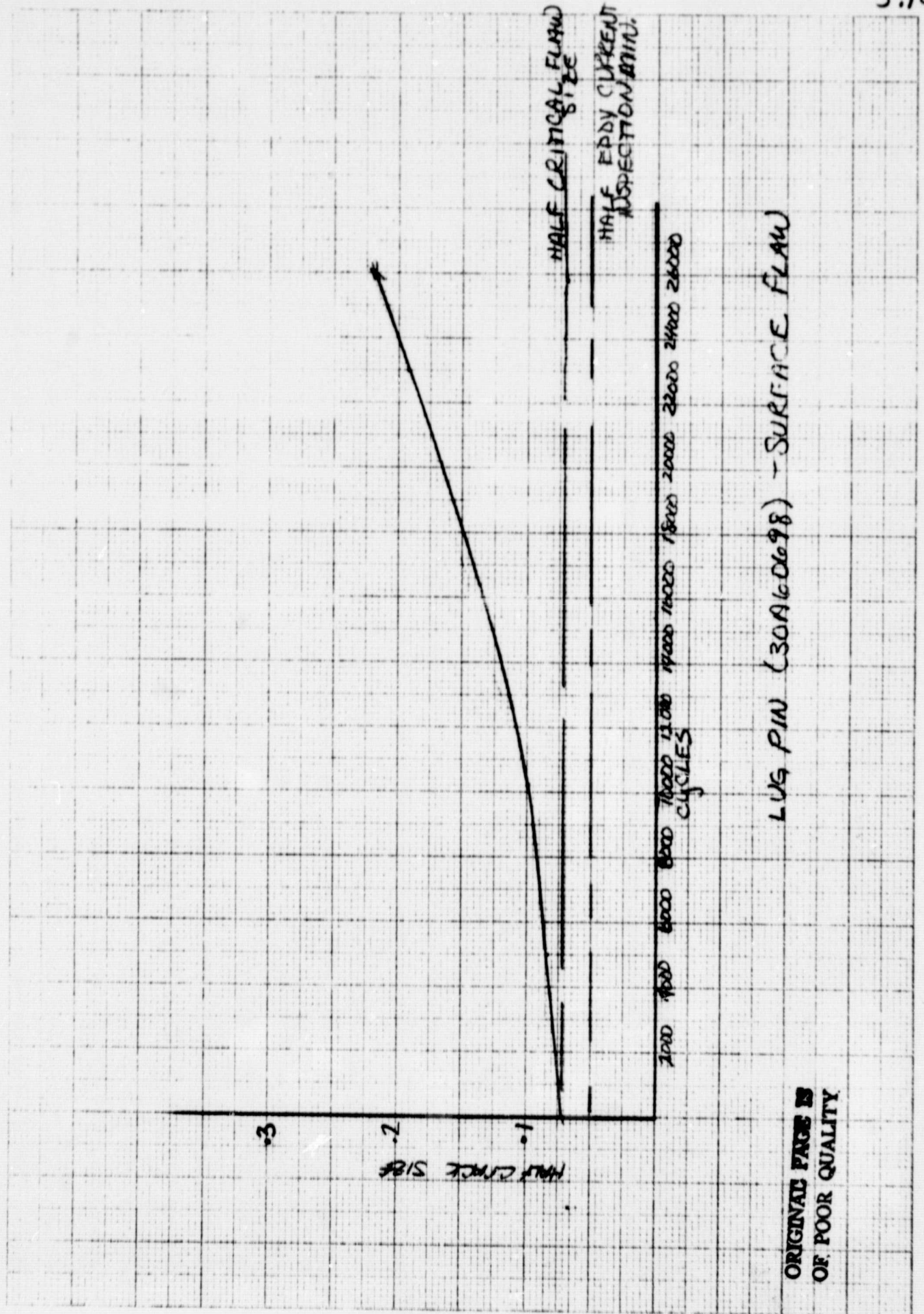
CHECK FOR SURFACE
FLAWS
FROM COMPUTER ANALYSIS
 $2CF = .073$
OR .146

INSPECTION DONE WITH EDDY
CURRENT -

MIN LIMITS ARE .1 IN

.146 > .1

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Checked by: E2W	Date: 6/84	Title: SPARTAN REM	Model		5.2
Approved by:	Date:		Report No.		

FRACTURE ANALYSIS

ROUND PIN (CONT.)

INSPECTION - EDDY CURRENT
LIMITS - .1 IN

FROM COMPUTER ANALYSIS -

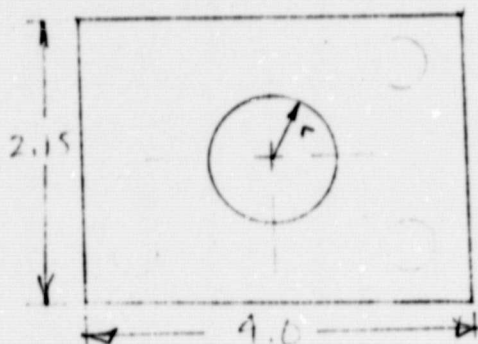
WITH INITIAL FLAWS OF 80% OF
THE PIN DIAMETER THERE WAS NO
CRACK GROWTH AFTER 240,000 cycles

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Checked by: E/W	Date: 6/84		Model		
Approved by:	Date:		Report No.		
		Title: SPARTAN REM			

ROUND PIN PLATE (30160643)



$r = .625$

MATL - TNE 718

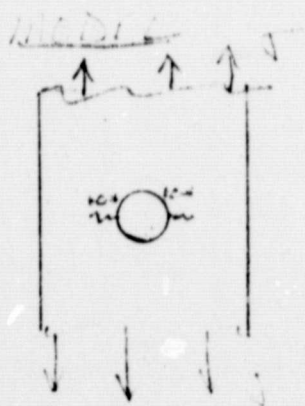
LOADED 26000 lb
(4 MISSILES WITH
S.F. = 4)

FROM STRESS ANALYSIS -

$P_{V \text{ MAX}} = 11.73 \text{ kips}$ (ref p. 4.28)

$A_c = (4.0 - 1.25)(.625) = 1.71 \text{ in}^2$

$\sigma = \frac{P_v}{A_c} = 6.86 \text{ ksi}$, ASSUME $\sigma_{\text{MIN}} = 0$
(conservative)



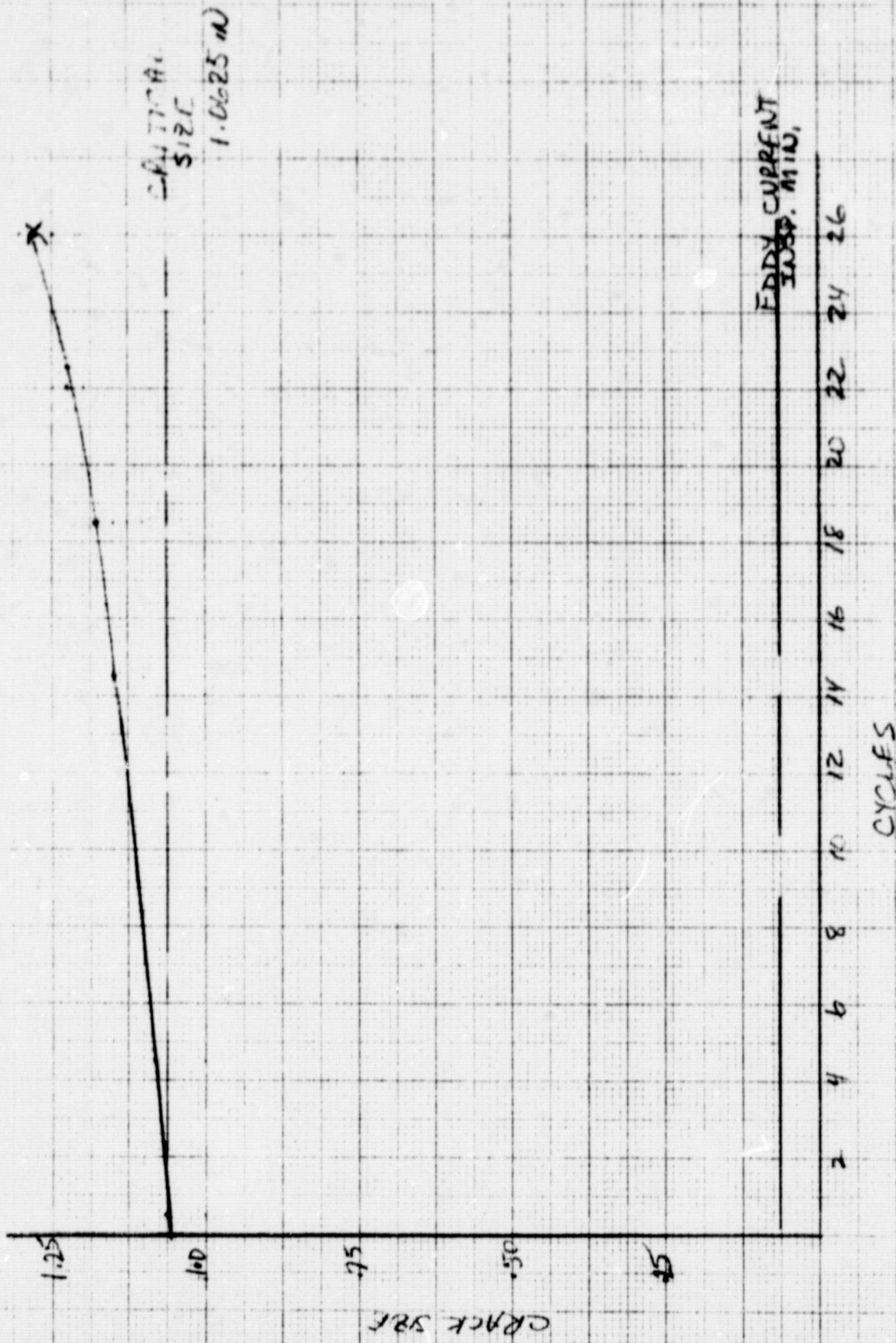
CHECK ROUND PIN
HOLE FOR CRACKS

FROM COMPUTER ANALYSIS -
 $CF = 1.0625 \text{ IN.}$

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INSPECTION - EDDY CURRENT
MINIMUM SIZE = .05 IN

$CF > \text{INSP MIN.}$
 $1.0625 > .05$



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ROUND PIN PLATE - 30A6045

DRIBBLE
BACK OFF ROUND PIN HOLE

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Approved by:	Date:		Report No.:

FRACTURE ANALYSIS

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SQUARE PIN (30A60643)

MATL - INC 718

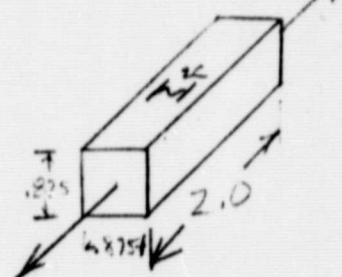
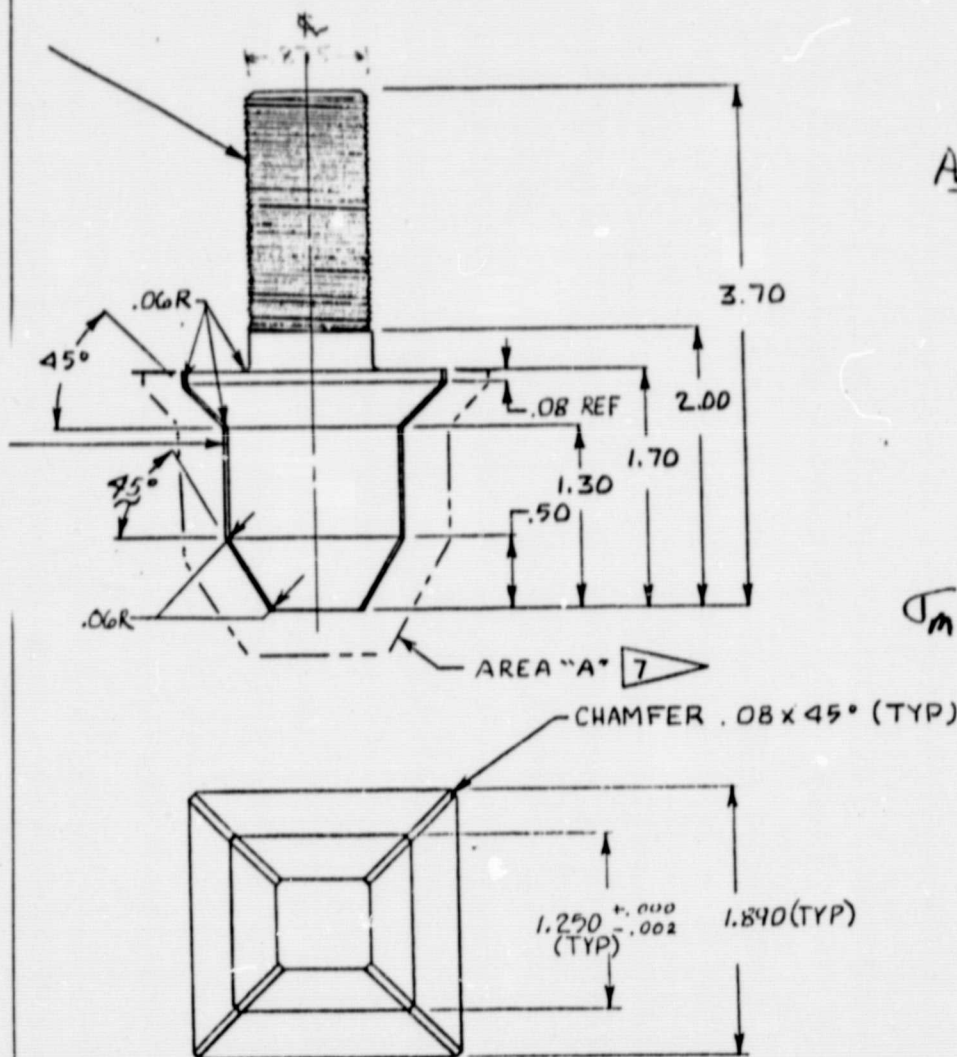
SHANK AREA -
 $A_s = .875 \times .875 = .765^2$

FROM STRESS ANALYSIS
MAY LOAD -
5.19 KIPS. (p 4.28)

$A_s = .875^2$
 $= .762 \text{ in}^2$

$\sigma_{max} = \frac{5.19}{.762} = 6.88 \text{ KSI}$

ASSUME $\sigma_{min} = 0$



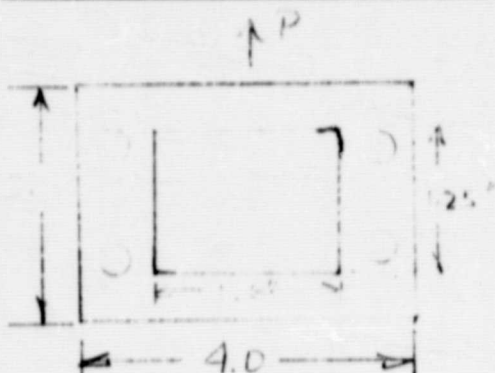
CHECKING FOR SURFACE FLAWS IN SHANK

WITH INITIAL FLAW SIZES OF 80% OF
THE SHANK AREA THERE WAS NO CRACK GROWTH
AFTER 260,000 CYCLES

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Approved by:	Date:		Report No.:

FRACTURE ANALYSIS SPARTAN REM (30A6644)

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MATERIAL: TMC 718

LOADED 26000 cycles
(4 missions with
S.F. = 4)

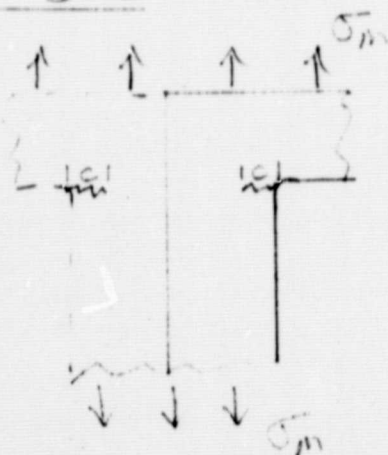
FROM STRESS ANALYSIS -
TENSION

$P_{\text{max}} = 11.69 \text{ Kips}$ (p.51)

$$A_t = (4 - 1.6)(1.25) = 1.5 \text{ in}^2$$

$$\sigma = \frac{P_{\text{max}}}{A_t} = \frac{11.69}{1.5} = 7.79 \text{ KSI} ; \text{ Assume } \sigma_{\text{min}} = 0 \text{ (CONSERVATIVE)}$$

MODEL



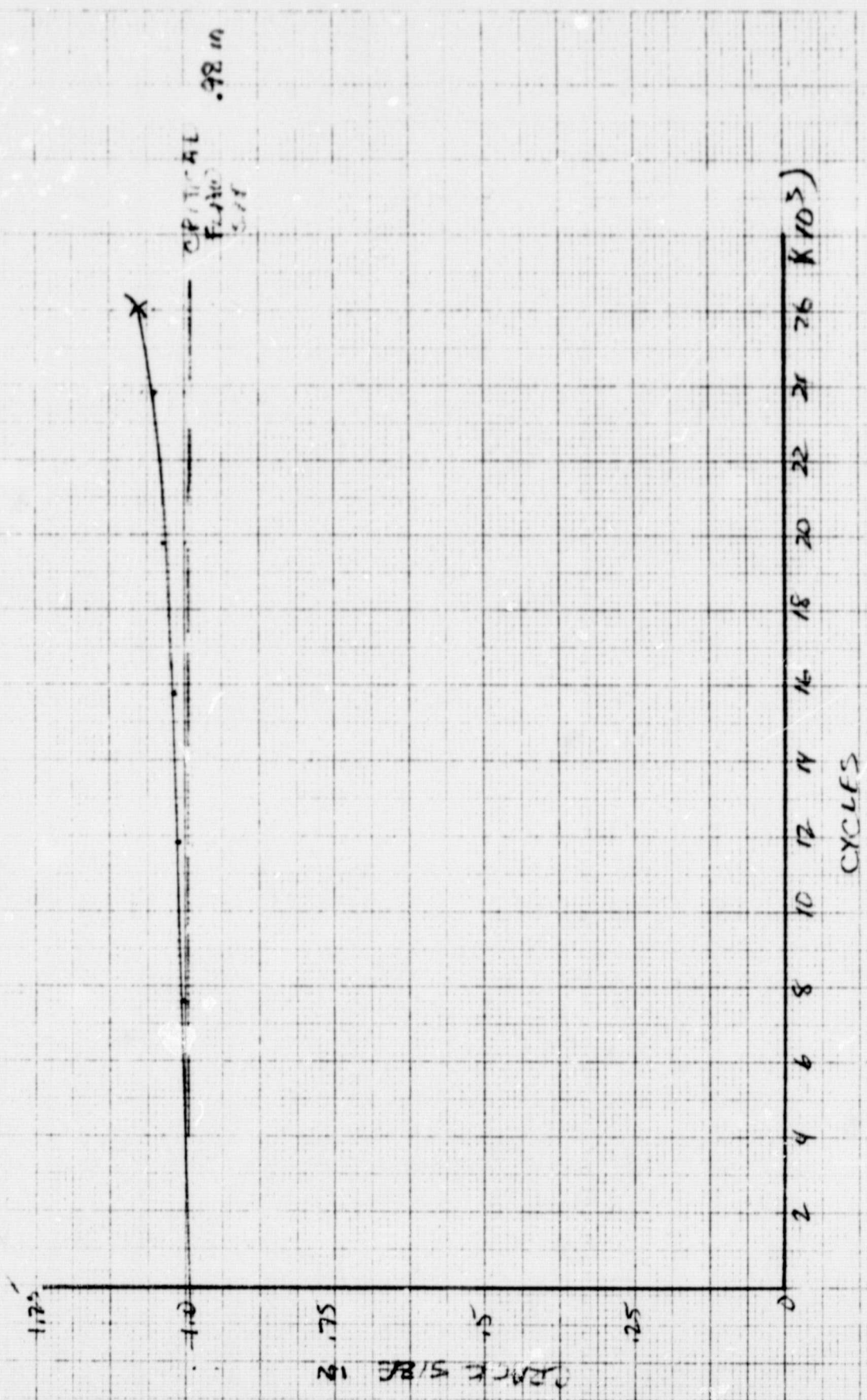
CHECK SQUARE HOLE
CORRECTION FOR CRACKS

FROM COMPUTER ANALYSIS:
 $CF = .98 \text{ IN}$

EDDY CURRENT INSPECTION
MIN SIZE = .05

$CF > \text{INSP. MIN.}$
 $.98 > .05$

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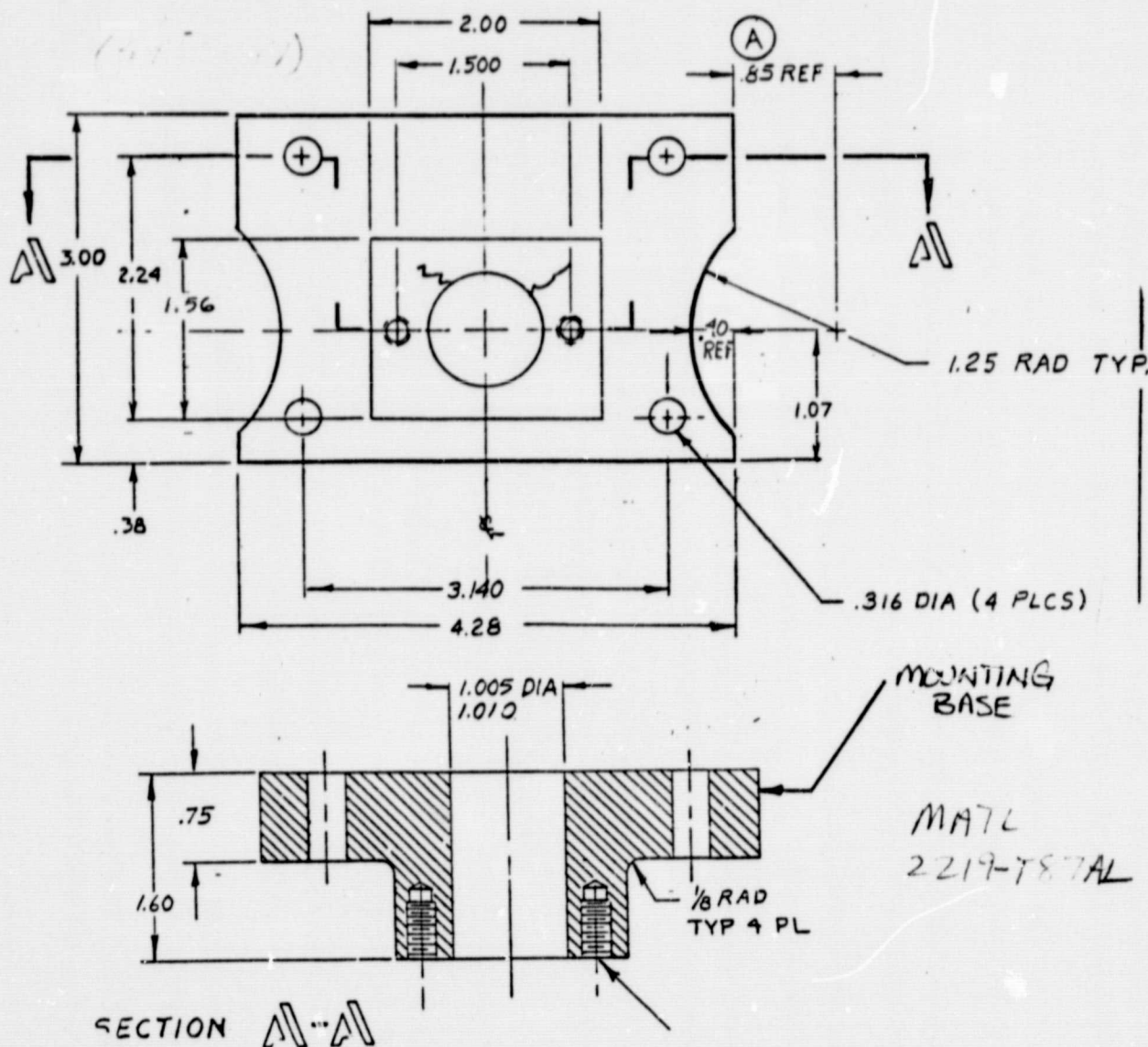
SQUARE PIN PLATE (30A60644) - CRACK AT CORNERS

Prepared by: <u>DM</u>	Date: <u>6/84</u>	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page: <u>5,18</u>	Temp.	Perm.
Checked by: <u>E/C</u>	Date: <u>6/84</u>		Title: <u>SPARTAN REM</u>	Model	
Approved by:	Date:		Report No.		

FRACTURE ANALYSIS

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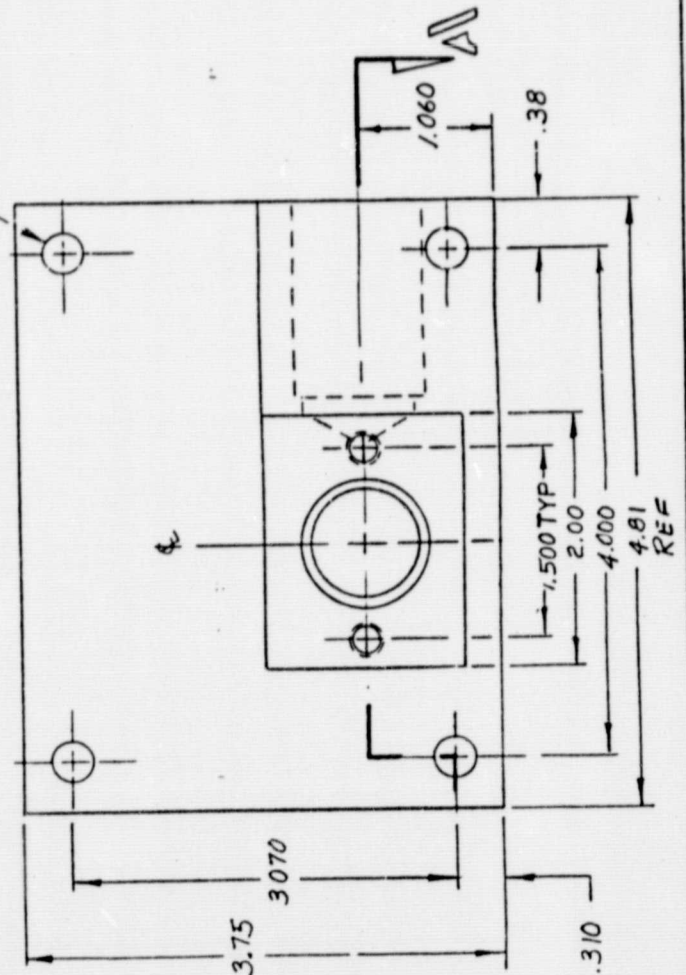
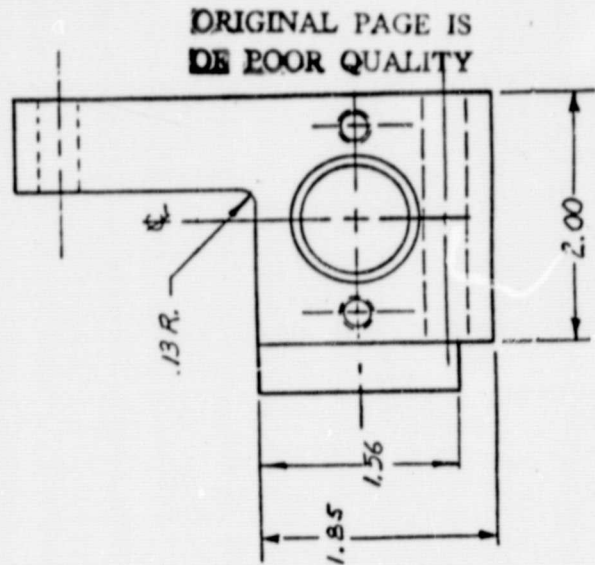
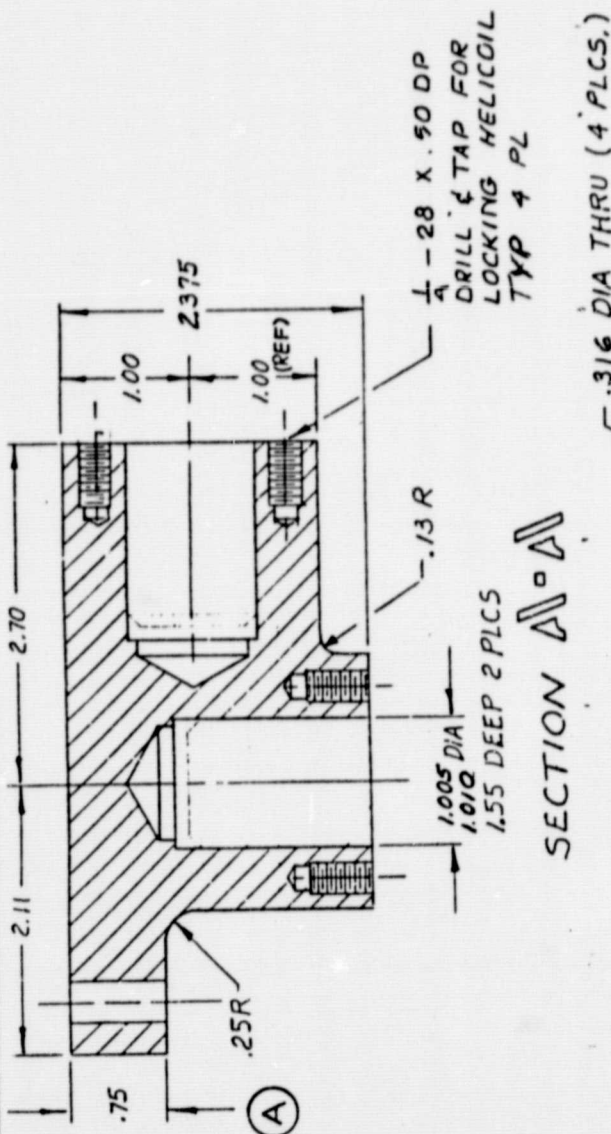
RUD HOLDERS (3046-647, 648)



CRACK OFF RUD MOLE - CHECK
LOAD - BERTHING LOADS
FRONT STRESS ANALYSIS - MAX LOAD = 1560 LB
ASSUME $P_{min} = 0$ (CONSERVATIVE)

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Approved by:		Date			Model			
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30A60648
BASE RCD
POSITIONS 2 & 3



DOUBLE RCD BASE

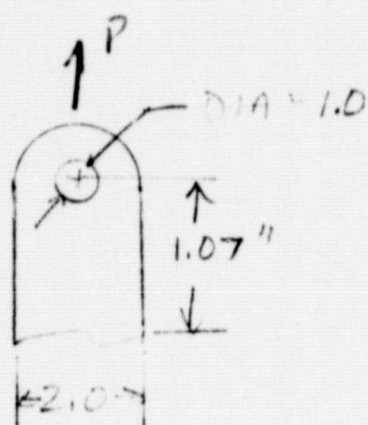
Prepared by: DSM	Date: 6/84	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page: 5.20	Temp.	Perm.
Checked by: E/C	Date: 6/84	Title: SPARTAN REM	Model		
Approved by:	Date:	Report No.			

175 HOLDERS --(CONT)

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WELL

ASSEMBLY IN LOADED LUG



FROM COMPUTER ANALYSIS-

$t = 1.6 \text{ in}$

INITIAL CRACKS
OF 20% OF LUG
AREA SHOW NO
GROWTH

$P = 156016$
ref p 4.36

WELL PDS HOLDER - (30A60645)

SAME AS (30A60647) SINGLE RID FOLDER

NO GROWTH

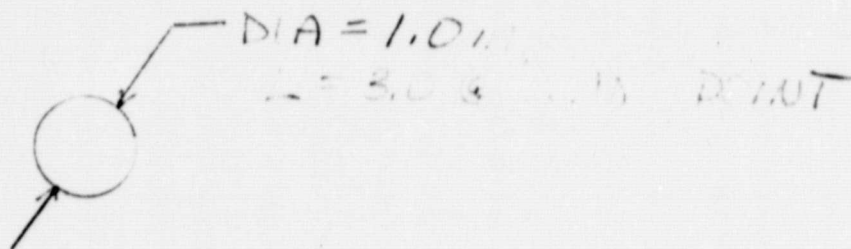
NOTE - NO INSPECTION CURRENTLY DONE
NONE RECOMMENDED

Prepared by: DSU	Date: 6/84	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page: 5.21	Temp.:	Perm.:
Checked by: EPL	Date: 6/84	Title: CARTAN PFI	Model:		
Approved by:	Date:		Report No.:		

FINITE ELEMENT ANALYSIS

1.00000 PTD (30460649)

APPL - 2.000-787 AL



FROM STRESS ANALYSIS - MAX LOAD = 1560 LB

$\sigma = \frac{P}{A}$
 $\sigma = \frac{1560}{.05} = 31200 \text{ PSI}$

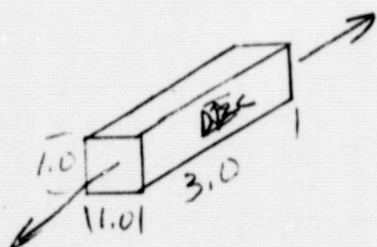
$\sigma_{MAX} = \frac{1560}{.05} = 31200 \text{ PSI}$

ASSUME $\sigma_{MIN} = 0$

$I = \frac{\pi}{64} \cdot \frac{\pi}{4} = .05 \text{ in}^4$

MODEL

ASSUME ROD

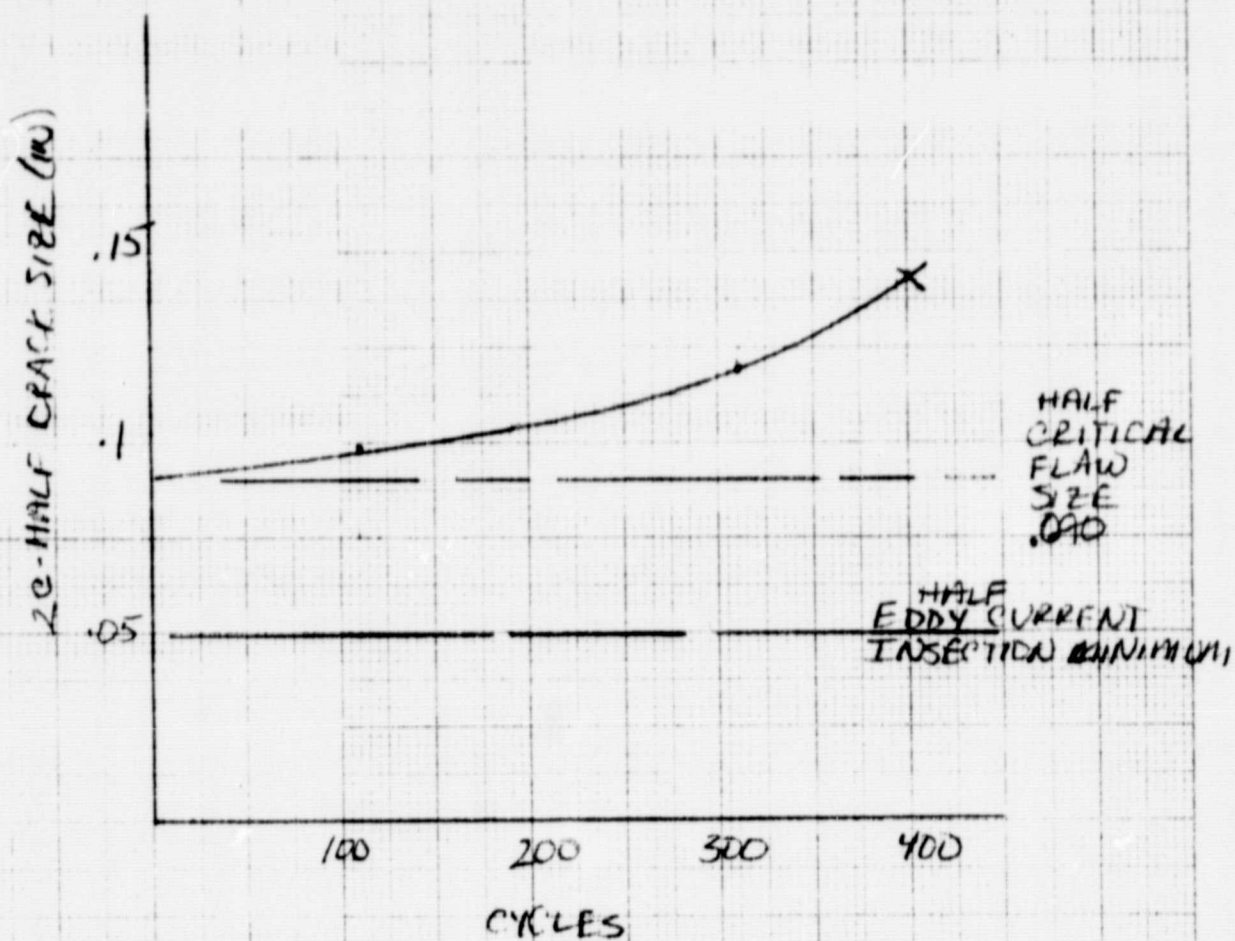


FROM COMPUTER ANALYSIS -
 $\frac{1}{2} CF = .090$
 $\sigma CF = .18 \text{ IN}$

INSPECTION - EDDY CURRENT
LIMITS - .1 IN

$.18 > .1$

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LOCATOR ROD (30A60649)
SURFACE FLAW

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Checked by: E W	Date: 6/84	Title: SPARTAN (PFA)	Model		
Approved by:	Date:		Report No.		

SPARTAN (30A6000)

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MATL - 416 CRES



1.2

FROM STRESS ANALYSIS
 $T_{max} = T_g = 66.78 \text{ KSI}$ (from previous analysis)

CONSERVATIVELY LET

$$T_{min} = 0$$

LIFE FACTOR = 4

$$\text{DESIGN LIFE} = 100 \text{ cycles} \times \text{LF} \\ = 100 \times 4 = 400 \text{ cycles}$$

CHECK CRACK ON TOOTH SURFACE

MODEL

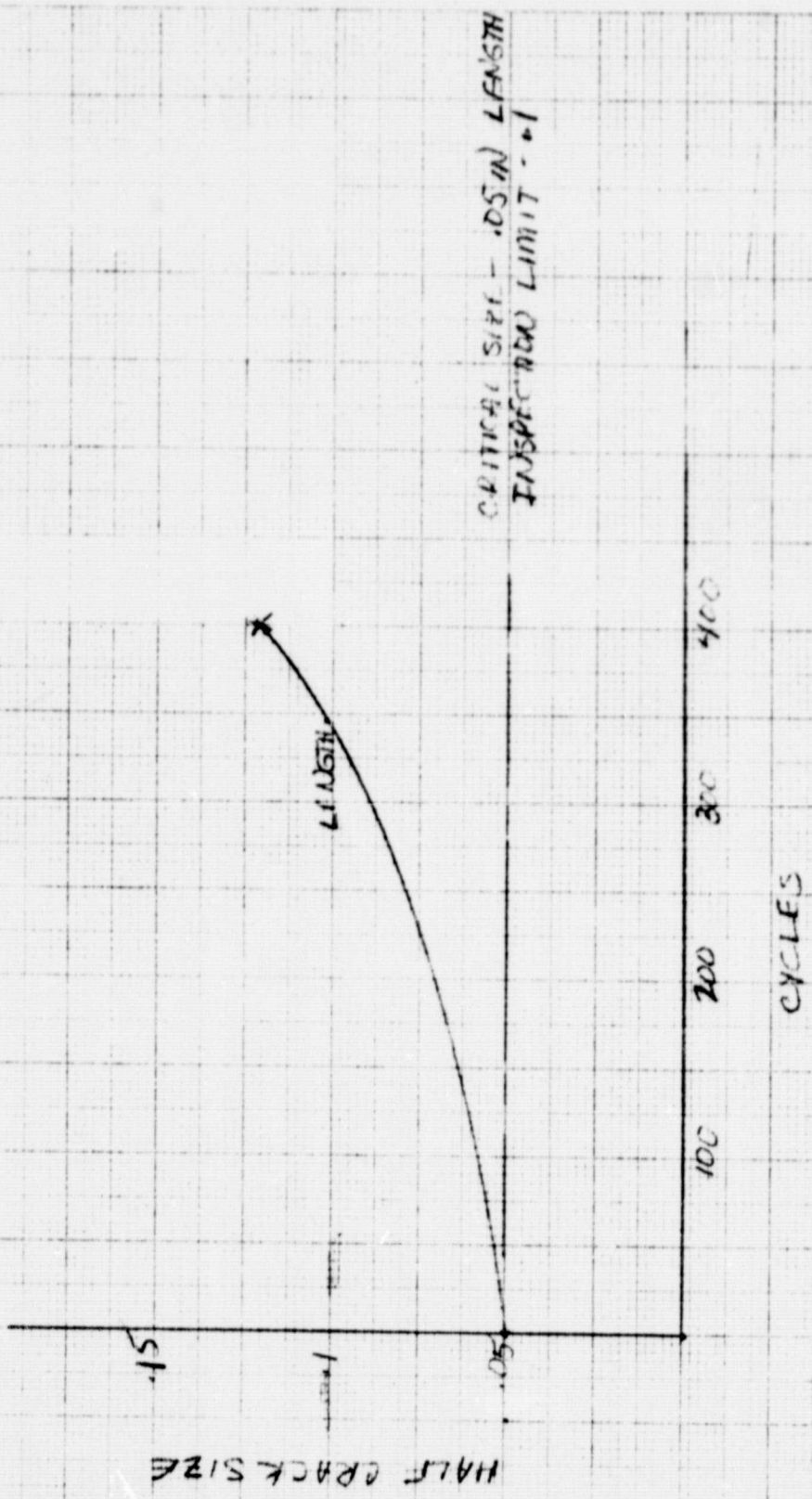


CRITICAL flaw size
 FROM COMPUTER ANALYSIS -
 $2CF = .05 \text{ in.}$
 OR $CF = .1 \text{ in.}$

EDDY CURRENT INSPECTION
 MIN SIZE = .1

$$CF = \text{MIN SIZE} \\ .1 = .1$$

* FROM EARLIER ANALYSIS - LOADS UNCHANGED



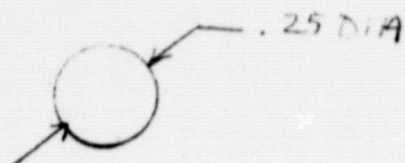
WORM GEAR - 30460079 - FLAW ON TOOTH

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Checked by: EPL	Date: 6/84	Title: SPARTAN REH	Model		
Approved by:	Date:	Report No.			

FRACTURE ANALYSIS

LOCKHEED GEAR PIN (30A60043)

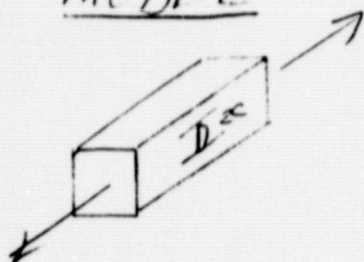


ALATL - MP35N SIL
FROM STRESS ANALYSIS -

$\sigma_{MAX} = 145.3 \text{ KSI} *$
ASSUME $\sigma_{MIN} = 0$ (CONSERVATIVE)

LOAD CYCLES = $100 \times SF$
= $100 \times 4 = 400$

MODEL



ASSUME ROD IN TENSION
CHECK FOR SURFACE FLAW

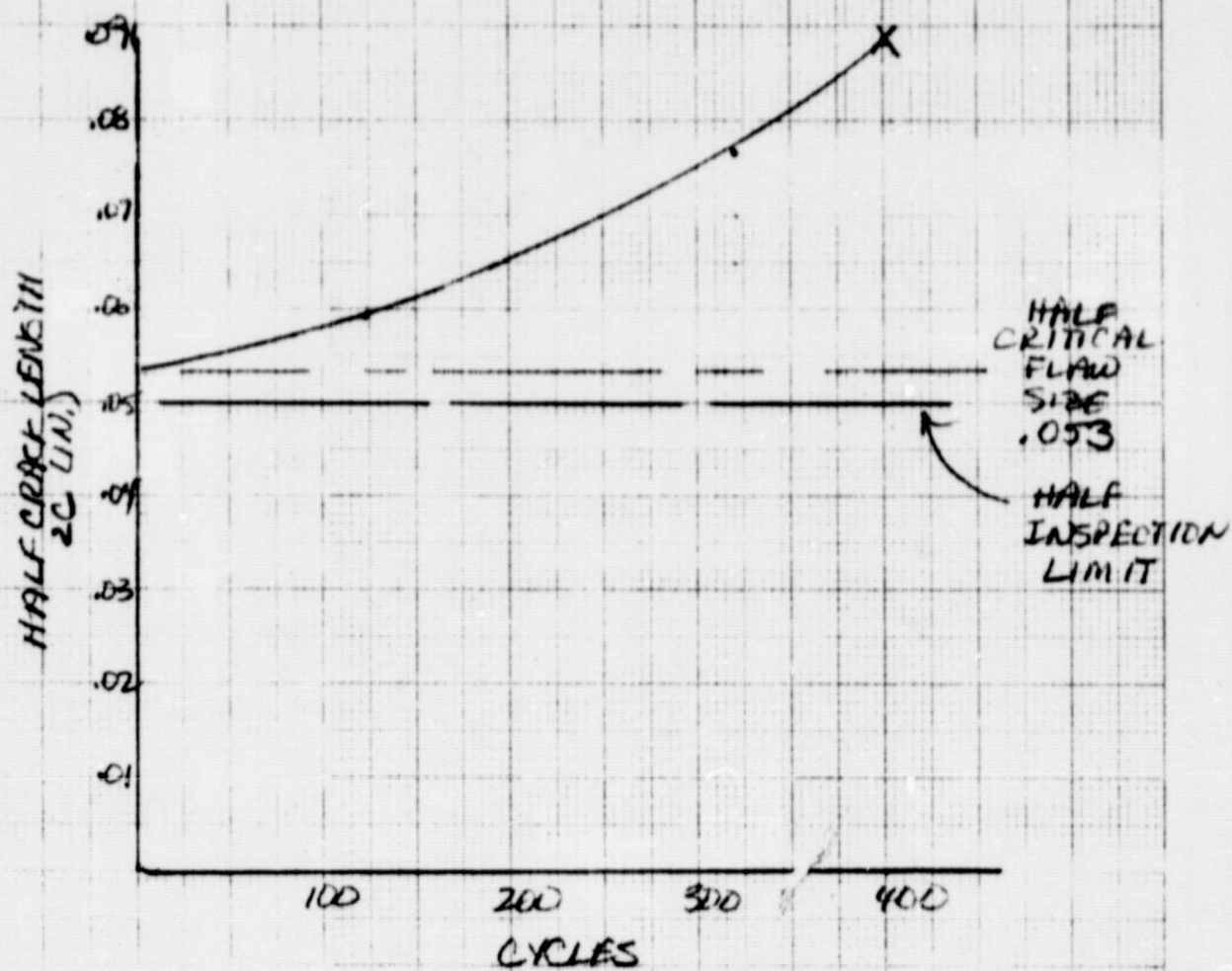
FROM COMPUTER ANALYSIS
 $\frac{1}{2} CF = .053 \text{ IN}$
OR $CF = .106 \text{ IN}$

EDDY CURRENT INSPECTION LIMIT - .1 IN

.106 > .1

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* FROM PREVIOUS SPARTAN REH ANALYSIS
LOADS UNCHANGED



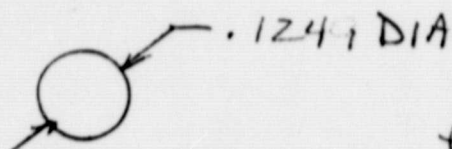
WORM GEAR PIN (30A60043)
SURFACE FLAW

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OF POOR QUALITY

Prepared by:	DSM	Date	6/84	LOCKHEED MISSILES & SPACE COMPANY, INC.	Page	Temp.	Perm.
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Approved by:		Date		SPARTAN REM	Model		
					Report No.		

SPARTAN REM ANALYSIS

WCPM PIN (30A60021)



MATL-MP35N ST

FROM STRESS ANALYSIS -

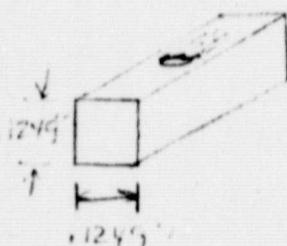
$$\sigma_{MAX} = 92.8 \text{ KSI} *$$

LET $\sigma_{MIN} = 0$ (CONSERVATIVE)

LOADED 400 CYCLES INCLUDING
A LIFE FACTOR OF 4.

MODEL

ASSUME A ROD IN TENSION



FROM COMPUTER ANALYSIS

$$\Sigma CF = .045 IN$$

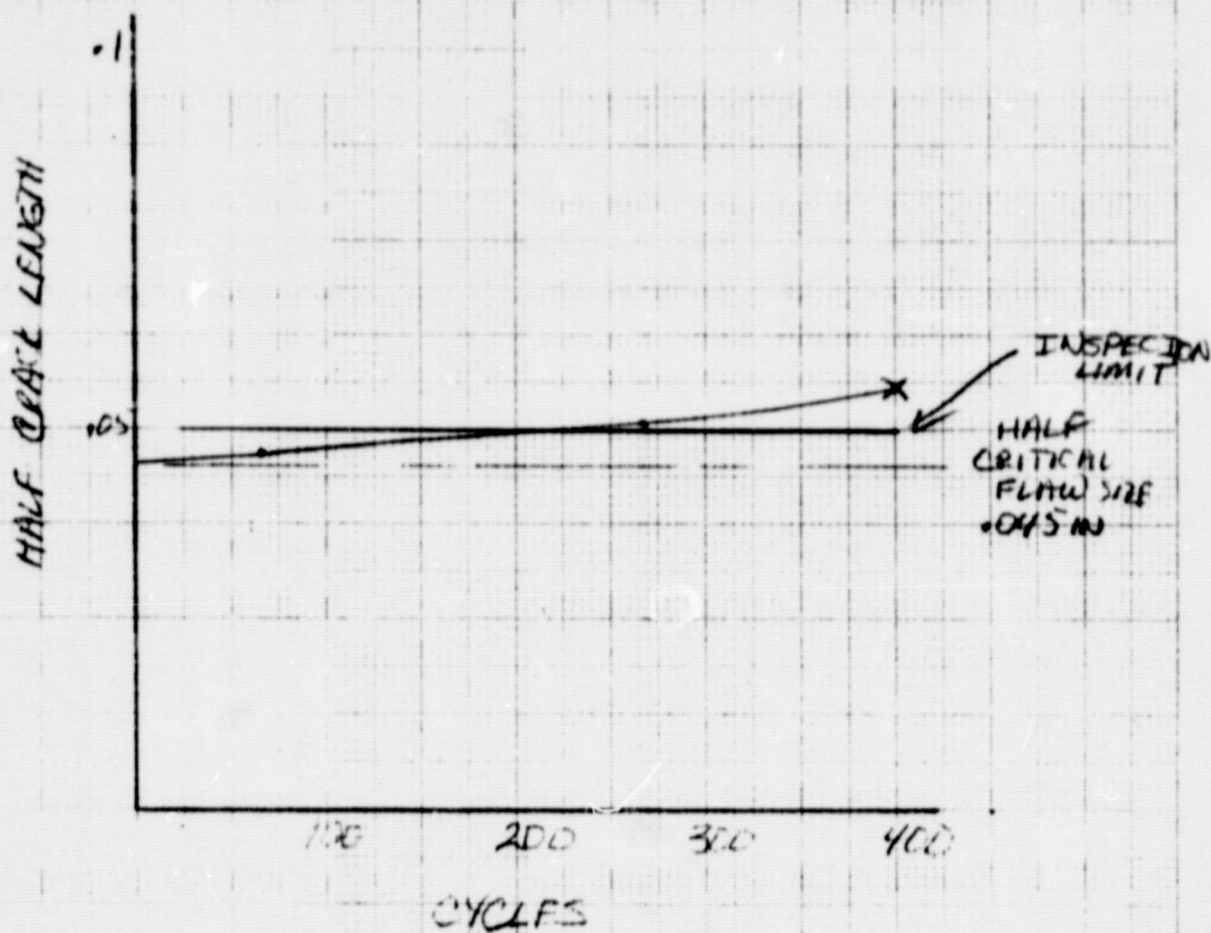
$$OF CF = .09 IN$$

EDDY CURRENT INSPECTION LIMIT
- .1 IN

* $CF < INSP \text{ LIMIT}$

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* FROM PREVIOUS SPARTAN REM ANALYSIS
LOADS UNCHANGED



WORM PIN - (30460021) SURFACE FLAW

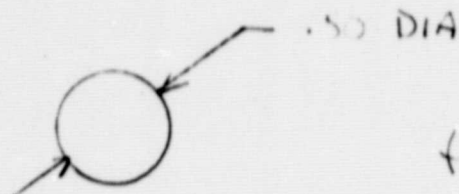
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Approved by:	Date:		Report No.:		

FRACTURE ANALYSIS

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WORK SHAFT (30460102)

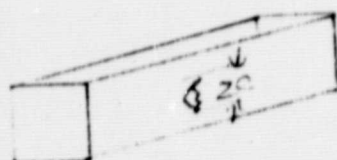


MATL- 416 STEEL

FROM STRESS ANALYSIS
 $T_{max} = 66,78 \text{ KSI}^*$
 CONSERVATIVELY let $T_{min} = 0$

LOADED 400 cycles
 $400 = 100 \text{ cycle} \times \text{LIFE FACTOR}$
 LIFE FACTOR = 4

11.6 IN



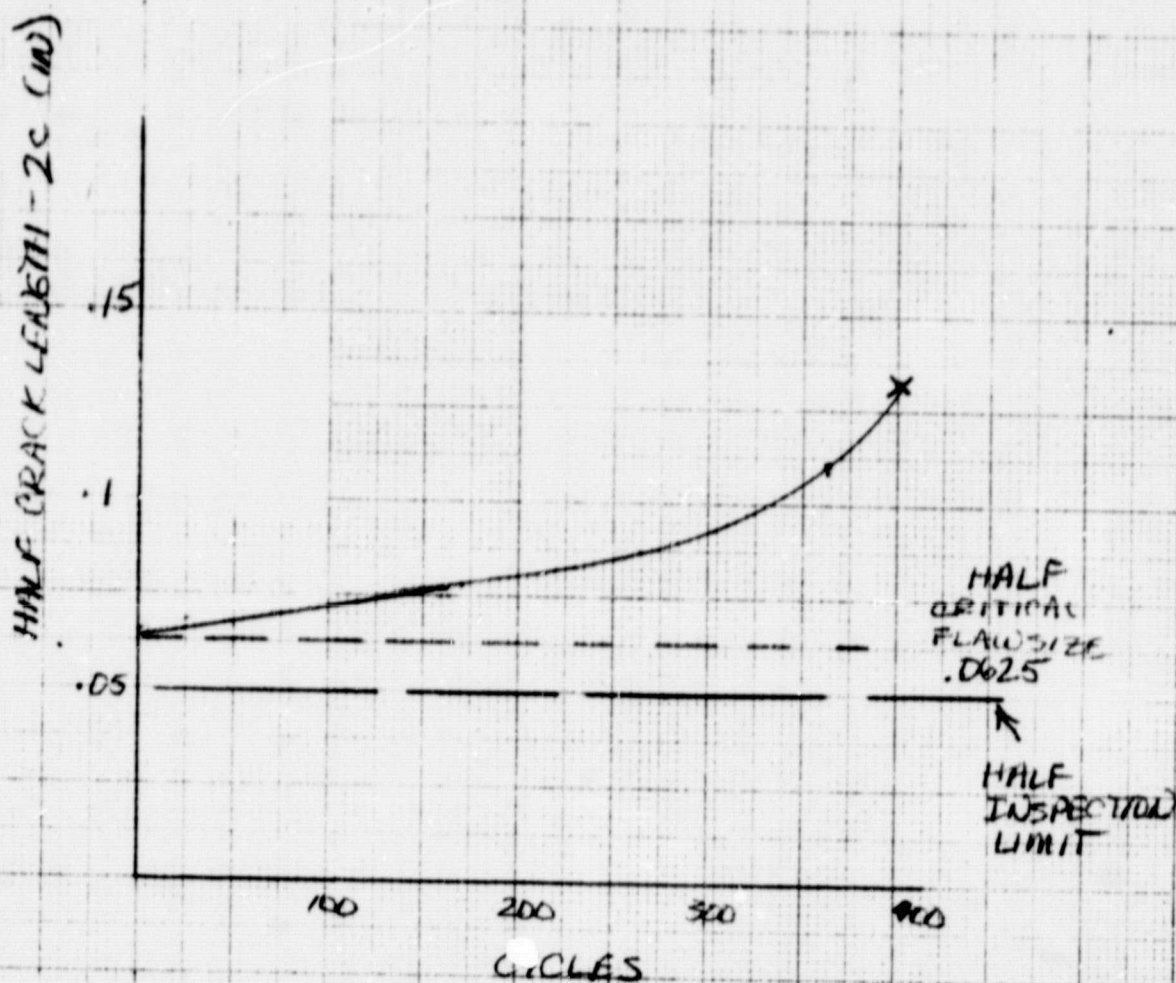
ASSUME RED IN TRANSITION
 WITH SURFACE FLAW

FROM COMPUTER ANALYSIS
 $\frac{1}{2} CF = .0625 \text{ IN}$
 OR $CF = .125 \text{ IN}$

INSPECTION - TDDY CURRENT
 MIN INSPECTION SIZE = .1 IN

$CF > \text{INS SIZE}$
 $.125 > .1$

* FROM PREVIOUS SPARTAN REM ANALYSIS
 LOADS UNCHANGED



WORM SHAFT (30A60102) - SURFACE FLAW

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